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# Computational Modeling of Indentation of Thin Films and Flow Through Porous Media 

A Dissertation Presented
by

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in
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# Abstract of the Dissertation <br> Computational Modeling of Indentation of Thin Films and Flow Through Porous Media 

 by
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A finite element model that captures the indentation force-depth response of a thin film system that exhibits isotropic elastic behavior and transversely isotropic plastic behavior on a substrate material that exhibits isotropic elastic behavior, indented by a sharp conical indenter was developed. Using dimensional analysis and a large number of finite element simulations, the relationships between the indentation response and the fundamental elastic and plastic properties of the substrate and the thin film were captured. It is demonstrated that both the forward analysis that predicts the indentation response from known material properties and the reverse analysis that predicts the material properties from known indentation responses were captured accurately. It is also demonstrated that the substrate's elastic property could also be simultaneously obtained along with the elastic and plastic properties of the indented thin film from the indentation analysis. Under conditions where the experimental results are very reliable with small errors, and within the range of material systems investigated in this study, the indentation method is expected to provide unique, robust and reliable predictions for the elastic and plastic properties of the thin film system.

A hybrid finite element/volume model that captures the flow and pressure drop characteristics in highly porous woven matrix media is developed. It is demonstrated that the geometric characteristics of a real woven matrix comprised of circular cross-section fibers and curvature due to fiber bending is captured well with an equivalent model system comprised of
fibers with square cross-section. A comprehensive study of the effects of changes in the finite element model size and defects in lay-up of the woven matrix layers on the predictions of the pressure drops was carried out. Changes in the in-plane size of the finite element model, lateral to the fluid flow direction, had relatively minor effects on the pressure drops predicted by the models. However, changes in the thickness of the finite element model in the fluid flow direction had significant effects on the pressure drops. In simulations with very thin models, the boundary effects had a greater influence on the overall flow behavior and caused the predicted pressure drops to increase proportionately. On the other hand, simulations with thick models indicated that the flows were fully developed and the boundary effects were minimized, resulting in relatively smaller pressure drops. Furthermore, defects in the lay-up of the woven matrix layers were also shown to have a significant impact on the pressure drops predicted by the simulations. Higher defect densities resulted in greater pressure drops as they disrupted the steady flow of fluid in the throughthickness direction. The pressure drops obtained in the finite element model simulations of thick models that contained some defective layers matched very well with experimental observations.

This dissertation is dedicated to my mother, my grandfather, my wife, and, who I care and who care me, of course.

## Table of Contents

Table of Contents ..... vi
List of Figures ..... viii
List of Tables ..... xii
Acknowledgments ..... xiii
Chapter 1: Computational Modeling of Indentation of Thin Films ..... 1
1.1 Introduction ..... 1
1.2 Background ..... 3
1.2.1 Introduction to instrumented indentation method ..... 3
1.2.2 Introduction to transversely isotropic materials ..... 8
1.2.3 Introduction to dimensionless analysis ..... 11
1.3 Finite element models ..... 12
1.4 Discussion ..... 17
1.4.1 Influences of transversely isotropic elastic properties of thin films on $\boldsymbol{P}-\boldsymbol{h}$ curves. ..... 17
1.4.2 Dimensionless analysis for current transversely isotropic thin film system ..... 19
1.4.3 Verification of dimensionless analysis on current system ..... 21
1.4.4 Expression creating based on dimensionless analysis ..... 23
1.4.5 Forward analysis for current transversely isotropic thin film system. ..... 27
1.4.6 Reverse analysis for current transversely isotropic thin film system ..... 30
1.4.7 Sensitivity of reverse analysis in present study ..... 35
1.5 Conclusions ..... 37
Chapter 2: Computational Modeling of Flow Through Porous Media ..... 38
2.1 Introduction ..... 38
2.2 Background - Pressure drops and friction factors in porous media ..... 42
2.3 Experiment Set-up for Characterizing Flow through Woven Matrix Porous Media ..... 48
2.4 Numerical Modeling ..... 51
2.4.1 Rationale for modeling woven networks with square fibers ..... 51
2.4.2 Numerical models for flow through woven matrix and random stacked media with square shaped fibers ..... 54
2.5 Discussion ..... 61
2.5.1 Comparison of simulation results between circular fiber and square fiber models.. ..... 61
2.5.2 Large numerical models for assessing size effects ..... 66
2.5.3 Comparison of simulation results with existing correlations and experiment data ..... 75
2.5.4 Effect of defects on flow behavior in porous media with woven structures ..... 78
2.5.6 A method to increase application scope of a correlation: modification of $\boldsymbol{d} \boldsymbol{h}$ ..... 83
2.5.7 Study of feasibility as applying small-scale models on random stacked media ..... 95
2.6 Conclusion ..... 99
Future Works ..... 101
References ..... 102
Appendix A ..... 109
Appendix B ..... 110
Appendix C ..... 147
Appendix D ..... 150

## List of Figures

Figure 1. Schematic illustrating a $P-h$ curve in instrumented indentation method..................... 3
Figure 2. Schematic illustrating the idea of forward and reverse analysis. ..................................... 6
Figure 3. Schematic illustrating the transversely isotropic thin film system of interesting in present study.............................................................................................................................................. 12

Figure 4. Plots displaying selected transversely isotropic thin film system properties: (a) $\sigma f$ vs $n$; (b) Ef vs Es; (c) $\sigma f$ vs $\sigma f L \sigma f T$................................................................................................... 13

Figure 5. Diagram displaying mesh details of a transversely isotropic thin film model. .............. 16
Figure 6. Plots illustrating how $P-h$ curves change with transversely anisotropic ratio of $E f L / E f T$ and $\sigma f L \sigma f T$ on different substrate: (a) and (c) elastic only substrate, (b) and (d) single crystal silicon.

Figure 7. Five thin film systems with different properties show clearly distinguishable $P-h$ curves. ........................................................................................................................................... 22

Figure 8. Schemitcs illustrating the strategy of grouping according to the value of yi4 and yi5. bi1, bi2 and bi3, and ci1, ci2 and ci3 are critical values of yi4 and yi5, respectively, where $i=$ 1,2 or 3 , presents the indenter with $60^{\circ}, 70.3^{\circ}$ and $80^{\circ}$, respectively. ......................................... 24

Figure 9. Sehcmetic illustrating forward analysis flow path. ....................................................... 27
Figure 10. Sehcmetic illustrating reverse analysis flow path. ...................................................... 31
Figure 11. Percentage and absolute errors of reverse analysis results of five properties of all 100 test samples: (a) $E f$; (b) $\sigma f$; (c) Es; (d) $\sigma f L \sigma f T$; (e) n............................................................... 33

Figure 12. Sensitivity of system properties for the reverse analysis in present study. (a)-(c) $E f$; (d)(f) $\sigma f$; (g)-(i) Es; (j)-(l) $\sigma f L \sigma f T$; (m)-(o) n................................................................................. 36

Figure 13. Schematic illustrating a $4 \times 4$ open porous section in a single layer woven matrix structure where $d, \mathrm{w}$ and t represent the fiber diameter, open pore size and the layer thickness, respectively, where $t=2 d$.

Figure 14. Schematic illustrating a porous medium (a random stacked medium) configure with randomly stacked fibers

Figure 15. (a) Schematic and (b) a photograph of the experimental set-up used for measuring pressure drops in woven matrix porous media. 48

Figure 16. Schematics illustrating fibers with circular cross-sections (a) and square cross-sections (b), and $a$ represents the thickness of square crossing sectional fiber. The woven matrix with fiber
bending (c) is modeled with a woven matrix with flat fibers with a step-type configuration in the cross-over regions (d).52

Figure 17. Optimal micrograph of a woven matrix open porous structure used in the experiments. 53

Figure 18. Schematics illustrating the spatial orientation and sequence of woven matrices........ 54
Figure 19. Schematics demonstrating a porous media contains eight layers of randomly distributed fibers. 55

Figure 20. ( $\mathrm{a}, \mathrm{b}, \mathrm{c}$ ) The geometric features of a numerical model developed in the present study to capture the flow behavior through a $4 \times 4$ open porous section of a woven matrix structure (4-layer
$\qquad$
Figure 21. Schematics illustrating mesh details used to represent the fluid part of the model: (a) a 4-matrix-layer with $4 \times 4$ opens woven matrix model; (b) an 8 -fiber-layer random stacked medium model. 58

Figure 22. The pressure drops obtained in the numerical model simulations of a 4 layered, $4 \times 4$ open porous section of a woven matrix where the fluid element size is varied from 6 microns to 18 microns indicate that the pressure drops decrease as the element size decreases. The pressure drop obtained in a model with element size ' i ' $\left(\Delta \mathrm{P}_{\mathrm{i}}\right)$ is normalized by the pressure drop observed in the model with the element size of 6 microns $\left(\Delta \mathrm{P}_{6}\right)$. 59

Figure 23. Schematics illustrating two kinds of 4-layer porous media with $4 \times 4$ open woven matrix comprised by circular fibers: (a) and (b) contact model; (c) and (d) compact model. 61

Figure 24. The characteristic relationship between the friction factor $C f$ and the Reynold's number $R e$ as predicted by the 4-layer numerical model simulations of a $4 \times 4$ open porous section of a woven matrix with square fibers and circular fibers compared to results by Costa [59].............. 62

Figure 25. Plots exhibiting resultant velocity maps of all three models (woven matrix media model, contact model and compact model) at each layer.

Figure 26. Plots exhibiting velocity distribution maps in all three directions of contact and compact models: (a)-(c) the contact model and (d)-(f) the compact model. ............................................... 65

Figure 27. The pressure drops obtained in the numerical model simulations of a 4-layer woven structure where the model size is increased from a $2 \times 2$ to a $10 \times 10$ open porous section. The pressure loss obtained in a model with open porous section ' $\mathrm{i} x \mathrm{i}$ ' $\left(\Delta \mathrm{P}_{\mathrm{ixi}}\right)$ is normalized by the pressure loss observed in the model with an open porous section of $10 \times 10\left(\Delta \mathrm{P}_{10 \times 10}\right)$............... 66

Figure 28. The resultant velocity distributions observed at the bottom of midstream part in the finite element models with (a) $2 \times 2$; (b) $4 \times 4$; (c) $6 \times 6$; and (d) $8 \times 8$, open porous sections.... 68

Figure 29. The pressure drops obtained in the finite element model simulations of a $4 \times 4$ open section of a woven structure, where the model size in the through-thickness direction is increased
from 4 layers to 110 layers. The pressure drop per unit length obtained in a model with ' i ' layers $(\Delta \mathrm{P} / \mathrm{L})_{\mathrm{i}}$ is normalized by the pressure drop observed in the model with 110 layers $(\Delta \mathrm{P} / \mathrm{L})_{110} \ldots . . .69$

Figure 30. The through-thickness flow velocity distributions observed in the numerical models of a $4 \times 4$ open porous section of a woven matrix with (a) 4 layers and (b) 50 layers in the throughthickness direction indicating that the outlet flow fields are different in the two cases. 70

Figure 31. The flow velocity distributions observed in the numerical models of a $4 \times 4$ open porous section of a woven matrix in the thin models in $4^{\text {th }}$ layer (a) and the thick models in the $4^{\text {th }}(\mathrm{b}), 10^{\text {th }}$ (c), $20^{\text {th }}(\mathrm{d}), 30^{\text {th }}$ (e) and $50^{\text {th }}$ (f) layers......................................................................................... 71

Figure 32. (a) The variation of fluid pressure as the fluid flows through a woven matrix porous medium observed in finite element modeling of a porous medium with 110 layers. (b) The rate of change in pressure as the fluid flows through the porous medium. ( L is the through thickness distance in the porous medium.) 72

Figure 33. The rate of change in pressure as the fluid flows through the porous medium with different values of Reynold number.73

Figure 34. The characteristic relationship between the friction factor $C f^{\prime}$ or $C f$ and the Reynold's number $R e^{\prime}$ or $R e$ as predicted by Sodré and Parise [69] or a two-parameter model [84] and a three-parameter model [83] for the friction factor compared to finite element model simulations and experiments.76

Figure 35. Schematics illustrating the introduction of a misaligned (defective) top layer by laterally displacing the top layer by a distance that is equal to half the open pore size.............................. 77

Figure 36. Schematics illustrating the introduction of a misaligned defective layer every 4, 5, 8, 10 or 20 layers in a model with 40 layers. ........................................................................................ 78

Figure 37. The characteristic relationship between the friction factor $C f$ and the Reynold's number $R e$ as predicted by the 40 -layer thickness numerical model simulations of a $4 \times 4$ open porous section of a woven matrix with misaligned defective layers and observed in experiments. ........ 79

Figure 38. Schematic illustrating the introduction of defective layers every 8 layers in a model with 40 layers. 80

Figure 39. Schematics illustrating the introduction of defective layers with different extents of misalignments (i.e., lateral displacements of $35 \mu \mathrm{~m}$ (a), $70 \mu \mathrm{~m}$ (b), $105 \mu \mathrm{~m}$ (c), and $140 \mu \mathrm{~m}$ (d) relative to the perfectly aligned position of the layers), in a model with 40 layers.80

Figure 40. The characteristic relationship between the friction factor $C f$ and the Reynold's number $R e$ as predicted by the numerical model simulations of a 40 -layer thickness $4 \times 4$ open porous section of a woven matrix with defective layers which have different extents of misalignments every eight layers and observed in experiments.81

Figure 41. Plot includes all 114 models with different fiber thickness and Reynold number. ..... 84

Figure 42. Plots illustrating the characteristic relationship between the friction factor $C f$ and the Reynold's number $R e$ as predicted by 4-layer numerical models with dissimilar fiber thickness and divergent porosity

Figure 43. Plot illustrating the characteristic relationship between the friction factor $C f$ and the Reynold number Re based on Gedeon's research [83] on regenerators comprised of woven screens. 86

Figure 44. Schematic illustrating the overlap area between any two square crossing sectional fibers which would greatly impact the calcualtion of specific surface of solid, $S v$.

Figure 45. Plot illustrating the characteristic relationship between the modified friction factor $C f *$ and the modified Reynold number $R e *$ based on Gedeon's research [83] on regenerators comprised of woven screens.

Figure 46. Plots illustrating the characteristic relationship between the modified friction factor $C f *$ and the modified Reynold number $R e *$ as predicted by 4-layer numerical models with dissimilar fiber thickness and divergent porosity. ........................................................................ 91

Figure 47 Plots displayed the simulation results from 114 models: (a) from the view of $C f$ and


Figure 48. Plots displayed pressure loss per unit length, $\Delta P / \Delta L$, for two groups of random stacked models. 95

Figure 49. Plots illustrating the dispersiosn of simualtion results for two groups of random stacked models. 96

Figure 50. The average friction factor, $C f$, obtained from two groups of random stacked models were compared with the consequence of Gedeon's work [83], and experimental and simulation results of present study.

## List of Tables

Table 1. Range of Material Properties ..... 14
Table 2. Five transversely isotropic thin film systems having diverse properties for both thin andsubstrate.21
Table 3. Five transversely isotropic thin film systems with diverse properties for both thin andsubstrate show the same value for $П 21$ to П2522
Table 4. Coefficient of determination, R2, for fitting yijk with $x 1, x 2, x 3$ and $x 4$. ..... 25
Table 5. Percentage errors of forward analysis results for all yijk. ..... 28
Table 6. Detailed reverse analysis results for test sample with number 19. ..... 30
Table 7. Statistics absolute errors of reverse analysis results for 100 test samples. ..... 32
Table 8. Statistics percentage errors of reverse analysis results for 100 test samples. ..... 32
Table 9. Sensor Information ..... 49

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## Chapter 1: Computational Modeling of Indentation of Thin Films

### 1.1 Introduction

With the development of science and technology, as well due to the needs of human life, materials from macro to micro, even to nano scales, have been applied widely. The performance of these materials varies significantly while their size changes. In order to characterize material properties in small scales, i.e., mechanical or piezoelectric properties, instrumented indentation method, which was developed from the traditional hardness testing method, emerged to provide people an easy, convenient and accurate technique. In recent decades, instrumented indentation based methods for determining mechanical properties of bulk specimen or thin film have received considerable and continue growing attention, due to its operability, potential applications and nondestructive features [1-3].

Rely on instrumented indentation method, numerous studies have successfully predicted material properties for the isotropic bulk sample [3, 4] and thin film [5], and even transversely isotropic bulk material [6]. Yet, most of those studies were focused on systems with a small number of unknown properties (less or equal to three), while fewer works tried to apply instrumented indentation method on a system with many variables, let alone serious discussions about the robustness, sensitivity or uniqueness in that case. Meanwhile, most studies about thin films assumed that the properties of the substrate were already known in advance, or they could be obtained through an alternative method. Moreover, fewer papers talked about if it was possible to extract the elastic or plastic properties of the substrate in a thin film system, even to get the properties of both the thin film and the substrate at the same time, i.e., from a single set of instrument indentation tests. In the end, although some works have been done for transversely isotropic bulk material [6], this property is obviously more important and popular for the thin film system. Nevertheless, almost no relevant literature could be found.

Based on the descriptions above, the three main purposes of the present study are:

1. To check the feasibility, together with the robustness, sensitivity, and uniqueness of
instrumented indentation method on a thin film system with a large number of variables;
2. To check the possibility of predicting the properties of the substrate and thin film simultaneously using instrumented indentation method;
3. To check the accuracy of using instrumented indentation method for extracting the transverse properties of thin film system.

The structure of this chapter is organized as follows. Background information and a summary of prior work done in predicting mechanical properties of bulk and thin film system in Section 1.2. The details of the numerical model developed in the present study are highlighted in Section 1.3. The results obtained from the present study are discussed in Section 1.4 and key conclusions from the present work are summarized in Section 1.5.

### 1.2 Background

### 1.2.1 Introduction to instrumented indentation method

$P-h$ curve, as shown in figure 1 , captures the indentation force-depth response of the specimen. The main idea of instrumented indentation method is to create a relationship between $P-h$ curves, and mechanical properties of the corresponding material system. In other words, in instrumented indentation method, mechanical properties of a bulk or a thin film system are obtained by collecting enough information from $P-h$ curve(s), which are obtained through indenting experiments.


Figure 1. Schematic illustrating a $P-h$ curve in instrumented indentation method.

In figure 1 , there are two solid curves and the left one is referred to as loading curve while the right one is referred to as unloading curve, both represent the relationship between ' $P$ ' and ' $h$ '. Here, ' $h$ ' represents the displacement of indenter, therefore, $h_{1}, h_{2}$ and $h_{\max }$ correspond to indenter displacement as it at position 1 , position 2 and maximum displacement, respectively. ' $P$ '
represents the force acted on the indenter, therefore, $P_{1}, P_{2}$ and $P_{\max }$ stand for the value of force as indenter at position 1, position 2 and maximum displacement. Moreover, ' $S$ ', in figure 1 , is the slope of the unloading curve, as $S_{\text {mas }}$ represents the slope at the very initial point of unloading curve when indenter is infinitely close to maximum displacement. ' $W$ ' is the work done as indenter moves, therefore, $W_{l}$ and $W_{u}$ represents the work done in loading and unloading process, i.e., the area under loading and unloading curves, respectively.

Sneddon derived a general equation for the indenter displacement, $h$, and the loading load, $P$, as a solid indenter that could be described as revolution of a smooth function forced into a bulk material, as [7]:

$$
\begin{equation*}
P=C h^{m} \tag{1}
\end{equation*}
$$

where constant $C$ is the loading curvature which is decided by the properties of both indenter and specimen, and $m$ is a constant for a certain kind geometric of the indenter, i.e., $m=2$ for conical indenters, $m=1.5$ for sphere and paraboloids of revolution tip indenters, and $m=1$ for flat cylinder indenters. Equation 1 was valid only when material deformations were limited to elastic.

With advanced studies, Oliver and Pharr found that when plastic deformation was included in consideration, above relationships would still work for cone indenter [8], however, the value of $m$ was no longer a fixed value for sphere punches with different radius [9]. Oliver and Pharr [8], based on Sneddon's analysis [7], proposed that the elastic property of bulk materials can be obtained, if $P-h$ curve was accessible, from the equation as:

$$
\begin{equation*}
S_{\max }=2 E_{r} r=\frac{2}{\sqrt{\pi}} E_{r} \sqrt{A_{c}} \tag{2}
\end{equation*}
$$

where, $r$ is the contact radius, $A_{c}$ is the contact area, and the effective elastic modulus $E_{r}$ could be expressed as:

$$
\begin{equation*}
\frac{1}{E_{r}}=\frac{1-v^{2}}{E}+\frac{1-v_{i}^{2}}{E_{i}} \tag{3}
\end{equation*}
$$

where the subscripts ' $i$ ' represents the indenter. Therefore, $E_{i}$ and $v_{i}$ are the elastic modulus and Poisson's ratio of indenter, respectively, while $E$ and $v$ stand for the elastic modulus and Poisson's
ratio of material of interest. They also got good results for determining the hardness of bulk specimen with above method [10]. Nevertheless, above analysis is restricted to the condition when contact edge, under the indenter, is sink-in, while for lots kinds of materials, contact edge is pileup [11]. In equation 2, the estimation of contact area $A_{c}$ between indenter and sample, which was usually inaccurate due to pile-up or sink in phenomena, has a direct influence on the calculation of material elastic modulus.

Dao [12], Cheng and Cheng [13, 14] were the pioneers of introducing the scaling laws and dimensionless analysis for developing relationships between the characteristic value on $P-h$ curves and material properties. By applying dimensionless analysis, the procedure of estimating contact area, $A_{c}$, could be avoided, firstly, and plastic properties of the specimen could also be obtained from $P-h$ curves through dimensionless equations. During this period, Venkatesh et al [2] proposed the idea of forward and reverse analysis, which extended the extent of instrumented indentation method, as shown in figure 2. It was not only possible to predict the properties of materials, but also possible to forecast $P-h$ curves if material properties were known in advance. Dimensionless analysis and forward and reverse analysis have almost become a fixed process of instrument indentation method today, esp., for the complex material systems.

Although based on the work of Oliver and Pharr [8], lots of improved empirical and semiempirical formulae have been proposed from both experimental observations and numerical simulations [ $9,12,15]$, there are many limitations when applying them. They either only work for the bulk system or only consider a simple system with a small number of variables. On the other hand, the intricate stress and strain status under indenters (in samples) during indenting is still poorly understood even as instrumented indentation method was widely accepted. So as to use instrumented indentation method on increasingly complex and practical problems, more studies would be expected on a more general material system, and also on the explanations for the stories happening inside it.

Thanks to the speedy development of computational technics, numerical methods today (finite element method, dimensionless analysis and forward and reverse analysis), can provide people a hands-on experience of understanding the relationship between material properties and $P-h$ curves (dimensionless analysis, forward and reverse analysis), and also enable people to
visualize the indentation processes and responses inside the materials which are with flexible property combinations (finite element method).

## Forward Analysis



Reverse Analysis

Figure 2. Schematic illustrating the idea of forward and reverse analysis.
Finite element method has been widely used for simulating instrumented indentation tests today [6]. Yet, in order to obtain a correct simulation outcome, the phenomenon of scale dependent effect (SDE) requires particular attention. When indentation happens in a significant small region, e.g., smaller than a micron, the phenomenon of scale dependent effect (SDE) becomes nonnegligible [16-18]. Several factors, such as friction between the indenter and the sample [19], strain gradient hardening [20] and surface free energy effect [21] could be helpful in explaining SDE. A pure Finite element method is difficult to take into account the influence of SDE since the classic continuum plasticity theory does not have a constituent internal length as a parameter for deformation [22]. In another word, the indentation depth should not be less than a micron [22], if conventional plasticity theories were used to describe the mechanic behaviors of a bulk specimen or a perfect connected thin film on a substrate.

Chen [23] proposes a method of using the impact of the substrate, i.e., introducing a characteristic length, to analyze the indentation process in a thin film system. They further put forward that $P_{1}, P_{2}$ and $W_{u}$ could be three independent characteristic responses on the $P-h$ curve. Bhat and Venkatesh [6] introduced a ratio between the plastic work, $W_{p}$, and total work done in the loading procedure, $W_{l}$, to interpret P-h curves for transversely isotropic bulk materials.

Except predicting elastic and plastic properties from $P-h$ curves of indenting specimens [24, 25], instrument indentation method can also help to learn the deformation, fatigue, creep, fail behaviors of materials [26-31]. Moreover, this method is not restricted in the area of engineering, it has been adopted in many fields like geology [32], biomedicine [33], marine biology [34] etc. In a word, instrumented indentation method is a potential but powerful technology which can be widely applied.

### 1.2.2 Introduction to transversely isotropic materials

A transversely isotropic material is one with physical properties symmetric about an axis normal to a plane of isotropy, which means within this plane, all properties of this material are the regardless of directions. In the present paper, the direction perpendicular to the plane of isotropy is defined as the longitudinal direction, as directions being parallel to this plane is defined as the transverse directions.

Many thin films and coating might exhibit transversely isotropic property due to their fabrication process and the resulting microstructures[35-37]. Nakamura et al[35] proposed that for a transversely isotropic material, there were five independent constants to describe the elastic properties. The compliance matrix of a transversely material can be represented as:

$$
\left[\begin{array}{l}
\varepsilon_{11}  \tag{4}\\
\varepsilon_{22} \\
\varepsilon_{33} \\
\gamma_{12} \\
\gamma_{23} \\
\gamma_{13}
\end{array}\right]=\left[\begin{array}{cccccc}
1 / E_{T} & -v_{L T} / E_{L} & -v_{T} / E_{T} & 0 & 0 & 0 \\
-v_{T L} / E_{T} & 1 / E_{L} & -v_{T L} / E_{T} & 0 & 0 & 0 \\
-v_{T} / E_{T} & -v_{L T} / E_{L} & 1 / E_{T} & 0 & 0 & 0 \\
0 & 0 & 0 & 1 / G_{L} & 0 & 0 \\
0 & 0 & 0 & 0 & 1 / G_{L} & 0 \\
0 & 0 & 0 & 0 & 0 & 1 / G_{T}
\end{array}\right]\left[\begin{array}{c}
\sigma_{11} \\
\sigma_{22} \\
\sigma_{33} \\
\sigma_{12} \\
\sigma_{23} \\
\sigma_{13}
\end{array}\right]
$$

where, the subscripts ' $L$ ' and ' $T$ ' are, respectively, denoted to directions of longitudinal and transverse. Similarly, subscripts ' 1 ' and ' 3 ' represent the directions within the transverse plane, while subscript ' 2 ' represents the direction of longitudinal. The Poisson's ratio $v_{T}, v_{L T}$ and $v_{T L}$ are defined as $-\frac{\varepsilon_{33}}{\varepsilon_{11}}$, $-\frac{\varepsilon_{11}}{\varepsilon_{22}}$ and $-\frac{\varepsilon_{22}}{\varepsilon_{33}}$, respectively. Meanwhile, the ratio between $E_{L}$ and $E_{T}$ and the ratio between $\nu_{L T}$ and $v_{T L}$ should be the same, as following:

$$
\begin{equation*}
\frac{E_{L}}{E_{T}}=\frac{v_{L T}}{v_{T L}} \tag{5}
\end{equation*}
$$

The in-plane shear modulus, $G_{T}$ could be expressed as:

$$
\begin{equation*}
G_{T}=\frac{E_{T}}{2\left(1+v_{T}\right)} \tag{6}
\end{equation*}
$$

In order to reduce the variable numbers of elastic properties, two assumptions were made here [6]. First, the sum of $v_{L T}$ and $v_{T L}$ is defined as twice the value of $v_{T}$, i.e., $2 v_{T}=v_{L T}+v_{T L}$. Second, the out-of-plane shear modulus, $G_{L}$ could be expressed as:

$$
\begin{equation*}
G_{L}=\frac{E_{0}}{2\left(1+v_{T}\right)} \tag{7}
\end{equation*}
$$

where, $E_{0}$ is the averaged elastic modulus of longitudinal and transverse direction of the thin film or the reference elastic modulus, i.e., $\left(E_{L}+E_{T}\right) / 2$. So far, the elastic properties of a transversely isotropic material might be defined by only 3 parameters, i.e., $E_{0}, E_{L} / E_{T}$ and $v_{T}$.

For the plastic properties, Hill's [38] developed a yield criterion for anisotropic plastic deformation as:

$$
\begin{align*}
F\left(\sigma_{22}-\sigma_{33}\right)^{2} & +G\left(\sigma_{33}-\sigma_{11}\right)^{2}+H\left(\sigma_{11}-\sigma_{22}\right)^{2}+2 L{\sigma_{23}}^{2}+2 M \sigma_{31}^{2}+2 N \sigma_{12}^{2} \\
& =1 \tag{8}
\end{align*}
$$

where, $F, G, H, L, M$ and $N$, are constants that should be determined experimentally. Equation 8 can be modified for transversely isotropic materials as [35]:

$$
\begin{aligned}
& f(\sigma)=\sqrt{P\left(\sigma_{22}-\sigma_{33}\right)^{2}+P\left(\sigma_{22}-\sigma_{11}\right)^{2}+Q\left(\sigma_{11}-\sigma_{33}\right)^{2}+2 R{\sigma_{23}^{2}+2 R{\sigma_{12}^{2}}^{2}+2 S \sigma_{13}^{2}} \quad-\sigma_{0}} \text {. }
\end{aligned}
$$

and $f(\sigma)=0$
where, $P, Q, R$ and $S$ are the dimensionless constants that related to $\sigma_{0}$ as:

$$
\left\{\begin{array}{l}
P=\frac{1}{2}\left(\frac{\sigma_{0}}{\sigma_{L}}\right)^{2}  \tag{10}\\
Q=\frac{1}{2}\left(2 \frac{\sigma_{0}{ }^{2}}{\sigma_{T}^{2}}-\frac{\sigma_{0}^{2}}{\sigma_{L}^{2}}\right) \\
R=\frac{1}{2}\left(\frac{\sigma_{0}}{\tau_{L}}\right)^{2} \\
S=\frac{1}{2}\left(\frac{\sigma_{0}}{\tau_{T}}\right)^{2}
\end{array}\right.
$$

where, $\sigma_{0}$ is the reference yield stress equal to $\left(\sigma_{L}+\sigma_{T}\right) / 2$, and $\sigma_{L}, \sigma_{T}, \tau_{L}$ and $\tau_{L}$ are yield stress along different directions, respectively. Lan et al. [3] described stress-strain and elastoplastic behavior of bulk materials as:

$$
\sigma=\left\{\begin{array}{c}
E \varepsilon,\left(\sigma \leq \sigma_{Y}\right)  \tag{11}\\
R \varepsilon^{n}=\sigma_{Y}\left(1+\frac{E}{\sigma_{Y}} \varepsilon_{n}\right)^{n},\left(\sigma \geq \sigma_{Y}\right)
\end{array}\right.
$$

where $n$ is the strain hardening exponent, $\sigma_{Y}$ is the yield stress at initial, and $\varepsilon_{n}$ is the non-linear strain. In order to reduce the variable numbers of plastic properties, three assumptions for plastic properties were made in present work [6]. First, so as to extend the isotropic power law hardening to transversely isotropic materials, the post yield behavior for both longitudinal and transverse directions would be assumed to follow the same rules as bulk materials. Second, longitudinal and transverse directions would share the same constant work hardening exponent. After introducing these two assumptions, elastoplastic behavior of transversely isotropic materials could be described as in equation 12 and equation 13, for longitudinal and transverse direction, respectively:

$$
\sigma_{L}=\left\{\begin{array}{c}
E_{L} \varepsilon_{L},\left(\sigma_{L} \leq \sigma_{Y L}\right)  \tag{12}\\
R \varepsilon_{L}^{n}=\sigma_{Y L}\left(1+\frac{E_{L}}{\sigma_{Y L}} \varepsilon_{n L}\right)^{n},\left(\sigma_{L} \geq \sigma_{Y L}\right)
\end{array}\right.
$$

and

$$
\sigma_{T}=\left\{\begin{array}{c}
E_{T} \varepsilon_{T},\left(\sigma_{T} \leq \sigma_{Y T}\right)  \tag{13}\\
R \varepsilon_{T}^{n}=\sigma_{Y T}\left(1+\frac{E_{T}}{\sigma_{Y T}} \varepsilon_{n T}\right)^{n},\left(\sigma_{T} \geq \sigma_{Y T}\right)
\end{array}\right.
$$

Third, shear yield stress of longitudinal and transverse directions follows the von Mises criterion, i.e., $\tau_{L}$ and $\tau_{T}$ can be approximately expressed as:

$$
\left\{\begin{array}{c}
\tau_{L}=\frac{\sigma_{0}}{\sqrt{3}} \sqrt{\frac{\sigma_{L}}{\sigma_{T}}}  \tag{14}\\
\tau_{T}=\frac{\sigma_{0}}{\sqrt{3}}
\end{array}\right.
$$

It can be seen, now the plastic properties of transversely isotropic materials may also be defined by 3 parameters, i.e., $\sigma_{0}, \sigma_{L} / \sigma_{T}$ and $n$.

In summary, totally six parameters $\left(E_{0}, E_{L} / E_{T}, v_{T}, \sigma_{0}, \sigma_{L} / \sigma_{T}\right.$ and $\left.n\right)$ are required to fully describe the mechanic behavior of transversely isotropic materials.

### 1.2.3 Introduction to dimensionless analysis

Two geometric objects are described as geometric similar as their lengths are all proportional to each other with the same proportional constant, and all angles are identical [39]. For instrumented indentation, sphere and flat cylinder indenters, which by themselves possess characteristic lengths, are not geometric similar, due to that there is another length parameter, i.e., displacement of the indenter. However, all cone indenters are geometric similar if they have an equal half angle, $\theta$, at the tip.

The dimensionless analysis is the elementary idea helping to create the relationships between $P-h$ curve and material properties which could be, then, applied for further forward and reverse analysis. The so-called $\Pi$-theorem, proposed by Buckingham [40], is the basic law of dimensionless analysis. Buckingham emphasized that physical laws do not depend on arbitrarily chosen basic units of measurement, or all terms that are added together must have the same unit. In instrumented indentation, there are two basic dimensions, i.e., length and mass. Therefore, an unknown quantity, $y$, could be, without loss of generality, written as:

$$
\begin{equation*}
y=f\left(x_{1}, x_{2}, \cdots, x_{n}\right) \tag{15}
\end{equation*}
$$

then assume that $x_{1}$ and $x_{2}$ have independent dimensions, so equation 15 might be modified as:

$$
\begin{equation*}
\frac{y}{x_{1}^{c_{1}} \times x_{2}^{c_{2}}}=f\left(x_{1}, x_{2}, \frac{x_{3}}{x_{1} c_{31} \times x_{2}^{c_{32}}} \cdots, \frac{x_{n}}{x_{1} c_{n 1} \times x_{2}^{c_{n 2}}}\right) \tag{Eq}
\end{equation*}
$$

where, $c_{1}, c_{2}, c_{n 1}$ and $c_{n 2}$ are all constants which make $\frac{y}{x_{1} c_{1 \times x_{2}} c_{2}}$ and $\frac{x_{n}}{x_{1} c_{n 1 \times x_{2}} c_{n 2}}$ dimensionless terms. Since in a physical law, all terms must have the same unit, here in equation 16, all terms should be dimensionless. Therefore, equation 16 can be simplified as:

$$
\begin{equation*}
\frac{y}{x_{1}^{c_{1}} \times x_{2} c_{2}}=\Pi\left(\frac{x_{3}}{x_{1}^{c_{31}} \times x_{2}^{c_{32}}} \cdots, \frac{x_{n}}{x_{1}^{c_{n 1}} \times x_{2}^{c_{n 2}}}\right) \tag{17}
\end{equation*}
$$

It's obviously by applying $\Pi$-theorem, number of variables for describing an unknown quantity was reduced by two.

### 1.3 Finite element models

The transversely isotropic thin film system of interest in present work is shown in figure
Figure 3.


Figure 3. Schematic illustrating the transversely isotropic thin film system of interesting in present study.

As shown in section 1.2.2, there are at least six variables for depicting a transversely isotropic material. Therefore, there should be more than twelve variables for a totally transversely isotropic thin film system. As an exploratory research, only transversely isotropic plasticities were assigned to the top thin film, with its elastic properties still being isotropic. Later on, it will be shown that transversely isotropic elasticities have little influence on $P-h$ curves, which means it might ask for other assistant methods if transversely isotropic elasticities are wanted. Meanwhile, the bottom substrate was simplified to possessed an unknown isotropic elastic modulus only. The indentation happened along the longitudinal direction with conical indenters. The half angle of the conical indenter, $\theta$, was set equivalent to $60^{\circ}, 70.3^{\circ}$ or $80^{\circ}$. The substrate thickness was as large as 3,000 times the thickness of thin film, $\delta$, which enabled the assumption that all far boundary
conditions had ignorable effect on simulation results [41]. In the present study, for the purpose of direct explanations of simulation results, the film thickness was assigned to be $2 \mu \mathrm{~m}$, which, of course, is not necessary. Therefore, substrate thickness would equal to 6 mm according to previous discussion. The maximum displacement of indenters was also assigned to be $2 \mu \mathrm{~m}$ for all simulations, i.e., $100 \%$ 'penetration ratio' (not really to pierce through the thin film). Since there was no constituent internal length in current models (or say there was no internal unit system in Abaqus), simulation results could be easily transferred to represent the indentation for such thin film system with any thin film thickness, while the only requirement was that the maximum indenter penetration ratio, $h / \delta$, should equal to $100 \%$.

All material property combinations for the transversely isotropic thin film system are shown in figure 4.

- Selected system propeties



Figure 4. Plots displaying selected transversely isotropic thin film system properties: (a) $\sigma_{f}$ vs $n$; (b) $E_{f}$ vs $E_{s}$; (c) $\sigma_{f}$ vs $\sigma_{f L} / \sigma_{f T}$.

In figure 4, each black square represents a property combination of the models. In general, the initial yield stress of most metal or alloy falls within the range of 30 MPa to 1100 Mpa , with elastic modulus within the range of 40 GPa to 210 GPa , while work hardening is about 0 to 0.5
[42]. In present work, the value of film elastic modulus, $E_{f}$, ranged from 50 to 250 GPa , reference yield strength, $\sigma_{f}$, from 200 to 1000 MPa , strain hardening index, $n$, ranged from 0 to 0.5 , transversely isotropic ratio of plasticity, $\sigma_{f L} / \sigma_{f T}$, ranged from 1.0 to 1.5 , while the substrate elastic modulus, $E_{S}$, ranged from 50 to $250 \mathrm{GPa} . v_{T}$ and $v_{s}$ was set to be 0.3 for all situations since this approximation value falls close to most metal. Moreover, effect of Poisson's ratio could be ignored when simulating an instrumented indentation.

Detailed information of the possible combination of thin film system properties can also be found in table 1 . From table 1, and be aware of that there are three different half angle conical indenters, about $13500(5 \times 5 \times 6 \times 6 \times 5 \times 3=13500)$ simulations need to be carried out in the present study. So many finite element models were included so as to exclude the impacts, due to insufficient data support, on the analysis of the feasibility, robustness, and sensitivity of instrumented indentation method later.

Table 1. Range of Material Properties

| Properties | Min |  |  |  | Max |
| :---: | :---: | :---: | :---: | :---: | :---: |
| $\boldsymbol{E}_{\boldsymbol{f}}$ | 50 GPa | 100 GPa | 150 GPa | 200 GPa | 250 GPa |
| $\boldsymbol{\sigma}_{\boldsymbol{f}}$ | 200 MPa | 400 Mpa | 600 MPa | 800 MPa | 1000 MPa |
| $\boldsymbol{n}$ | 0 | 0.1 | 0.2 | 0.3 | 0.4 |
| $\boldsymbol{\sigma}_{\boldsymbol{L}}$ | 1.0 | 1.1 | 1.2 | 1.3 | 1.4 |
| $\boldsymbol{\sigma}_{\boldsymbol{T}}$ |  |  |  |  | 0.5 |
| $\boldsymbol{E}_{\boldsymbol{s}}$ | 50 GPa | 100 GPa | 150 GPa | 200 GPa | 250 GPa |

All simulations of the instrumented indentation in the present study were conducted by using two-dimensional (2D) axisymmetric models included in Abaqus 6.14. Although threedimensional models are preferred in recent studies, due to a lot of simulations need to be carried out, only 2D models were created in the present work. Moreover, 2D models can fully meet the requirement of describing the transversely isotropic properties of the system with features of
conical indenters. Thirdly, 2D models show excellent agreement with experimental results and simulation outcomes based on 3D models [43-45].

Roller boundary condition was assigned to the axis of in-plane symmetric. The displacement of nodes at bottom along the out-of-plane direction was fixed to zero while all nodes at other boundary sides unrestrained. Bucaille et al. [46] found that the friction coefficient between slave and master surfaces was very small for the large value semi-angle ( $60^{\circ}$ and $70.3^{\circ}$ ) conical indenters. Antunes et al. [47] further pointed it out that the influence of friction coefficient could be neglected. No friction between the surface of indenters and samples was considered in present work. Moreover, indenters were with perfect sharp tip and rigid body in all simulations. And the thin film was designed to be completely flat at the top so as to possess perfect connections to the substrate.

For each model, there were more than 18,000 elements, mixed of CAX4R and CAX3 types, as shown in figure 5. The element size was designed to be fine enough to make sure there were more than 100 elements to describe the interaction between slave and master surfaces from samples and indenters, respectively, as maximum displacement reached. Element size became increasingly coarse when near the far away boundary on considering of time consumed. All simulations are carried out by using general purpose finite element package of ABAQUS on DELL workstation T7500 with up to 8 cores.


Figure 5. Diagram displaying mesh details of a transversely isotropic thin film model.

### 1.4 Discussion

### 1.4.1 Influences of transversely isotropic elastic properties of thin films on $\boldsymbol{P} \boldsymbol{-} \boldsymbol{h}$ curves

Before analyzing the results from the systematical simulations, the influences from transversely isotropic properties of the thin film were first studied. The reference elastic modulus of the thin film was fixed as 800 MPa , while reference yield strength of the thin film fixed as 200 GPa, and strain hardening index fixed as 0.2. In figure 6(a) and (c), substrates was only assigned with isotropic elasticity as 150 GPa , while in figure 6 (b) and (d), substrates possess properties like silicon, i.e., $E_{s}$ as $190 \mathrm{GPa}, \sigma_{s}$ as 7 GPa [48] and $v_{s}$ as 0.17 [49]. In figure 6(a) and (b), the plastic properties were isotropic for the thin film while its elastic anisotropic ratio varied from 1.0 to 1.5 . At the same time, in figure 6(c) and (d), the plastic anisotropic ratio of the thin film could change from 1.0 to 1.5 as its elastic anisotropic ratio was fixed at 1.0.


Figure 6. Plots illustrating how $P-h$ curves change with transversely anisotropic ratio of $E_{f L} / E_{f T}$ and $\sigma_{f L} / \sigma_{f T}$ on different substrate: (a) and (c) elastic only substrate, (b) and (d) single crystal silicon.

It's very clear that the impact from transversely isotropic elasticity of thin film on the $P$ $h$ curves is very slight regardless the type of substrate materials, as shown in figure 6(a) and (b).

When the value of $E_{f L} / E_{f T}$ grows from 1.0 to 1.5 , the maximum force applied on the indenter only decreased by $0.7 \%$. Even while the substrate is a real material i.e., single crystal silicon, which has a pretty similar property to present system, the impact is still unnoticed. Yet, it can be noticed in figure 6(c) and (d) transversely isotropic plasticity has a significant influence on the results of instrumented indentation, either for pure elastic substrate or single crystal silicon substrate. The force need for a $100 \%$ penetration of indenter increased by about $7.5 \%$ when the transversely isotropic ratio of plasticity, $\sigma_{f L} / \sigma_{f T}$ is raised from 1.0 to 1.5 . Such phenomenon can be understood as a value of 1.5 for $\sigma_{f L} / \sigma_{f T}$ means higher yield strength in the direction of longitudinal, i.e. opposite to the moving direction of the indenter.

By carefully considering the three main purposes of present work, and assess the trade-off between time consumed and the completeness of simulation results, transversely isotropic plasticity would not be considered more in following sections.

### 1.4.2 Dimensionless analysis for current transversely isotropic thin film system

According to the discussion above, any unknown quantity in such system can decided by the variables as: indenter displacement, $h$, half angle of indenter, $\theta$, thin film thickness, $\delta$, together with the mechanical properties of the transversely isotropic system as: $E_{f}, v_{T}, \sigma_{f}, \sigma_{f L} / \sigma_{f T}, n, E_{S}$, $v_{s}$. Moreover, as $v_{\mathrm{T}}$ and $v_{s}$ were set to the value of 0.3 , and $\theta$ were already known in advance, then the unknown quantity, $Y$, can be expressed as:

$$
\begin{equation*}
Y=F\left(h, \delta, E_{f}, \sigma_{f}, \frac{\sigma_{f L}}{\sigma_{f T}}, n, E_{s}\right) \tag{18}
\end{equation*}
$$

Choosing $\delta$ and $\sigma_{f}$ as two quantities with independent dimensions, and applying the $\Pi$ theorem, equation 18 can be modified as:

$$
\begin{equation*}
Y_{\pi}=\Pi\left(\frac{h}{\delta}, \frac{E_{f}}{\sigma_{f}}, \frac{E_{f}}{E_{s}}, \frac{\sigma_{f L}}{\sigma_{f T}}, n\right) \tag{19}
\end{equation*}
$$

where $Y_{\pi}$ is the dimensionless term of $Y$.

Theoretically, there are total 5 undetermined properties of the transversely isotropic thin film system of interest. It's almost impossible to get an analytical solution for such a complicated system with single instrumented indentation as a single $P-h$ might not be enough to figure out so many properties [50]. This was why, conical indenters with three different semi-angle degrees, i.e., $60^{\circ}, 70.3^{\circ}$ and $80^{\circ}$ were used in the present study. Thus, for each thin film system, there were three $P-h$ curves and on each $P-h$ curve, five characteristic responses were collected. They were $P_{1}\left(h_{1}=0.6 \mu \mathrm{~m}\right.$, i.e., penetration ratio at $\left.30 \%\right), P_{2}\left(h_{1}=0.4 \mu \mathrm{~m}\right.$, i.e., penetration ratio at $70 \%$ ), $W_{l}$ (work done by loading, i.e., penetration ratio reached $100 \%$ ), $W_{u}$ (work done by unloading, i.e., penetration ratio reached $100 \%$ ), $S_{\max }$ (the slope at the very initial point of unloading curve, i.e., penetration ratio reached $100 \%$ ). Therefore, a total of 15 characteristic responses needs to be recorded for complete instrumented indentation method. Without a doubt, the penetration ratio, $h / \delta$, in $P_{1}$ and $P_{2}$ can be adjusted as required or for convenience in a practical use. The only restriction here is that penetration ratio should be the same for all three indenters. Since the penetration ratio, $h / \delta$, for each characteristic response was already known before any analysis, equation 19 can be further simplified here as:

$$
\begin{equation*}
Y_{\pi}=\Pi\left(\frac{E_{f}}{\sigma_{f}}, \frac{E_{f}}{E_{s}}, \frac{\sigma_{f L}}{\sigma_{f T}}, n\right) \tag{20}
\end{equation*}
$$

After applying $\Pi$-theorem on all five characteristic responses from a single $P-h$ curve, a group of $\Pi$ functions which would be widely referred later was obtained:

$$
\left\{\begin{array}{l}
\frac{P_{1}}{\sigma_{f} h_{1}^{2}}=\Pi_{i 1}\left(\frac{E_{f}}{\sigma_{f}}, \frac{E_{f}}{E_{s}}, \frac{\sigma_{f L}}{\sigma_{f T}}, n\right)  \tag{21}\\
\frac{P_{2}}{\sigma_{f} h_{2}^{2}}=\Pi_{i 2}\left(\frac{E_{f}}{\sigma_{f}}, \frac{E_{f}}{E_{s}}, \frac{\sigma_{f L}}{\sigma_{f T}}, n\right) \\
\frac{W_{l}}{\sigma_{f} h_{\max }{ }^{3}}=\Pi_{i 3}\left(\frac{E_{f}}{\sigma_{f}}, \frac{E_{f}}{E_{s}}, \frac{\sigma_{f L}}{\sigma_{f T}}, n\right) \\
\frac{W_{u}}{\sigma_{f} h_{\max }{ }^{3}}=\Pi_{i 4}\left(\frac{E_{f}}{\sigma_{f}}, \frac{E_{f}}{E_{s}}, \frac{\sigma_{f L}}{\sigma_{f T}}, n\right) \\
\frac{S_{\max }}{\sigma_{f} h_{\max }}=\Pi_{i 5}\left(\frac{E_{f}}{\sigma_{f}}, \frac{E_{f}}{E_{s}}, \frac{\sigma_{f L}}{\sigma_{f T}}, n\right)
\end{array}\right.
$$

where, subscript $i$ equals to 1,2 or 3 , represents the indenter with half angel of $60^{\circ}, 70.3^{\circ}$ and $80^{\circ}$ respectively.

For efficient and convenient purposes, hereafter, $\frac{P_{1}}{\sigma_{f} h_{1}{ }^{2}}, \frac{P_{2}}{\sigma_{f} h_{2}{ }^{2}}, \frac{W_{l}}{\sigma_{f} h_{\max }{ }^{3}}, \frac{W_{u}}{\sigma_{f} h_{\max }{ }^{3}}$ and $\frac{s_{\max }}{\sigma_{f} h_{\max }}$ are referred to $y_{i 1}, y_{i 2}, y_{i 3}, y_{i 4}$ and $y_{i 5}$, respectively. And the terms of $\frac{E_{f}}{\sigma_{f}}, \frac{E_{f}}{E_{s}}, \frac{\sigma_{f L}}{\sigma_{f T}}, n$ are referred to $x_{1}, x_{2}, x_{3}$ and $x_{4}$, respectively. Thus, equation 21 could be condensed as:

$$
\begin{equation*}
y_{i \mathrm{j}}=\Pi_{i j}\left(x_{1}, x_{2}, x_{3}, x_{4}\right) \tag{22}
\end{equation*}
$$

where subscript $j$ equals a value from 1 to 5 , stands for the 5 characteristic responses of $\frac{P_{1}}{\sigma_{f} h_{1}{ }^{2}}$, $\frac{P_{2}}{\sigma_{f} h_{2}^{2}}, \frac{W_{l}}{\sigma_{f} h_{\max }^{3}}, \frac{W_{u}}{\sigma_{f} h_{\max }^{3}}$ and $\frac{S_{\max }}{\sigma_{f} h_{\max }}$, respectively.

Thus, the major responsibility of present study became into definding a precise expression of $\Pi_{i j}$, which is the basis of forward and reverse analysis.

### 1.4.3 Verification of dimensionless analysis on current system

In equation 21, if $\sigma_{f L} / \sigma_{f T}$ and $n$ are unchanged, characteristic responses, $y_{i j}$, becomes dependent on only the ratios of $\sigma_{f L} / \sigma_{f T}$ and $E_{f L} / E_{f T}$. That is to say, even when the values of $E_{f}$, $\sigma_{f}$ and $E_{s}$ are totally different for systems, characteristic responses, $y_{i j}$, could be totally the same as long as two systems possess equal value for $\sigma_{f L} / \sigma_{f T}$ and $E_{f L} / E_{f T}$.

In table 2, there were 5 transversely isotropic thin film systems. They shared the same value of $\sigma_{f L} / \sigma_{f T}$ and $n$, yet, totally different values for $E_{f}, \sigma_{f}$ and $E_{S}$. Five thin film systems with different property combination were, then, indented by the $70^{\circ}$ conical indenter which leads into 5 distinguishable $P-h$ curves just as shown in figure 7.

Table 2. Five transversely isotropic thin film systems having diverse properties for both thin and substrate.

| System <br> Number | $\boldsymbol{E}_{\boldsymbol{f}} / \mathbf{G P a}$ | $\boldsymbol{\sigma}_{\boldsymbol{f}} / \mathbf{M P a}$ | $\boldsymbol{E}_{\boldsymbol{s}} / \mathbf{G P a}$ | $\frac{\boldsymbol{\sigma}_{\boldsymbol{f} \boldsymbol{L}}}{\boldsymbol{\sigma}_{\boldsymbol{f} \boldsymbol{T}}}$ | $\boldsymbol{n}$ |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 1 | 50 | 200 | 50 | 1.2 | 0.2 |
| 2 | 100 | 400 | 100 | 1.2 | 0.2 |
| 3 | 150 | 600 | 150 | 1.2 | 0.2 |
| 4 | 200 | 800 | 200 | 1.2 | 0.2 |
| 5 | 250 | 1000 | 250 | 1.2 | 0.2 |

The five characteristic responses were collected from each of the curves in figure 7. The $\Pi$ values of characteristic responses, were included in table 3. Obviously, five systems possessed the same values (with the maximum deviation less than $0.03 \%$ ) for all characteristic responses. In summary, dimensionless analysis works well on transversely isotropic thin film systems and can be greatly helpful for forward and reverse analysis.


Figure 7. Five thin film systems with different properties show clearly distinguishable $P-h$ curves.

Table 3. Five transversely isotropic thin film systems with diverse properties for both thin and substrate show the same value for $\Pi_{21}$ to $\Pi_{25}$

| System <br> Number | $\frac{\boldsymbol{P}_{\mathbf{1}}}{\sigma_{f} \boldsymbol{h}_{\mathbf{1}}{ }^{2}}$ | $\frac{\boldsymbol{P}_{\mathbf{2}}}{\sigma_{f} \boldsymbol{h}_{\mathbf{2}}{ }^{2}}$ | $\frac{W_{l}}{\sigma_{f} \boldsymbol{h}_{\max }{ }^{3}}$ | $\frac{W_{u}}{\sigma_{f} \boldsymbol{h}_{\max }{ }^{3}}$ | $\frac{\boldsymbol{S}_{\max }}{\sigma_{f} \boldsymbol{h}_{\max }}$ |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 1 | 140.2778 | 177.6786 | 60.78125 | 13.7875 | 1761.125 |
| 2 | 140.2778 | 177.6658 | 60.78125 | 13.7875 | 1761.125 |
| 3 | 140.2778 | 177.6701 | 60.78125 | 13.78958 | 1760.833 |
| 4 | 140.2778 | 177.6722 | 60.78125 | 13.78906 | 1760.625 |
| 5 | 140.2778 | 177.6735 | 60.78125 | 13.78875 | 1760.75 |

### 1.4.4 Expression creating based on dimensionless analysis

According to figure 2, the forward analysis means to predict the characteristic responses on $P-h$ given that properties of thin film system were already known, and vice-versa for the reverse analysis. But, before the discussing of details about the forward and reverse analysis, it's better to be aware of the truth there would a huge amount of data need to be explained and the range for any single property, elastic or plastic, isotropic or anisotropic, is very wide. Therefore, it could be very hard to find an individual relationship which can describe all related data very well. In order to improve the efficiency and accuracy, grouping, divide the database into smaller clusters, is suggested. Bhat[6] performed grouping by dividing materials into smaller groups according to their property. This method works well but does bring users a little trouble when doing the reverse analysis later. It requires the knowledge of material properties in advance, then a user can decide which group the sample belongs to. And reverse analysis is more likely to be used for practical problems which means, in the process of reverse analysis, people might know nothing about the sample at the very beginning. Certainly, alternative methods could be applied so as to learn something about the samples beforehand, which definitely would increase the cost of both money and time.

In the present study, a change was made based on work of Bhat[6]. All data were divided into smaller groups according to corresponding indentation outcomes (the value of two characteristic responses). A detailed grouping strategy is given in figure 8 . Since there were three indenters, i.e., three $P-h$ curves, for a certain thin film system, data for this system might belong to different a group for different indenters. The critical values of $y_{i 4}$ and $y_{i 5}$ for grouping in present study, i.e., $b_{i 1}$ and $b_{i 2}$, and $c_{i 1}, c_{i 2}$ and $c_{i 3}$, can be found in Appendix A. Of course, the value of $b_{i 1}$ and $b_{i 2}$, and $c_{i 1}, c_{i 2}$ and $c_{i 3}$ could be adjusted so as to serve the analysis better.

In order to distinguish $y_{i j}$ or $\Pi_{i j}\left(x_{1}, x_{2}, x_{3}, x_{4}\right)$ after grouping and highlight its group number, the number of digits of subscript was increased from two, $i j$, to three, $i j k$, where $k$ can take the value from 1 to 6 , standing for the number of groups, as $i$ and $j$ with unchanged meanings.


Figure 8. Schemitcs illustrating the strategy of grouping according to the value of $y_{i 4}$ and $y_{i 5} . b_{i 1}$, $b_{i 2}$ and $b_{i 3}$, and $c_{i 1}, c_{i 2}$ and $c_{i 3}$ are critical values of $y_{i 4}$ and $y_{i 5}$, respectively, where $i=1,2$ or 3 , presents the indenter with $60^{\circ}, 70.3^{\circ}$ and $80^{\circ}$, respectively.

Technically, there are various methods and types of function that could be used for fitting $y_{i j k}$, with $x_{1}, x_{2}, x_{3}$ and $x_{4}$, so as to relate the properties were input for simulation and the calculation outcomes from the 13,500 models. $y_{i j k}$ may or may not share the same expression style for different characteristic responses. To make the analysis here more general, only polynomial expression were applied and all characteristic responses were required to share the same form of polynomial expression, as:

$$
\begin{equation*}
y_{i j k}=\sum_{l} a_{i j k l} x_{1}^{m} x_{2}^{n} x_{3}^{p} x_{4}^{q} \tag{23}
\end{equation*}
$$

where $a_{i j k l}$ is the parameter in front of any term of polynomial, as subscript ' $l$ ' represents the sequence number of the terms in the polynomial. $m, n, p$ and $q$ are exponents.

In order to find the appropriate expression of $y_{i j k}$ and control the degrees of polynomial function so as to control the scale of polynomial, some limitation must be acted on the value of $m$, $n, p$ and $q$, although most of the time, larger the polynomial scale, better the fitting results. Two limits are as:

$$
\begin{gather*}
|m|,|n|,|p|,|q| \leq 3 \\
|m|+|n|+|p|+|q| \leq 4 \tag{24}
\end{gather*}
$$

In actual operation, one term might be removed as it cannot contribute to improving the fitting results at all. At last, there were 75 terms left. The specific expression of $y_{i j k}$, together with the tables for $a_{i j k l}$ were included in Appendix B.

In table 4, the coefficient of determinations, $R^{2}$, for all $y_{i j k}$, were included. All the values of the coefficient of determination, $R^{2}$, are almost equal to 1 for all characteristic responses, $y_{i j k}$, indicated good fitting results in all groups with all indenters, which verifies the strategies of creating polynomial and grouping.

After the determination of both expression and parameters in the polynomial which could be used to describe the relationship between thin film system properties and characteristic responses on $P-h$ curves, forward and reverse analysis now can be performed based on it.

Table 4. Coefficient of determination, $R^{2}$, for fitting $y_{i j k}$ with $x_{1}, x_{2}, x_{3}$ and $x_{4}$.

| Group \# | $\boldsymbol{y}_{\mathbf{1 1}}$ | $\boldsymbol{y}_{\mathbf{1 2}}$ | $\boldsymbol{y}_{\mathbf{1 3}}$ | $\boldsymbol{y}_{\mathbf{1 4}}$ | $\boldsymbol{y}_{\mathbf{1 5}}$ |
| :---: | :---: | :---: | :---: | :---: | :---: |
| $\mathbf{1}$ | 1 | 1 | 1 | 0.9999 | 0.9999 |
| $\mathbf{2}$ | 1 | 1 | 1 | 0.9999 | 1 |
| $\mathbf{3}$ | 1 | 1 | 1 | 1 | 1 |
| $\mathbf{4}$ | 1 | 1 | 1 | 1 | 1 |
| $\mathbf{5}$ | 1 | 1 | 1 | 1 | 1 |


| Group \# | $y_{21}$ | $y_{22}$ | $y_{23}$ | $y_{24}$ | $y_{25}$ |
| :---: | :---: | :---: | :---: | :---: | :---: |
| $\mathbf{1}$ | 0.9999 | 1 | 1 | 0.9999 | 1 |
| $\mathbf{2}$ | 0.9999 | 1 | 1 | 1 | 1 |
| $\mathbf{3}$ | 1 | 1 | 1 | 0.9999 | 1 |
| $\mathbf{4}$ | 1 | 1 | 1 | 1 | 1 |
| $\mathbf{5}$ | 1 | 1 | 1 | 1 | 1 |
| $\mathbf{6}$ | 1 | $y_{32}$ | $y_{33}$ | $y_{34}$ | $y_{35}$ |
| $\mathbf{G r o u p} \#$ | $y_{31}$ | 0.9999 | 0.999 | 0.9999 | 0.9998 |
| $\mathbf{1}$ | 0.9999 | 1 | 1 | 1 | 0.9999 |
| $\mathbf{2}$ | 1 | 0.9999 | 0.9999 | 0.9999 | 1 |
| $\mathbf{3}$ | 1 | 1 | 1 | 1 | 1 |
| $\mathbf{4}$ | 1 | 1 | 1 | 1 | 1 |
| $\mathbf{5}$ | 1 | 1 | 1 | 1 |  |

### 1.4.5 Forward analysis for current transversely isotropic thin film system

A combination of total 100 simulation samples was used to verify the forward and reverse analysis results for present studies. 70 samples are with new system properties which are not included in the 13,500 models. The simulation outcomes of these 70 samples were then compared to the calculation results through equation 23 with suitable values of $a_{i j k l}$. The left 30 were selected from the 13,500 models so as to see if these two groups of test samples show a significant difference in accuracy. Complete information about system properties for all 100 test samples can be found in appendix C .

The strategy of the forward analysis is quite straightforward now and is shown in figure 9 . As mentioned above, the flow path is independent for three indenters, which means the process needs to be carried for three indenters independently.


Figure 9. Sehcmetic illustrating forward analysis flow path.
In table 5, a comprehensive error information (in percentage) of the forward analysis of all $y_{i j k}$ was included. The average errors for the predictions all 15 characteristic responses ( 5 for each
indenter) are fairly low, less than $0.77 \%$. Even the maximum errors of 100 test samples are all below $5.88 \%$ which is acceptable when considering such a complicated thin film system. As the standard deviation values are also pretty low for all cases, it's clear that most samples have comparable error level. That is to say, the expression obtained in section 1.4.4 is stable and precise in relating thin film system properties and characteristic responses on $P-h$ curves. $y_{i 4}$, in another word, $S_{\max }$, the slope of initial unloading curve, exposes the highest error level for all three indenters which might due to the method how this value extracted from $P-h$ curves or an unstable status at the very beginning of unloading simulations. Moreover, it can be found that predicting of $y_{i j}$ for smaller half angle indenter is more accurate than for larger one.

Table 5. Percentage errors of forward analysis results for all $y_{i j k}$.

| $\boldsymbol{\theta = \mathbf { 6 0 } ^ { \circ }}$ | $\boldsymbol{y}_{\mathbf{1 1}}$ | $\boldsymbol{y}_{\mathbf{1 2}}$ | $\boldsymbol{y}_{\mathbf{1 3}}$ | $\boldsymbol{y}_{\mathbf{1 4}}$ | $\boldsymbol{y}_{\mathbf{1 5}}$ |
| :---: | :---: | :---: | :---: | :---: | :---: |
| Maximum <br> error | $1.22 \%$ | $1.27 \%$ | $1.33 \%$ | $5.88 \%$ | $1.97 \%$ |
| Minimum <br> error | $0.00 \%$ | $0.00 \%$ | $0.00 \%$ | $0.00 \%$ | $0.01 \%$ |
| Average <br> error | $0.20 \%$ | $0.12 \%$ | $0.12 \%$ | $0.77 \%$ | $0.26 \%$ |
| Standard <br> deviation | $0.21 \%$ | $0.15 \%$ | $0.16 \%$ | $1.06 \%$ | $0.30 \%$ |
| $\boldsymbol{\theta}=\mathbf{7 0 . 3} \mathbf{3}^{\circ}$ | $y_{21}$ | $y_{22}$ | $y_{23}$ | $y_{24}$ | $y_{25}$ |
| Maximum <br> error | $1.45 \%$ | $1.67 \%$ | $1.74 \%$ | $5.13 \%$ | $1.17 \%$ |
| Minimum <br> error | $0.00 \%$ | $0.01 \%$ | $0.00 \%$ | $0.01 \%$ | $0.00 \%$ |


| Average error | 0.28\% | 0.17\% | 0.15\% | 0.55\% | 0.19\% |
| :---: | :---: | :---: | :---: | :---: | :---: |
| Standard deviation | 0.29\% | 0.20\% | 0.20\% | 0.87\% | 0.20\% |
| $\boldsymbol{\theta}=80^{\circ}$ | $y_{31}$ | $y_{32}$ | $y_{33}$ | $y_{34}$ | $y_{35}$ |
| Maximum error | 2.49\% | 4.33\% | 4.21\% | 4.55\% | 2.79\% |
| Minimum error | 0.00\% | 0.00\% | 0.00\% | 0.00\% | 0.01\% |
| Average error | 0.27\% | 0.39\% | 0.38\% | 0.57\% | 0.30\% |
| Standard deviation | 0.35\% | 0.63\% | 0.62\% | 0.83\% | 0.39\% |

In summary, the forward analysis of the present study is quite successful on transversely isotropic thin film systems.

### 1.4.6 Reverse analysis for current transversely isotropic thin film system

There are various strategies of reverse analysis. Most of them emphasized on finding thin film system properties which give global minimum mismatch between $f_{i j k}$ and $y_{i j k}$, where the values of $y_{i j k}$ are already known in advance (usually from experiments), and values of $f_{i j k}$ are obtained through polynomial functions been created in the present study. However, few discussed much about how they achieved this objective.

In the present work, given there are 15 polynomial functions, each with 75 terms, that need to be analyzed at the same time, Nelder-Mead Simplex Algorithm was applied to help to find the property combinations with local minimum error, instead of global minimum. Since the obtained polynomial is expected to be close to the analytical solution, blinding pursuit of global minimum is not necessary and inefficient, sometimes may even result in the missing of correct answers. As all solutions with local minimum error have a chance to be the final solution, it's better to find them all. Moreover, this tactic could be helpful when checking the uniqueness of reverse analysis results. A complete reverse analysis flow chart is shown in figure 10.

In table 6, the detailed reverse analysis results were shown for test sample \#19. There were total three distinguishable solutions. Apparently, in output II and III, there is some property value exceeding the range greatly, based on which, all analysis was performed, e.g., $\sigma_{f L} / \sigma_{f T}$ for output II, $E_{f}, \sigma_{f L} / \sigma_{f T}$ and $n$ for output III. Therefore, output II and III would be eliminated automatically by the reverse analysis steps.

Table 6. Detailed reverse analysis results for test sample with number 19.

| \#19 Sample | $E_{f} / G P a$ | $\sigma_{f} / M P a$ | $E_{s} / G P a$ | $\sigma_{f L} / \sigma_{f T}$ | $n$ |
| :---: | :---: | :---: | :---: | :---: | :---: |
| Input | 225 | 300 | 75 | 1.35 | 0.05 |
| Output I | 241 | 301 | 75 | 1.45 | 0.04 |
| Output II | 167 | 353 | 77 | 0.57 | 0.05 |
| Output III | 407 | 437 | 71 | 0.71 | -0.01 |



Figure 10. Sehcmetic illustrating reverse analysis flow path.
Obviously, according to the discussions above, the reverse analysis strategy in the present work can save multiple solutions if they do exist. But, by following the steps in the reverse analysis flow chart, all 100 test samples give only one solution at last, that's why the lower right corner part of figure 10 was surrounded by dash lines, which might indicate a very good uniqueness of the relationship between transversely isotropic thin film system properties and characteristic responses on $P-h$ curves, at least in property range of interesting here. The detailed output of all 100 test samples are included in Appendix D. Statistics for all 100 test samples are shown in table 7 and 8. A more intuitive view can be found in figure 11. Evidently, points closer to x axis in figure 11 represent a better predicting result of corresponding property. Most points fall between two dash lines representing positive $10 \%$ and negative $10 \%$ error respectively, indicated acceptable results of the current reverse analysis method.

Table 7. Statistics absolute errors of reverse analysis results for 100 test samples.

|  | $E_{f} / G P a$ | $\sigma_{f} / M P a$ | $E_{s} / G P a$ | $\sigma_{f L} / \sigma_{f T}$ | $n$ |
| :---: | :---: | :---: | :---: | :---: | :---: |
| Maximum <br> error | 19 | 277 | 2 | 0.53 | 0.09 |
| Minimum <br> error | 0 | 0 | 0 | 0 | 0 |
| Average <br> error | 5 | 8 | 1 | 0.07 | 0.01 |
| Standard <br> deviation | 4 | 42 | 0 | 0.09 | 0.01 |

Table 8. Statistics percentage errors of reverse analysis results for 100 test samples.

|  | $\boldsymbol{E}_{\boldsymbol{f}}$ | $\boldsymbol{\sigma}_{\boldsymbol{f}}$ | $\boldsymbol{E}_{\boldsymbol{s}}$ | $\boldsymbol{\sigma}_{\boldsymbol{f L}} / \boldsymbol{\sigma}_{\boldsymbol{f} \boldsymbol{T}}$ | $\boldsymbol{n}$ |
| :---: | :---: | :---: | :---: | :---: | :---: |
| Maximum <br> error | $19.29 \%$ | $29.14 \%$ | $1.70 \%$ | $45.71 \%$ | $72.99 \%$ |
| Minimum <br> error | $0.01 \%$ | $0.01 \%$ | $0.00 \%$ | $0.32 \%$ | $0.05 \%$ |
| Average <br> error | $2.63 \%$ | $1.91 \%$ | $0.33 \%$ | $6.22 \%$ | $2.94 \%$ |
| Standard <br> deviation | $2.83 \%$ | $6.10 \%$ | $0.36 \%$ | $7.41 \%$ | $9.16 \%$ |



Figure 11. Percentage and absolute errors of reverse analysis results of five properties of all 100 test samples: (a) $E_{f}$; (b) $\sigma_{f}$; (c) $E_{s}$; (d) $\sigma_{f L} / \sigma_{f T}$; (e) $n$.

It's also clear from table 8, the average error level for all test samples is below $6.22 \%$. For $E_{s}$, the average percentage error is even as small as $0.33 \%$ with the average absolute error less than 1 Gpa (Table 7) verified the ability of instrumented indentation method of detecting the properties of the substrate, although indenter would not interact with the substrate directly. The high precision of prediction of $E_{s}$ might be due to the deep penetration ratio in the present study. Although in table 8 , it shows that the maximum percentage error of $n$ is about $73 \%$, it's very clear that all points in figure 11(e) are quite close to the x axis. Such contradiction can be explained with the original input and output data. This phenomenon happened on sample \#33, in which, the input value of $n$ was 0.05 , and the reverse output value is 0.0135 . Even though the absolute error is as small as
0.0365 , due to the extremely low input value, the percentage error seems quite significant. The forecasting of anisotropic ratio, $\sigma_{f L} / \sigma_{f T}$, is good for most testing samples, with only 7 of them falling outside $\pm 20 \%$ dash lines. Since the indentations were only, and could only be carried out in the out-of-plane direction, results are still acceptable.

### 1.4.7 Sensitivity of reverse analysis in present study

Artificial $\pm 1 \%, \pm 3 \%$ and $\pm 5 \%$ input error had been imposed to each characteristic response on $P-h$ curves to check the sensitivity of the reverse analysis in the present study. Figure 12 (a) to (c) displays how input errors affect the predicting of $E_{f}$. Figure 12 (d) to (f) shows the influence of input errors on $\sigma_{f}$. Figure $12(\mathrm{~g})$ to (i) shows the influence of input errors on property $E_{s}$. Figure 12 (j) to (l) shows the influence of input errors on $\sigma_{f L} / \sigma_{f T}$, while (m) to (o) shows the influence of input errors on $n$.

Obviously, when input error is less than $\pm 1 \%$, reverse analysis outcomes are still very good, which means reverse analysis is still reliable at this stage. When input errors increase to $\pm 3 \%$, the average error of reverse analysis grows by about $10 \%$. As continued increasing the input error to $\pm 5 \%$, the error of the prediction of $\sigma_{f}$ reaches a level of $40 \%$, while the result of predicting $E_{f}$ and $E_{S}$ is still acceptable. $\sigma_{f L} / \sigma_{f T}$ is most sensitive to input errors while $E_{S}$ is almost not sensitive to input errors. Figure $12(\mathrm{~g})$ to (i) shows that even when input errors are as large as $\pm 5 \%$, the reverse analysis error of $E_{S}$ is always less than $8 \%$. Meanwhile, it's very clear that $Y_{i 5}$, i.e. $S_{\text {max }}$, has the most dominant influence on the predicting of all properties. Forces responses, $Y_{i 1}$ and $Y_{i 2}$ have stronger effect than work responses $Y_{i 3}$ and $Y_{i 4}$. Additionally, responses of indenter $80^{\circ}$ can affect the reverse analysis more intensively than indenter $60^{\circ}$ and $70.3^{\circ}$


Figure 12. Sensitivity of system properties for the reverse analysis in present study. (a)-(c) $E_{f}$; (d)(f) $\sigma_{f}$; (g)-(i) $E_{S}$; (j)-(l) $\sigma_{f L} / \sigma_{f T}$; (m)-(o) $n$.

### 1.5 Conclusions

Due to the increasing applications of instrumented indentation method for solving complex and practical problems, here is a strong motivation for improving its scope, from bulk to thin film, from isotropic to anisotropic. Due to the complexity of stress status under indenter tips and operability of experiments, numerous studies have focused on developing numerical models. Due to the computational cost considerations, many of the numerical studies reported thus far have been able to model only material systems with small number of properties. Little information about the robustness, sensitivity or uniqueness of instrumented indentation method on material systems with large number of properties was at present available. Hence, present study focused on developing numerical models that capture the $P-h$ curves for transversely isotropic material systems with large number of properties. Optimized forward and reverse analysis methods were carried out on explaining simulation results. The determination of substrate properties through instrumented indentation method was also discussed. The main conclusions obtained from the present study are given below.

1. The influence from transversely isotropic elasticity of thin film on $P-h$ curves can be ignored.
2. Dimensionless relationships between a large number of material properties and a larger number of characteristic responses on $P-h$ curves were created. Forward and reverse analysis were carried out based on above relationships. The accuracy of both forward and reverse analysis are acceptable.
3. Reverse analysis results are still acceptable as input error is less than $\pm 3 \%$.
4. Unique solution was found for all 100 test samples indicate a possible uniqueness of instrumented indentation method when material properties were restricted in the range of interesting.
5. Instrumented indentation method can be applied to predicate thin film properties precisely if penetration ratio is high enough.

# Chapter 2: Computational Modeling of Flow Through Porous Media 

### 2.1 Introduction

Porous media are widely found in nature word and modern industry [51-54]. Understanding flow characteristics through porous media is important for several practical applications from water filtration [55, 56], and chemical separation [57] to heat exchangers [58-60] and biological systems [61-63]. For example, within the context of heat exchangers, it is well known that the porous regenerator, is one of the most essential and important parts in Sterling cycle machines and cogeneration systems. Therefore, considerable research work has been done in order to understand the pressure losses and heat transfer characteristics of the regenerator which directly impact the efficiency of the heat engine [54, 64-68].

The nature of the material used and the geometry of the porosity present in the regenerator play important roles in determining the efficiency with which heat is transferred from the working fluid to the regenerator and the vice-versa [54]. Experimental, theoretical and numerical studies have been carried out on a variety of porous regenerators such as metal-felt matrix, sponge metal and fiber woven screen matrices in order to identify a regenerator that would provide the ideal combination of characteristics, i.e., maximizing heat transfer while minimizing pressure losses, as a working fluid flows across the porous medium. Kays and London [54] emphasized that for lowdensity fluid, like gas, the mechanical energy expended in overcoming friction power could have the same magnitude as the transferred heat. However, it's well known the mechanical energy is more valuable than it's in heat. Thus, within the context of Stirling engines with the gaseous fluid, there is a strong motivation to understand the pressure drop characteristics as a working gas flows through the porous media.

A porous medium comprised of a fiber mesh in a woven screen format, hereafter referred to as a woven medium, is the most popular configuration of the Sterling engine regenerator [69]. A single layer of woven screen or woven matrix is as shown in figure 13. It consists of two groups of fibers which are perpendicular to each other, and are interwoven, namely the warp and weft, respectively [70]. The approximately square-shaped flow channel that is surrounded by fibers is
referred to an "open-pore" as shown in figure 13, while the entire structure (that is comprised of one set of warp and weft fibers) is designated as a single-layered, woven matrix with a $4 \times 4$ openpores porous structure.


Figure 13. Schematic illustrating a $4 \times 4$ open porous section in a single layer woven matrix structure where $d, \mathrm{w}$ and t represent the fiber diameter, open pore size and the layer thickness, respectively, where $t=2 d$.

Fibers are not necessarily well arranged in a porous medium. A porous medium with randomly stacked fibers, as shown in figure 14 , hereafter referred to as a random stacked medium, is also of interest and were carefully studied in present work.


Figure 14. Schematic illustrating a porous medium (a random stacked medium) configure with randomly stacked fibers.

Although several studies have focused on developing numerical models to characterize the fluid flow behavior through woven networks or screens, [59, 71-73], might due to the limitations of computational capability or the difficulties of creating large scale models, many of these models have modeled fluid flow across small cross-sections (usually no more than a $3 \times 3$ open porous structure) of the woven network and over relatively shallow depths (usually no more than 6 layers). Given that the mechanics of fluid flow across a woven matrix structure tends to be fairly complex, it is unclear if the relatively small numerical models developed thus far, accurately capture the characteristics of fluid flow in woven matrix structures. Hence, the main objectives of the present study are:
(i) To develop numerical models that will capture the fluid flow characteristics through a porous medium over large physical size;
(ii) To assess the effect of defects in the lay-up of woven networks on the fluid flow characteristics;
(iii) To compare the pressure loss between woven medium and random stacked medium as
gas flows through them;
(iv) To compare the results of the numerical models with experiments;
(v) To optimize the configures of a porous medium in order to reduce the pressure loss.

The structure of this chapter is organized as follows. Background information and a summary of prior work done in predicting important characteristics of flow through a porous media such as pressure drops and friction factors are presented in Section 2.2. The experimental set-up used to characterize flow through woven matrix porous media is presented in Section 2.3 The details of the numerical model developed in the present study are highlighted in Section 2.4. The results obtained from the present study are discussed in Section 2.5 and key conclusions from the present work are summarized in Section 2.6.

### 2.2 Background - Pressure drops and friction factors in porous media

Cauchy momentum equation (equation 25) is a vector partial differential equation describing the micro mechanical balance of viscous fluid [74], where $\rho$ is the density of the fluid, $t$ is the time, $\boldsymbol{u}$ is the velocity vector of fluid, $p$ is the pressure, $\boldsymbol{I}$ is the identity matrix, $\boldsymbol{\tau}$ is the shear stress tensor, $\boldsymbol{g}$ represents the body force.

$$
\begin{equation*}
\frac{\partial}{\partial t}(\rho \boldsymbol{u})+\nabla \cdot(\rho \boldsymbol{u} \times \boldsymbol{u})=-\nabla p \boldsymbol{I}+\nabla \boldsymbol{\tau}+\rho \boldsymbol{g} \tag{25}
\end{equation*}
$$

Stokes's stress constitutive equation (equation 26) defined the relationship between the shear stress, $\boldsymbol{\tau}$, dynamic viscosity $\mu$ and velocity gradient of fluid when the flowing fluid is incompressible [75].

$$
\begin{equation*}
\boldsymbol{\tau}=\mu\left(\nabla \boldsymbol{u}+\nabla \boldsymbol{u}^{T}\right) \tag{26}
\end{equation*}
$$

Now, by joining equation 25 and equation 26, it's not difficult to obtain the very famous Navier-Stokes equation which controlled the behavior of incompressible Newtonian fluid as following [76]:

$$
\begin{equation*}
\frac{\partial \boldsymbol{u}}{\partial \mathrm{t}}+(\boldsymbol{u} \cdot \nabla) \boldsymbol{u}-\frac{\mu}{\rho} \nabla^{2} \boldsymbol{u}=-\frac{1}{\rho} \nabla p+\boldsymbol{g} \tag{27}
\end{equation*}
$$

In addition to equation 27, continuity equation for the incompressible fluid (equation 28) is also required to attain an analytical solution finally [75]. However, the analytical solution is only found when boundary and initial conditions are extremely simple. In other words, it's almost impossible to get an analytical solution for gas or liquid flowing through porous media.

$$
\begin{equation*}
\nabla \cdot \boldsymbol{u}=0 \tag{28}
\end{equation*}
$$

Therefore, plentiful experimental studies have been carried out initially. Kozeny assumed that the bed filled with fine powders is equivalent to a group of parallel and equal-sized channels and firstly developed the following equation of describing the relationship between the viscosity of the fluid, $\mu$, the frontal velocity of fluid, $u_{i n}$, the thickness of the porous media in the direction of fluid flow, $L$, the porosity of the media, $\beta$, specific surface of solid, $S_{v}$, and the pressure loss as fluid flowing through the porous media, $\Delta P$, at low flow rates [77].

$$
\begin{equation*}
S_{v}=\left(\frac{1}{5} \frac{\Delta P}{\mu u_{i n}} \frac{1}{L} \frac{\beta^{3}}{(1-\beta)^{2}}\right)^{1 / 2} \tag{29}
\end{equation*}
$$

Following Stanton and Pannell's [78] approach towards understanding the characteristics of fluid flow through a pipe, Blake [79] analyzed the problem of fluid flow through a porous medium comprised of particulates and obtained a relationship between two dimensionless groups: $\frac{\Delta P}{\rho u_{i n}^{2}} \frac{D_{p}}{L} \frac{\beta^{3}}{1-\beta}$ and $\frac{D_{p} \rho u_{i n}}{\mu(1-\beta)}$, where $D_{p}$ is the diameter of the solid particle within the porous media. Chilton [80] later proposed equation 30, also known as a correlation, to calculate the pressure loss, $\Delta P$, as the fluid flowing through packed columns at high flow rate and introduced the definition of friction factor, $C_{f}$.

$$
\begin{equation*}
\Delta P=2 C_{f}^{\prime \prime \prime} \rho u_{i n}{ }^{2} / D_{c} \tag{30}
\end{equation*}
$$

where $D_{c}$ is the diameter of the column and $C_{f}^{\prime \prime \prime}$ depends on Reynolds number Re. Leva and Grummer [81] further modified equation 30 by including the effect of porosity of the media of the factor of $\frac{\beta^{m}}{(1-\beta)^{2}}$, where $m$ is either 1 or 2 in value. Based on former work and experimental data, Ergun postulated that the pressure drop observed as the fluid flows through a porous medium was due to both kinetic and viscous effects [82] and proposed an important and comprehensive relationship as given below in Equation 31 which is suggested to work at both low and high flow rate:

$$
\begin{equation*}
\Delta P / L=150 \frac{(1-\beta)^{2}}{\beta^{3}} \frac{\mu \bar{u}}{D_{p}^{2}}+1.75 \frac{1-\beta}{\beta^{3}} \frac{G \bar{u}}{D_{p}} \tag{31}
\end{equation*}
$$

where, $\bar{u}$ stands for the superficial velocity that is obtained by considering the mean pressure of the fluid at the entrance and the exit regions, and $G$ is the mass flow rate. The differential form of the Ergun's Equation (equation 31) is given below in Equation 32 as:

$$
\begin{equation*}
d P / d L=2 A \frac{(1-\beta)^{2}}{\beta^{3}} \mu u_{i n} S_{v}^{2}+\frac{B}{8} \frac{1-\beta}{\beta^{3}} \rho u_{i n}^{2} S_{v} \tag{32}
\end{equation*}
$$

where, A and B are coefficients that are determined experimentally and $S_{v}$ represents the specific surface area of the solid particles, i.e., the ratio of the surface area exposed to the flowing fluid to
the volume of the solid particles [58]. Ergun also derived an expression for the friction factor, $C_{f}^{\prime \prime}$, as the ratio of the total energy loss to kinetic energy loss which is analogous to the Darcy friction factor [82]:

$$
\begin{equation*}
C_{f}^{\prime \prime}=\frac{150(1-\beta)}{R e^{\prime \prime}}+1.75 \tag{33}
\end{equation*}
$$

where the Reynold number, $R e^{\prime \prime}$, is defined as:

$$
\begin{equation*}
R e^{\prime \prime}=\frac{\rho}{\mu} u_{i n} D_{p} \tag{34}
\end{equation*}
$$

Sodré and Parise [69] applied Ergun's law [82] of pressure losses for a porous medium system that is comprised of an annular bed of woven screens and proposed that the modified friction factor (not considering the boundary wall effect), $C_{f}^{\prime}$, could be defined as:

$$
\begin{equation*}
C_{f}^{\prime}=\frac{a_{1}(1-\beta)}{R_{e}^{\prime}}+a_{2} \tag{35}
\end{equation*}
$$

where coefficients $a_{1}$ and $a_{2}$ were identified, respectively, as 100 and 0.73 . In equation $35, R_{e}^{\prime}$ is the Reynolds number for porous media made up of woven screens, which is based on the wire diameter, $d$ :

$$
\begin{equation*}
R e^{\prime}=\frac{\rho}{\mu} u_{i n} d \tag{36}
\end{equation*}
$$

The pressure drop that occurs as the fluid flows through the porous media is then given in following correlation as:

$$
\begin{equation*}
\Delta P=\frac{C_{f}^{\prime} \rho L u_{i n}^{2}}{d} \frac{(1-\beta)}{\beta^{3}} \tag{37}
\end{equation*}
$$

According to Kays and London's [54] and Gedeon's [83] research, the prediction of pressure drops in fluid flow across a porous medium could be further simplified by using another definition of the Reynolds number using the hydraulic diameter as given in equation 38:

$$
\begin{equation*}
R e=\frac{\rho}{\mu} u_{m} d_{h} \tag{38}
\end{equation*}
$$

In equation 38, the mean flow velocity $u_{m}$ is obtain by dividing the frontal velocity $u_{i n}$ by porosity $\beta$ and $d_{h}$ stands for the hydraulic diameter [84-86] which is defined as:

$$
\begin{equation*}
d_{h}=\frac{4 \beta}{S_{v}(1-\beta)} \tag{39}
\end{equation*}
$$

For a woven medium made up of fibers with a circular cross-section, $S_{v}$ is equal to $\frac{4}{d}$. By substituting $D_{p}$ with $d$ and $\bar{u}$ with $u_{i n}$, and rearranging equation 31 , the pressure losses expected across the porous media can be obtained using equation 40 as:

$$
\begin{equation*}
\Delta P=C_{f} \frac{\rho}{2} \frac{L}{d_{h}} u_{m}^{2} \tag{40}
\end{equation*}
$$

Equation 40 has also been widely accepted and adopted in recent work in the field [59, 87]. Thus, it is evident from equation 40 that it would be easy to predict the pressure losses if the friction coefficient $C_{f}$ is known a priori. In general, the friction factor $C_{f}$ is primarily a function of the Reynolds number $R e$. The exact form and specific coefficients in the expression for $C_{f}$ mainly depend on the porosity geometry of the porous media [58, 59, 65, 83]. Essentially, if $C_{f}$ in equation 40 has the same form as in equation 35 i.e., with two parameters in the expression, it can be found that equations 37 and equation 40 will also have the same form but might be with slight differences in the coefficients.

Several forms have been identified for the friction coefficient $C_{f}$ in previous studies. Jones [67] presented an expression for the friction factor $C_{f}$ in the form $a_{1} R e^{m}$. In the specific regime of laminar flow, $m=-1$, i.e., $C_{f}=a_{1} / R_{e}$. Bernd [88] proposed that there are two components to the pressure drops in a woven matrix medium, form drag, $\Delta P_{f d}$, and skin friction, $\Delta P_{s f}$, such that the total pressure drop $\Delta P$ is given as:

$$
\begin{equation*}
\Delta P=\Delta P_{f d}+\Delta P_{s f} \tag{41}
\end{equation*}
$$

where:

$$
\left\{\begin{array}{l}
\Delta P_{f d}=C_{f d} \frac{\rho}{2} u_{m}^{2}  \tag{42}\\
\Delta P_{s f}=C_{s f} \frac{\rho}{2} u_{m}^{2}
\end{array}\right.
$$

Bernd [89] further reasoned that since $\Delta P_{f d}$ is not a boundary layer phenomenon, $C_{f d}$ should not be a function of the Reynolds number while $C_{s f}$ is proportional to $1 / R e$. Also, Bernd [89] (and Tanaka [84]) thus suggested the following form for the friction coefficient $C_{f}$ :

$$
\begin{equation*}
C_{f}=a_{1} / R e+a_{2} \tag{43}
\end{equation*}
$$

Armour and Cannon [65] obtained similar correlations for the friction factor by combining models of flow past submerged spheres and capillary tube-bundles at low and high flow velocity regimes, respectively. Macdonald et al. [66] verified equation 43 with a large database of experimental results available in the literature. Gedeon and Wood [83] later found that equation 43 can better track observed experimental results, esp., for flows with higher Reynolds numbers, by introducing a third parameter and presented a modified version of equation 43 for the friction factor as follows:

$$
\begin{equation*}
C_{f}=a_{1} / R e+a_{2} R e^{a_{3}} \tag{44}
\end{equation*}
$$

In general, the numerical values of the coefficients $-a_{1}, a_{2}$ and $a_{3}$, are affected by the geometry of the porous media (or weave style) [58, 65], wall effects [69, 90], oscillating density and oscillating pressures [67], compressibility of the fluid and the oscillating characteristics of the frontal velocity [68, 84].

Recently, several studies have invoked numerical methods to characterize the friction factor $C_{f}$, quantify the pressure losses and to obtain a mechanistic understanding of pressure loss phenomenon in flow through the porous medium [59, 70-73, 91-96]. Green et al. [73] researched the effects of boundary conditions and geometry of the porosity on flow characteristics in porous media. Costa et al. [59] focused on the wound woven configuration and identified values of $a_{1}, a_{2}$ and $a_{3}$, to predict the pressure loss using the correlation as equation 40. Xueliang et al. [71] created five gradual converging-diverging ducts to understand the flow behavior and found that a woven
matrix with higher porosity and smaller thickness may result in a larger dynamic through-thickness permeability. Ponzio et al. [72] considered a specific screen aspect ratio and two different screen orientations ( $0^{\circ}$ and $45^{\circ}$ ) with respect to the main flow direction.

However, much of the models developed thus far relied on relatively smaller model sizes. Hence, larger models are developed in the present study to obtain a better insight into the flow behavior through woven matrix porous medium together with random stacked porous medium. The details of the experimental set-up used to characterize the flow behavior through woven medium are presented in the following Section 2.3.

### 2.3 Experiment Set-up for Characterizing Flow through Woven Matrix Porous

## Media

The experiments are prepared by Hanfei Chen from the group advised by Professor Longtin, J. P. from the Department of Mechanical. The experimental set-up designed to measure the frontal flow velocity $u_{i n}$ and pressure drop $\Delta P$ across the porous media is shown in figure 15 . The main portion of the test rig is a 1 inch $(2.54 \mathrm{~cm})$ inner diameter PVC pipe. The woven matrix porous media samples, with wire diameter as $56 \mu \mathrm{~m}$, are cut into 2.54 cm diameter cylinders with different lengths (greater than half 1.27 cm ) and placed inside the PVC test section. Nitrogen gas from pressurized cylinders is passed through a regulator (not shown) and then introduced to the PVC test section. The flow is from left to right.
(a)

(b)


Figure 15. (a) Schematic and (b) a photograph of the experimental set-up used for measuring pressure drops in woven matrix porous media.

The experimental pressure loss, $\Delta P_{d}$, across the porous media sample is measured using a high-precision differential pressure sensor (Freescale Semiconductor, MPX4250DP). A $k$-type thermocouple (Omega Engineering) is used to measure the nitrogen gas temperature just upstream of the sample. The absolute gas pressure, $P_{a}$, downstream of the sample is measured with an absolute pressure sensor (Freescale Semiconductor, MPX4205AP), which is required to determine the density of the gas both upstream and downstream of the sample. A PVC throttling valve is used to adjust the flow rate for different tests, and the flow itself is measured using an Omega flowmeter. The details for each sensor are listed in table 9.

Table 9. Sensor Information

| Sensor | Model | Range | Accuracy |
| :---: | :---: | :---: | :---: |
| Differential pressure sensor | MPX4250DP | $0-250 \mathrm{KPa}$ | $\pm 3.45 \mathrm{KPa}$ |
| Absolute pressure sensor | MPX4205AP | $20-250 \mathrm{KPa}$ | $\pm 0.075 \mathrm{KPa}$ |
| Flowmeter | FMA1700A/1800A | $0-500 \mathrm{~L} / \mathrm{min}$ | $\pm 7.5 \mathrm{~L} / \mathrm{min}$ |
| Thermocouple | K-type | $0-1000^{\circ} \mathrm{C}$ | $\pm 1^{\circ} \mathrm{C}$ |
|  |  |  |  |

To minimize density effects, the absolute pressure at the middle point of the sample, $P_{c}$, is maintained at a constant value by adjusting the nitrogen flow with the tank regulator and the back pressure behind the sample with the throttling valve. The pressure at the center of the regenerator sample can be determined by:

$$
\begin{equation*}
P_{c}=P_{a}+\frac{\Delta P_{d}}{2} \tag{45}
\end{equation*}
$$

For example, at lower flow rates, the throttling valve will be slightly closed to ensure that the pressure at the center of the sample remains constant. If this were not done, the absolute pressure of the nitrogen gas would vary considerably as the flow rate is changed. Since the gas properties are pressure-dependent, this would introduce error into the final results. The above procedure is a simple yet effective means to minimize this error source.

A photograph of the completed test assembly is shown in figure 15. All sensor data are recorded using a Keithley Model 2000 high-resolution digital multimeter. The multimeter is connected by Ethernet to a laptop PC running ExcelLINX, which transfers the measured data to Microsoft Excel. Measurements of all sensor data are taken once per second.

### 2.4 Numerical Modeling

### 2.4.1 Rationale for modeling woven networks with square fibers

In general, considerable efforts are needed for creating numerical models of woven fiber networks with circularly shaped fibers as the curved surfaces typically require a careful design of the element mesh to capture the geometrical details of the curved surfaces in a reasonably accurate manner. On the other-hand, developing numerical models of woven networks with square fibers tend to be much simpler as the mesh design for systems with fewer curved surfaces is much less cumbersome. Thirdly, it's difficult to evaluate the accurate porosity of a random stacked model with circular fibers. Especially, for numerical models that capture physically large woven networks, the efforts needed for creating models with flat surfaces is substantially less than those needed for models with curved surfaces.

Hence, the present study is focused on developing an efficient, yet accurate method of building a numerical model of woven networks and random stacked media without curved surfaces, and checking the feasibility of utilizing such a model to understand flow behavior in physically large systems firstly. For woven matrix models, the two main sources of curved surfaces in the numerical model are the circular cross-section of the fiber itself and the bending of the fibers as the fibers cross over each other. The former can be addressed with square cross-section fibers (figure 16(a), (b)), while the latter is addressed by introducing a flat and step-type configuration in the fiber cross-over regions (figure 16(c), (d)). For random stacked media, curved surfaces are only from the exterior appearances of these fibers. That is to say, by simply replacing the circular fibers with square cross-section fibers, all curved surfaces could be eliminated easily. Consequently, establishing that equations 38-44 are applicable to both circular and square cross-section fibers, at least within the Reynolds number range of interest, is required.

Gedeon [87] observed out that if non-circular fibers are all oriented perpendicular to the axial flow direction, then the definition of hydraulic diameter and the friction factor correlations should still hold. Armour and Cannon [65] found that in the low Reynolds number regime, the friction factor, $C_{f}$, is not sensitive to the specific configuration of the woven matrix. In addition, Macdonald et al. [66] argued that the shape and size distribution irregularity of media particles only increased the difficulty of calculating the characteristic length, but did not change the form
of the expression for $C_{f}$. Thus, it is reasonable to expect that the change of cross-sectional shape of fibers will not affect the overall flow characteristics in a significant manner.


Figure 16. Schematics illustrating fibers with circular cross-sections (a) and square cross-sections (b), and $a$ represents the thickness of square crossing sectional fiber. The woven matrix with fiber bending (c) is modeled with a woven matrix with flat fibers with a step-type configuration in the cross-over regions (d).

In principle, fibers with other cross-sectional shapes, like triangular and polygonal, are also candidates for the present study. However, due to its operational convenience, square shaped fibers are selected for this study. The most important geometric factors, $d_{h}$, associated with the fiber shape, $S_{v}$, as defined in equations 39 are equivalent for fibers with circular and square crosssections. Hence, the Reynolds number predicted by equation 38 will be the same for flow through a woven matrix or a random stacked media with square fibers if the fiber diameter, $d$, is replaced with the fiber thickness, $a$, provided the value of porosity, $\beta$, remains unchanged.

Thus, in the present work, it is expected that flow through a woven medium with square fibers will exhibit a relationship between the Reynold number, $R e$, and the frontal flow velocity, $u_{i n}$, that would be the same as in the case of flow through a woven matrix with circular fibers and the same should be true for random stacked media. By rewriting equation 38, as:

$$
\begin{equation*}
R e=\frac{\rho u_{i n}}{\mu} \frac{d_{h}}{\beta} \tag{46}
\end{equation*}
$$

information about the fluid $\left(\frac{\rho u_{i n}}{\mu}\right)$ and the woven media $\left(\frac{d_{h}}{\beta}\right)$ can be explicitly identified. In the present work, while developing the numerical model, care has been taken to ensure that the model accurately captures the geometric parameters of the woven media, i.e., the hydraulic diameter, $d_{h}$, and the porosity, $\beta$, respectively, in a real experimental set-up (and not just the ratio $\frac{d_{h}}{\beta}$ ). To preserve the $80 \%$ porosity of the original circular-wire mesh, the open pore-size in the numerical model is increased to $224 \mu \mathrm{~m}$ over the original open-pore size of $198 \mu \mathrm{~m}$ in the circular-wire mesh (Figure 17).


Figure 17. Optimal micrograph of a woven matrix open porous structure used in the experiments.

### 2.4.2 Numerical models for flow through woven matrix and random stacked media with square shaped fibers

Numerical models are developed to capture the flow characteristics through the porous media. Abaqus CFD version 6.14 was selected as the pre-processor, solver and post-processor. Dry nitrogen was used as the working fluid, as this gas is inexpensive and readily available for the experimental testing reported later.

For woven matrices, square crossing sectional fibers with six different thickness ( $40 \mu \mathrm{~m}$, $56 \mu \mathrm{~m}, 60 \mu \mathrm{~m}, 80 \mu \mathrm{~m}, 110 \mu \mathrm{~m}$ and $140 \mu \mathrm{~m}$ ) were woven into screens with different open size, so as to achieve four distinguished porosity, i.e., $45 \%, 60 \%, 75 \%$ and $80 \%$. Fine woven matrices were, then, placed together to form a regenerator sample following the sequence as shown in figure 18. Alternating layers were rotated 45 degrees relative to the previous layer just as observed in experiments (figure 17), with the axis of rotation located at the geometric center.


Figure 18. Schematics illustrating the spatial orientation and sequence of woven matrices.

For random stacked media, fiber with and only with the thickness $a$ as $56 \mu \mathrm{~m}$ was applied. The sample porosity was fixed at $80 \%$ while the sample thickness was either 8 times or 80 times the thickness of fibers. To be more specific, fibers, in random stacked media, were not completely distributed by random. Fibers could only extend in the plane perpendicular to the direction of flow. And at a certain fiber layer (for differentiating the matrix layer), all fibers were restricted to be parallel to each other (Figure 19). The configures of random stacked medium were generated automatically by Python programs.


Figure 19. Schematics demonstrating a porous media contains eight layers of randomly distributed fibers.

The complete fluid space consists of three parts: upstream, midstream (shaped according to the configure of the porous media), and downstream (Figure 20). The borders between the midstream and the upstream and downstream regions are formed by the top and bottom surfaces of the porous media, respectively. The depth of the fluid in the upstream and downstream sections are set to 5 and 10 times the thickness of the midstream section, to eliminate flow-reversal effects at the inlet and the outlet boundaries [59].


Figure 20. (a, b, c) The geometric features of a numerical model developed in the present study to capture the flow behavior through a $4 \times 4$ open porous section of a woven matrix structure (4-layer model).

The flow behavior through a woven matrix medium was captured by modeling the fluid part as a representative volume (mid-stream part) of the woven matrices that are comprised of a certain number of open porous section in the in-plane (XY plane) direction and a certain number of screen layers in the out-of-plane direction, i.e., through-thickness direction ( Z direction). The specific size of representative volume depended on several factors: fiber thickness, $a$, media porosity, $\beta$, and purpose of the model. Because of the alternative high-low asymmetry of the weave steps, at least four layers ( 180 degrees rotation), i.e., 8 times the thickness of fiber, were required before the pattern repeats. A model with up to 110 matrix layers was created to study the size effects in Z direction (figure 20). The sectional size in XY plane varied from $580 \mu \mathrm{~m} \times 580 \mu \mathrm{~m}$ to $2240 \mu \mathrm{~m} \times 2240 \mu \mathrm{~m}$. A range of flow velocities from $0.06 \mathrm{~m} / \mathrm{s}$ to $40 \mathrm{~m} / \mathrm{s}$ was modeled. The maximum Reynolds number was 262 as the maximum Mach number is 0.117 .

The representative volume of the middle fluid part for the random stacked media models has physical dimensions of $1500 \mu \mathrm{~m} \times 1500 \mu \mathrm{~m}$ in the in-plane direction and from 0.448 to 4.48
mm in the through-thickness direction. The frontal velocities, $u_{i n}$, were set to be $1.08 \mathrm{~m} / \mathrm{s}, 2.35 \mathrm{~m} / \mathrm{s}$, $4.18 \mathrm{~m} / \mathrm{s}$ and $8.27 \mathrm{~m} / \mathrm{s}$ and the corresponding values of Reynold number, $R e$, were $19.8,43.6,78.4$ and 163.3 respectively.

Costa [59] suggested that turbulence should be included in a model when the Reynolds number is larger than 160. However, Dybbs and Edwards [97] suggested that the flow is expected to be laminar for Reynolds numbers between 10 and 175 while the range between 175 and 250 is 'separated' laminar flow. Since most of the simulation cases in the present study cover the laminar flow regime, the gas flow is taken to be a viscous, Newtonian, incompressible flow with laminar flow behavior. Thus, the incompressible Naiver-Stokes equation (equation 27) governs the behavior of nitrogen and are solved based on a hybrid finite-volume/ finite-element method. The viscosity of nitrogen was taken to be constant at $1.74 \times 10^{-5} \mathrm{~Pa} \cdot \mathrm{~s}$, while its density was allowed to vary according to the ideal gas law based on the absolute pressure at the middle point of the experimental sample, $P_{c}$, but still stay unchangeable during the simulation process. The boundary conditions invoked in the numerical models are summarized below (figure 20):

1. At the inlet: velocity boundary, $u_{x}=u_{y}=0 ; u_{z}=u_{i n}$;
2. At the outlet: pressure boundary, $P_{\text {out }}=1 \mathrm{bar}$ or $P_{c}$ (if data available);
3. At the side faces: symmetric boundary conditions are used with the normal velocity and the velocity gradient with each side face being set to zero;
4. At fiber surface: no-slip wall boundary conditions are used for the fluid-fiber interface.

The linear convergence limit was $10^{-5}$ and the convergence-checking frequency was set to two iterations which are default settings in Abaqus. The present study is focused on steady inlet flows. The characteristics associated with oscillating flow will be discussed in a future study.

Tetrahedral elements were used to model the fluid in the present study. Figure 21 presents the details of the mesh generated to capture the fluid part of the simulation. The fluid part of the model was filled with unstructured, but regular mesh. The total number of elements varies from 0.6 M to 16 M , mainly depending on the number of fiber layers modeled.


Figure 21. Schematics illustrating mesh details used to represent the fluid part of the model: (a) a 4-matrix-layer with $4 \times 4$ opens woven matrix model; (b) an 8 -fiber-layer random stacked medium model.

In order to assess the trade-off between accuracy of the numerical model and the size of it, the effect of changing the (fluid) element size from $18 \mu \mathrm{~m}$ (about $1 / 3^{\text {rd }}$ of the fiber thickness) to $6 \mu \mathrm{~m}$ (about $1 / 9^{\text {th }}$ of the fiber thickness) on the pressure drops obtained in the simulations (with a four-layered woven matrix model, fiber thickness, $a$, as $56 \mu \mathrm{~m}$ ) was carried out over a range of inlet flow velocities from $1.08 \mathrm{~m} / \mathrm{s}$, to $12.26 \mathrm{~m} / \mathrm{s}$, i.e., from Reynolds number of 19.8 to 163.3 . As indicated in Figure 22, the pressure drops predicted by the models with larger elements are about $6 \%$ lower than that predicted by the models with smaller elements. However, there is a ten-fold decrease in the size of the model created with the larger elements, thus enabling the development of numerical models that can capture physically large systems at lower computational cost without compromising accuracy in a significant manner. This trend is observed to be valid for flows over a wide range of Reynolds numbers. Thus, element size of about $1 / 3^{\text {rd }}$ of the fiber thickness or diameter were used for all following models would be discussed later.


Figure 22. The pressure drops obtained in the numerical model simulations of a 4 layered, $4 \times 4$ open porous section of a woven matrix where the fluid element size is varied from 6 microns to 18 microns indicate that the pressure drops decrease as the element size decreases. The pressure drop obtained in a model with element size ' i ' $\left(\Delta \mathrm{P}_{\mathrm{i}}\right)$ is normalized by the pressure drop observed in the model with the element size of 6 microns $\left(\Delta \mathrm{P}_{6}\right)$.

Since the numerical model developed in the present work is based on the incompressibility assumption, the density, $\rho$, of the gaseous fluid that flows through the porous medium is considered as a constant. Hence, from equation 32, it can be deduced that the pressure drops expected from the flow of the fluid through the porous medium, should be proportional to the size of the porous media in the direction of fluid flow. Thus, the pressure drops obtained from the model simulations, $\Delta P_{m o}$, for a given thickness of the stacked woven matrices or the random stacked fiber layers, $t_{m o}$, could be scaled linearly with mid-part thickness for comparison of the pressure drops measured in the experiments, $\Delta P_{r e}$ using samples with a thickness of $t_{r e}$ as:

$$
\begin{equation*}
\Delta P_{r e}=\Delta P_{m o} \frac{t_{r e}}{t_{m o}} \tag{47}
\end{equation*}
$$

The corresponding expressions for the friction factors $C_{f}^{\prime}$ and $C_{f}$ are as follows:

$$
\begin{equation*}
C_{f}^{\prime}=\frac{\Delta P_{m o} a}{\rho t_{m o} u_{i n}^{2}} \frac{\beta^{3}}{1-\beta} \tag{48}
\end{equation*}
$$

and

$$
\begin{equation*}
C_{f}=\frac{2 \Delta P_{m o}}{\rho} \frac{d_{h}}{t_{m o}} \frac{1}{u_{m}^{2}} \tag{49}
\end{equation*}
$$

### 2.5 Discussion

### 2.5.1 Comparison of simulation results between circular fiber and square fiber models

To assess whether the flow characteristics captured in models with square fibers are accurate, a series of numerical simulations were also carried out on models with circular fibers with identical weave patterns, matrix thickness, porosity and frontal velocity conditions as on models with square fibers. In these simulations, the diameter of circular fibers was also set to be equal to the thickness of square fibers, i.e., $56 \mu \mathrm{~m}$.


Figure 23. Schematics illustrating two kinds of 4-layer porous media with $4 \times 4$ open woven matrix comprised by circular fibers: (a) and (b) contact model; (c) and (d) compact model.

During the process of creating woven media models with the circular cross-section fibers, there were two strategies on how to deal with the junction between any two matrices, which is rarely deliberated by others. The first is just as shown in figure 23(a) and (b), where top matrix just contacts bottom layer at certain points, hereafter referred to contact model. According to the
calculation[88], with fixed fiber diameter as $56 \mu \mathrm{~m}$, the open size need to be modified to $168 \mu \mathrm{~m}$ to make sure the porosity of the midstream part still as $80 \%$. It's worth noting that, under such restrictions, the matrix thickness is exactly equal to twice the diameter of fibers, i.e., $112 \mu \mathrm{~m}$ which is greater than experimentally measured value, i.e., about $100 \mu \mathrm{~m}$. The second is to compress the mid-stream part at two ends so as to reduce the layer thickness to exact $100 \mu \mathrm{~m}$, which means the bottom boundary of upper layer will enter the top boundary of lower layer and vice versa, which is displayed in figure 23(c) and (d), namely compact model. It would not be difficult to find that open size now is equal to $198 \mu \mathrm{~m}$, consistent with practical measurement on a real sample.


Figure 24. The characteristic relationship between the friction factor $C_{f}$ and the Reynold's number Re as predicted by the 4-layer numerical model simulations of a $4 \times 4$ open porous section of a woven matrix with square fibers and circular fibers compared to results by Costa [59].

As indicated in Figure 24, it is evident that the relationship between the friction factor and the Reynolds number observed in models with square fibers is within $10 \%$ of that obtained with circular fibers. At lower Reynolds numbers, the results obtained from the simulations with square fibers are almost the same as those obtained from compact models, while at higher Reynolds numbers, results tend to approach those from contact models. As well, the modeling simulations (which feature a fiber lay-up that follows a regular $0^{0}, 45^{\circ}, 90^{\circ}$ rotational sequence in the throughthickness direction) agree reasonably well with the predictions of Costa's correlations [59] (which are most accurate for structures with laterally displaced stacked layers).


Figure 25. Plots exhibiting resultant velocity maps of all three models (woven matrix media model, contact model and compact model) at each layer.

Figure 25 exhibited resultant velocity maps of all three models (woven matrix media model, contact model and compact model) at the bottom of each layer, i.e. from the 1 st layer to the 4th layer. Inlet flow velocity, $u_{i n}$, is set to be $4.18 \mathrm{~m} / \mathrm{s}, \operatorname{Re}$ as 78 . The woven matrix media model with square fibers successfully reproduced most flow features throughout the whole fluid part as in models with circular fibers. In other words, matching between simulation results of two circular fiber models and the square fiber model is not a coincide. Overall, it can be concluded that numerical models with square fibers accurately capture the flow characteristics through porous media with circular fibers. Therefore, all numerical models following will rely on square crossing sectional fibers with confidence.

There is another observed phenomenon need to be pointed out before going to next section. The detailed velocities distribution maps of $\mathrm{X}, \mathrm{Y}$ and Z directions, for the two circular fiber models, are demonstrated in figure 26 as well. Inlet flow velocity, $u_{i n}$, at inlet boundary is set to be $2.35 \mathrm{~m} / \mathrm{s}$, i.e. Re is equal to 44 this time. The displayed plane is the middle XY plane of middle stream part, i.e., the bottom plane of the 2nd layer or top plane of the 3rd layer. Clearly, the inplane velocities distribution, i.e. velocities in X and Y directions, of contact model are more intensive comparing compact model. In the meantime, the out-of-plane velocity plots look pretty much the same for both models which explained why compact models tend to slightly underestimate the pressure loss when compared to contact models.


Figure 26. Plots exhibiting velocity distribution maps in all three directions of contact and compact models: (a)-(c) the contact model and (d)-(f) the compact model.

### 2.5.2 Large numerical models for assessing size effects

As demonstrated in Section 2.5.1, results obtained from the numerical models developed in the present study which had a representative volume that comprised of a $4 \times 4$ open porous section in the in-plane direction and four layers of the woven matrix in the through-thickness direction compare well with Costa's results. To assess the sensitivity of the pressure drops predicted by the numerical model developed in the present study to the model size, a systematic study was carried out by varying the model size in the in-plane direction and in the throughthickness direction based on the setting from section 2.5.1.

### 2.5.2.1 Sensitivity to the model size in the in-plane direction

As presented in Figure 27, the model size was increased in the in-plane direction from a 2 x 2 open porous section to a $10 \times 10$ open porous section, while the size in the out-of-plane direction kept unchanged, and a comparison of the pressure drops predicted by the corresponding numerical simulations was made.


Figure 27. The pressure drops obtained in the numerical model simulations of a 4-layer woven structure where the model size is increased from a $2 \times 2$ to a $10 \times 10$ open porous section. The pressure loss obtained in a model with open porous section ' $\mathrm{i} x \mathrm{i}$ ' $\left(\Delta \mathrm{P}_{\mathrm{ixi}}\right)$ is normalized by the pressure loss observed in the model with an open porous section of $10 \times 10\left(\Delta \mathrm{P}_{10 \times 10}\right)$.

In all cases the mid-stream part thickness was maintained constant at four layers. It is observed that changes in the model size, in the in-plane direction have only a very limited influence on the simulation results, especially, for models that have an open structure that is $4 \times 4$ or greater, which is also consistent with Green and Zhishuo's [73] observation. The difference in the results obtained in the model with a $4 \times 4$ open porous section and those obtained for a model with a $10 \times 10$ open porous section is less than $4 \%$. Hence, it is evident that the model that features a $4 \times 4$ open porous section adequately captures the characteristics of flow through a woven matrix structure, while minimizing computational cost.

The significantly larger pressure loss observed in the $2 \times 2$ open porous structure is most likely due to boundary effects. Figure 28 illustrates the velocity distribution across the models for the $2 \times 2,4 \times 4,6 \times 6$ and $8 \times 8$ cases. It can be seen that the velocity distribution observed in the $2 \times 2$ open porous structure is different from that observed in other models. For example, the models with $4 \times 4$ and $8 \times 8$ open porous structures have lower speeds along diagonal directions, but higher speeds at the side borders which may contribute to the lower pressure drops observed in the larger models. Furthermore, it can be readily observed that the velocity patterns observed in the models with a $4 \times 4$ structure is also a part of the velocity patterns observed in the larger models with $6 \times 6$ and $8 \times 8$ open porous structures. Hence, it can be concluded that the models with a $4 \times$ 4 open porous structure adequately capture the flow behavior of porous media with woven matrix structures.

The results observed in the present study are also consistent with the assessment of Mehta and Hawley [90] that when the ratio between the in-plane size of the woven matrix and the fiber diameter increase, boundary become less important and even negligible. In the models developed in the present study, the ratio between the woven matrix size and the fiber size is about 20 for the $4 \times 4$ open porous model and about 40 for the $8 \times 8$ open porous model. Thus, the boundary effects are expected to be reduced as the model size is increased.


Figure 28. The resultant velocity distributions observed at the bottom of midstream part in the finite element models with (a) $2 \times 2$; (b) $4 \times 4$; (c) $6 \times 6$; and (d) $8 \times 8$, open porous sections.

### 2.5.2.2 Sensitivity to the model size in the through-thickness direction

The effect of the model size of out-of-plane direction on the pressure drop was also explored. The thickness of the porous media was progressively increased to assess effects of model thickness on the prediction of pressure losses while size at in-plane directions was fixed.

As shown in Figure 29(a), for a given Reynolds number, i.e., inlet flow velocity, the pressure drops per unit length observed in the thin models are consistently greater than those observed in simulations of thicker models. This trend is more distinct in Figure 29(b) where the pressure drop per unit length obtained in a model with ' i ' layers, $(\Delta P / L)_{i}$, is normalized by the pressure drop observed in the model with 110 layers $(\Delta P / L)_{110}$ for several frontal velocities. When the model thickness is increased to 100 layers, the pressure drop value approaches that of the reference (110 layer) value.



Figure 29. The pressure drops obtained in the finite element model simulations of a $4 \times 4$ open section of a woven structure, where the model size in the through-thickness direction is increased from 4 layers to 110 layers. The pressure drop per unit length obtained in a model with ' i ' layers $(\Delta \mathrm{P} / \mathrm{L})_{\mathrm{i}}$ is normalized by the pressure drop observed in the model with 110 layers $(\Delta \mathrm{P} / \mathrm{L})_{110}$.

Thus, it is evident that a numerical model with just a few layers of the woven matrix will not accurately capture the pressure drops observed in real systems where the number of layers is significantly more, especially, for flows with higher inlet velocities and higher Reynolds numbers.

This observation is attributed to the fact that boundary effects at the inlet and the outlet are captured at a disproportionately higher level in the case of thin models.


Figure 30. The through-thickness flow velocity distributions observed in the numerical models of a $4 \times 4$ open porous section of a woven matrix with (a) 4 layers and (b) 50 layers in the throughthickness direction indicating that the outlet flow fields are different in the two cases.

Figures 30 and 31 present the resultant velocity distributions of two models for which the Reynolds number are identical, i.e., 78.4. Figure 30 shows the axial velocity through the mid-plane of the matrix. The top distribution the top figure contains four layers of the woven matrix and the bottom figure contains 50 layers. In Figure 31, the cross-sectional velocity profiles are shown. Figure 31(a) shows the exit velocity at the fourth layer for a 4-layer matrix. Figures 31(b) - (f) show the cross-sectional velocity at layers $4,10,20,30$ and 50 , respectively. While the velocity distribution in the fourth layer looks similar in both the models, the flow pattern continues to develop and reaches a steady state only around the $30^{\text {th }}$ layer. Thus, the flow patterns observed at the respective outlets (fourth layer for the thin model and the $50^{\text {th }}$ layer for the thick model) display significant differences. The velocity distribution in the $30^{\text {th }}$ layer and $50^{\text {th }}$ layer is similar indicating
that the flow pattern starts to converge from about the $30^{\text {th }}$ layer, which is consistent with the results for the pressure drops presented in Figure 29.


Figure 31. The flow velocity distributions observed in the numerical models of a $4 \times 4$ open porous section of a woven matrix in the thin models in $4^{\text {th }}$ layer (a) and the thick models in the $4^{\text {th }}(\mathrm{b}), 10^{\text {th }}$ (c), $20^{\text {th }}(\mathrm{d}), 30^{\text {th }}(\mathrm{e})$ and $50^{\text {th }}(\mathrm{f})$ layers.

By tracking the average pressure along the through-thickness direction of the fluid as it flows through the woven network porous media (Figure 32(a)) with large thickness in out-of-plane direction, it is evident that the fluid pressure drops at a medium flow rate as it flows through the first few layers (Figure 32(b)) and as the flow pattern reaches a steady state, the pressure drop reaches a steady state as well.

Thus, in order to obtain accurate estimates of the pressure drops in flow through woven matrices, it is important to conduct a sensitivity study and use models that are sufficiently thick such that the flow patterns are fully developed in the through-thickness direction


Figure 32. (a) The variation of fluid pressure as the fluid flows through a woven matrix porous medium observed in finite element modeling of a porous medium with 110 layers. (b) The rate of change in pressure as the fluid flows through the porous medium. ( L is the through thickness distance in the porous medium.)

### 2.5.2.3 dP/dL in large size numerical models

In previous sections, it has been clearly demonstrated that the change of pressure drop rate at different thickness of models would help to understand the flow behavior as the fluid flows through the porous medium. Therefore, in this section, more outcomes will be displayed from this point of view.

According to Ergun's analysis (equation 32), the pressure loss when incompressible gas flowing through porous media should be proportional to the sample thickness. In other words, the value of $d P / d L$ needs to be constant throughout the whole model which can be seen at high layer number of figure 33 .


Figure 33. The rate of change in pressure as the fluid flows through the porous medium with different values of Reynold number.

Ergun equation is based on the mean flow velocity whose value is exactly a "constant" when fluid density fixed, i.e., fluid cannot be compressed. However, the implication of the
"constant", i.e., distribution of velocity could be varied via distance. Therefore, deviation happened at beginning layers for perfectly stacked woven screens. Higher the Reynold number, means higher inlet gas velocity, more layers are affected, by more intensive degree. However, even when Re as large 163.3, the model finally reached a status consist with Ergun's prediction when the velocity distribution become stable and no more change with the increasing of screens layer.

The dramatic changing of $d P / d L$ at top and bottom of calculating models might due to the entrance and exit effects. It looks like these two effects are constrained in the very first layers at two ends.

### 2.5.3 Comparison of simulation results with existing correlations and experiment data

The characteristic relationship observed between the friction factor $C_{f}^{\prime}$ and the Reynolds number $R e^{\prime}$ is shown in Figure 34(a). While Sodré and Parise [69] predict the correct trends (with an average error of about $25 \%$ ), the 4-layer model simulations match the experimental results very well (with an average error of $6 \%$ ) over the range of Reynolds numbers tested. All models with larger thicknesses in the flow direction underestimate the pressure loss to varying extents.

Simulation results are also compared to the two-parameter [84] and three-parameter [83] correlations for the friction factor as shown in Figure 34(b), indicating the relationship between friction factor in $C_{f}$ and Reynolds number in $R e$. The results predicted by Tanaka's two-parameter correlation has almost the same error levels as those predicted by Sodré and Parise's while the average error of directly applying Gedeon and Wood's three parameters correlation is about $10 \%$. Thus, the three-parameter correlation is observed to provide a better match than the two-parameter correlation for the range of Reynolds numbers explored.

One possible reason for the observed differences between the simulation results in the present study and those predicted by the friction factor correlations reported in the literature [69, $83,84]$ could be due to the differences in the geometry of the woven matrix considered. The woven matrix media model considered in the present study, which captures the real system used in experiments, features a fiber lay-up that follows a $0^{0}, 45^{\circ} 90^{\circ}$ rotational sequence in the throughthickness direction, which is not captured in the correlations reported in the literature.

The observed, larger than average deviation, in the friction factor, between the simulated results and experiments, at very low Reynolds numbers regime, might be due to the inaccurate measurement of pressure differences, in the experimental set-up, according to Green and Zhishuo [73], where the pressures were about only about 300 Pa , while there is a relatively better match at higher Reynolds numbers where the pressures were about $15,000 \mathrm{~Pa}$.


Figure 34. The characteristic relationship between the friction factor $C_{f}^{\prime}$ or $C_{f}$ and the Reynold's number $R e^{\prime}$ or $R e$ as predicted by Sodré and Parise [69] or a two-parameter model [84] and a threeparameter model [83] for the friction factor compared to finite element model simulations and experiments.

Overall, the simulations of the 4-layer woven media models developed in the present study provide a numerically better match to experimental results. However, as discussed in Section 5.2.2, models with larger through-thicknesses are conceptually expected to provide more accurate results
in capturing the flow and pressure drop characteristics. This apparent paradox can be rationalized as follows.

The numerical models of thick systems where many layers were considered, inherently assumed that the layers are perfectly stacked on top of each other such that all the centers of each layer are all aligned with no lateral displacement in any layer. While theoretically it is possible to create such a perfect matrix, in reality it very likely that a geometrically defective misaligned layer whose center is laterally displaced relative to the other layers, may be created while the woven structure is fabricated (Figure 35). Hence, it is important to assess the influence of such defects on the overall flow behavior through the woven matrix models.


Figure 35. Schematics illustrating the introduction of a misaligned (defective) top layer by laterally displacing the top layer by a distance that is equal to half the open pore size.

### 2.5.4 Effect of defects on flow behavior in porous media with woven structures

A systematic study of the effects of defect density and defect intensity on the flow behavior though porous media with woven matrices was carried out. The detailed setting from section 2.5.3, i.e., $d=56 \mu \mathrm{~m}, w=280 \mu \mathrm{~m}, \beta=80 \%$ would be prolonged in this section.

### 2.5.4.1 Effect of defect density

As shown in Figure 36 in a numerical model with 40 layers, a misaligned defective layer is introduced every $4,5,8,10$ or 20 layers. Each defective layer is dislocated from its perfect position by half the open-pore size, i.e., $140 \mu \mathrm{~m}$.


Figure 36. Schematics illustrating the introduction of a misaligned defective layer every 4, 5, 8, 10 or 20 layers in a model with 40 layers.

From the results presented in Figure 37, it can be found that as the density of the defective layers increases, the pressure drops predicted by the model increases, as each defective layer disrupts the steady flow pattern and impedes the easy flow of fluid. For flow conditions where the Reynolds number ( $R e$ ) is small, the pressure loss difference between models with the most and
least defect densities is less than $15 \%$, while for flows with large Reynolds numbers ( Re ), the pressure loss difference increases to about $35 \%$. Moreover, the results from the 40-layer thickness woven matrix model with defective layer every 8 layers match the experimental records best, especially at higher Reynolds number, which is rarely seen in small size models.


Figure 37. The characteristic relationship between the friction factor $C_{f}$ and the Reynold's number $R e$ as predicted by the 40 -layer thickness numerical model simulations of a $4 \times 4$ open porous section of a woven matrix with misaligned defective layers and observed in experiments.

### 2.5.4.2 Effect of defect intensity

In order to ascertain the effect of the extent of misalignment in the defective layers on the flow behavior in porous media with a woven matrix structure, four representative defect configurations whose sizes scale with the length-scale of the open porous structure $\left(1 / 8^{\text {th }}, 2 / 8^{\text {th }}\right.$, $3 / 8^{\text {th }}$ and $4 / 8^{\text {th }}$ of the open porous size, i.e., $35 \mu \mathrm{~m}, 70 \mu \mathrm{~m}, 105 \mu \mathrm{~m}$, and $140 \mu \mathrm{~m}$ ), were introduced into a numerical model that had a defect every eight layers (Figures 38 and 39). (In order to eliminate boundary effects, the locations of the defective regions were chosen such that a defective layer was not present at the end of the midstream section of the model (Figure 39).)


Figure 38. Schematic illustrating the introduction of defective layers every 8 layers in a model with 40 layers.


Figure 39. Schematics illustrating the introduction of defective layers with different extents of misalignments (i.e., lateral displacements of $35 \mu \mathrm{~m}$ (a), $70 \mu \mathrm{~m}$ (b), $105 \mu \mathrm{~m}$ (c), and $140 \mu \mathrm{~m}$ (d) relative to the perfectly aligned position of the layers), in a model with 40 layers.

As illustrated in Figure 40, the pressure drops, in general, increase with the extent of misalignment in the defective layers. The maximum deviation, of about $20 \%$, is observed for flows where the Reynolds number ( $R e$ ) is high (262), between models of which have the smallest (35 $\mu \mathrm{m})$ and the largest $(140 \mu \mathrm{~m})$ misalignment. However, the difference in the pressure drops observed for models where the defective layers had misalignments of $140 \mu \mathrm{~m}$ and $105 \mu \mathrm{~m}$ is quite small, which two match the experimental consequences mostly.


Figure 40. The characteristic relationship between the friction factor $C_{f}$ and the Reynold's number $R e$ as predicted by the numerical model simulations of a 40 -layer thickness $4 \times 4$ open porous section of a woven matrix with defective layers which have different extents of misalignments every eight layers and observed in experiments.

Thus, it is evident that the extent of misalignment (i.e., defect intensity) and the frequency of defects (i.e., defect density) in the perfect arrangement of layers in the through thickness direction affect the flow behavior and in general, increase the pressure drops observed in flow through porous media with woven matrix structures. Therefore, in order to model the flow characteristics in woven porous media accurately, appropriate defects in their lay-up structures need to be considered as well. Ergun's analysis works well for most practical sample might thanks to the contribution of randomly dispersed defects which, therefore, resulting in a statistically stable velocity distribution throughout the whole model.

In summary, the finite element simulations on thin models with just four woven matrix layers predict larger pressure drops because they capture the inlet and outlet boundary effects in a disproportionate manner, while simulations of larger models with defective layer structures also predict large pressure drops because of the consideration of the defects in the lay-up. Hence, even
though a good match observed between the simulations using thin models and experiments, it is physically more appropriate to make a comparison of the experimental results with the predictions of simulations with larger models which incorporate defective layers.

### 2.5.6 A method to increase application scope of a correlation: modification of $\boldsymbol{d}_{\boldsymbol{h}}$

From the previous discussion, it is clear models with same Reynold number, Re, but different fiber thickness, $a$, exhibit similar characteristics as fluid flow through woven matrix media, while models with same Reynold number, $R e$, but different porosity, $\beta$, show significant dissimilarity. The same phenomenon was also observed in Costa's [59, 64] work, where friction factor, $C_{f}$, for woven matrix media models with divergent porosity values are evidently different from each other and therefore, could be fitted by separated curves, which were also based on numerical methods. Gedeon [83] tried to explain such dependence on porosity, $\beta$, of friction factor, $C_{f}$, and capture all experimental data just using one single equation, which is an advanced version of equation 44 after including the term of $\beta^{a_{4}}$ :

$$
\begin{equation*}
C_{f}=\left(a_{1} / R e+a_{2} R e^{a_{3}}\right) \beta^{a_{4}} \tag{50}
\end{equation*}
$$

Yet, equation 50 seems not work well for its original intentions. Here, in present work, 114 woven matrix models were created to further study how $R e, \beta$ and $a$ can affect the expression of $C_{f}$, i.e., in an independent way or in a comprehensive way. Although in section 2.5.3, a conclusion was proposed that small size models, especially thin models are not suitable for making a prediction of pressure loss as fluid flow through porous media, they can still be applied for unearthing some important rules, since boundary effect and size effect are hoped to be the same for all models at a certain value of $R e$. The out-of-plane dimension of the midstream part for these models was fixed at 4 times the thickness of screens, $t$, i.e., 8 time the fiber thickness, $a$. The inplane size of the midstream part was delicately designed to make the ratio between the woven matrix size and the fiber size equivalent to or greater than 20 . Under such ratio, it's believed boundary effects could be ignored for all models in this section, according to the discussions from section 2.5.2.1. To be more specific, the in-plane sectional size would be $6 \times 6,8 \times 8,12 \times 12$ open pores for models with porosity as $45 \%, 60 \%$ and $75 \%$, respectively.

### 2.5.6.1 Setting of 114 models

These models were obtained by weaving square fibers (with five different thickness: $40 \mu \mathrm{~m}, 60 \mu \mathrm{~m}, 80 \mu \mathrm{~m}, 110 \mu \mathrm{~m}$ and $140 \mu \mathrm{~m}$ ) into the woven pattern as shown in figure 16 . Three equal spaced porosity value, $45 \%, 60 \%$ and $75 \%$ were gained by controlling the open size of
models. The frontal velocities of fluid were increased from $0.125 \mathrm{~m} / \mathrm{s}$ to as high as $40 \mathrm{~m} / \mathrm{s}$, while corresponding values of Reynold's number varied from 2 to about 250 . The detailed records of all 114 simulation models were compacted into figure 41.


Figure 41. Plot includes all 114 models with different fiber thickness and Reynold number.
According to equation 43 and equation 44 , friction factor, $C_{f}$, changes more dramatically at low Reynold number. Therefore, more models are concentrated at this region, just as displayed in figure 41 , in order to capture the correct tendency for $C_{f}$.

### 2.5.6.2 Raw simulation results analysis

Simulation outcomes of total 114 models were first divided into smaller groups according to their fiber thickness and friction factor, $C_{f}$, calculated through equation 40, then were displayed in figure 42(a) - (e).

(b)
(d)

Figure 42. Plots illustrating the characteristic relationship between the friction factor $C_{f}$ and the Reynold's number $R e$ as predicted by 4-layer numerical models with dissimilar fiber thickness and divergent porosity.

In the first place, regardless the value of fiber thickness, $a$, curves standing for the relationship between $C_{f}$ and $R e$ of models with divergent porosities undoubtedly apart from each other. Second, when Reynold's number, $R e$, is less than 50 , the friction factor, $C_{f}$, of a model with higher porosity is much greater than those of models with lower porosity, which is consistent with Costa's observation [64]. Moreover, curves standing for models with divergent porosity tend to converge at a higher Reynold number. And as $R e$ continues to increase, curves for lower porosity
models would even make a crossover of those for higher porosity models, which means the corresponding values of friction factor $C_{f}$ of lower porosity models become greater than those of higher porosity models. At the right end of the curve for models with porosity equal to $45 \%$, there is a clear tendency for it to become horizontal, which could be a signal as flow behavior transfer from laminar to turbulent. Nevertheless, such signal is not strong for other two curves. Third, curves represent a specific porosity value for models with dissimilar fiber thickness seem to occupy the same position in all figures. In other words, the relationship between friction factor, $C_{f}$, and Reynold number, $R e$, looks like only dependent on the variable of porosity, $\beta$.


Figure 43. Plot illustrating the characteristic relationship between the friction factor $C_{f}$ and the Reynold number Re based on Gedeon's research [83] on regenerators comprised of woven screens.

Above observations should not appear theoretically as stated by equation 35, 43 and 44, in which $R e$ is the only declared mutable factor for $C_{f}$. However, same phenomena were also verified experimentally by Gedeon [83] as displayed in figure 43. In figure 43, the red line represents the relationship between the friction factor, $C_{f}$, and the Reynolds number, $R e$, for the generator of
porosity equal to $78.1 \%$ and made up by circular wires with its diameter equal to $55.9 \mu \mathrm{~m}$, i.e., 100 mesh per inch. The red line is obtained by fitting the three-parameter equation (equation 44) for friction factor, $C_{f}$, and $a_{1}, a_{2}$ and $a_{3}$ equal to $138.9,2.567$ and -0.0816 , respectively. $a_{1}, a_{2}$ and $a_{3}$ have the value of $120.1,2.369$ and -0.0836 , correspondingly, for the blue line and 129.3, 2.99 and -0.0758 , respectively, for the black line. It is clear that the red line, standing for the highest porosity media, is at the top of the plot, and crossed by the black line, standing for the lowest porosity media when $R e$ is raised to about 25 . This value is somehow smaller than the observation in the present work. The other inconsistency between present work and Gedeon's results is that the black line, represents a low porosity regenerator of $62.32 \%$, is above the blue line, represents a medium porosity regenerator of $71.02 \%$, even at low Reynold number. Both discrepancies can be explained as the values of $a_{1}, a_{2}$ and $a_{3}$ from Gedeon's work were obtained by fitting the pressure loss, $\Delta P$, with Reynold number, $R e$, which was able to be as big as 6,000 in experiments. Since the Re region of interesting in present work is only a very small part of it, and is at one end of it, some deviations are reasonable and acceptable. In general, 114 simulations successfully captured the basic rules between $C_{f}$ and $\operatorname{Re}, \beta$ and $a$.

### 2.5.6.3 Modification of expressions for $d_{h}$ and $R e$

As stated by above discussions, it looks like the best way of accurately describing the relationship between friction factor, $C_{f}$, and Reynold number, $R e$, is to create expressions case by case, i.e., to create an individual expression for a woven matrix model with a specific porosity value. Or, by sacrificing accuracy for simplicity, an overall expression could be obtained by fitting all data together, just as what has been tried by Gedeon [83]. However, it's well known, in the region of laminar flow, an alike flow system with the same value of Reynold number, $R e$, which is based on the characteristic length of the system, should give out consistent results. Therefore, a thought floated that might not all the key issues were about the equation form, but about the definitions in it.

In equation $39, S_{v}$ is only depended on the crossing sectional shape ( $S_{v}$ is $4 / d$ for circular fiber and $4 / a$ for square crossing sectional fiber), and have nothing to do with porosity $\beta$. This is basically correct when $\beta$ is very high, e.g. $80 \%$ or $90 \%$. However, the effect of overlapped area between any two fibers cannot be ignored anymore while $\beta$ is low. Overlapped area does evidently
exist in square fiber models, as displayed in figure 44, and real stacked woven screens samples which have been compressed at both ends.


Figure 44. Schematic illustrating the overlap area between any two square crossing sectional fibers which would greatly impact the calcualtion of specific surface of solid, $S_{v}$.

Based on the definition of specific surface of solid, $S_{v}$, and associated with the schematic for overlap area between any two square fibers as above, i.e., between green and orange fibers or between orange and blue fibers, an equation for calculating the modified value of $S_{v}^{*}$ for a square crossing sectional fiber is given as:

$$
\begin{equation*}
S_{v}^{*}=\frac{4}{a}-\frac{2}{(w+a)}=\frac{4}{a}\left(1-\frac{a}{2 w+2 a}\right) \tag{51}
\end{equation*}
$$

where, the superscript ' ${ }^{*}$ ' is applied to distinguish between original specific surface (in equation 39) and the one after modification. Moreover, as open size, $w$, is function of fiber thickness, $a$, and model porosity, $\beta$, equation 51 is rewritten like below:

$$
\begin{equation*}
S_{v}^{*}=\frac{4}{a}\left(1-\frac{1-\beta}{2}\right)=\frac{2(1+\beta)}{a} \tag{52}
\end{equation*}
$$

In equation 52 , when porosity $\beta$ is very high, the value of $\frac{a}{2 w+2 a}$ could be ignored, equation 52 approach the value of $4 / a$. For example, if $\beta$ is $90 \%$, then $\frac{a}{2 w+2 a}$ equals to 0.05 . The Difference between equation 52 and the value of $4 / a$ is only 5 percent. However, if the value of porosity $\beta$ is very low, e.g. $45 \%$, then $\frac{a}{2 w+2 a}$ is as large as 0.275 . Deviation magnitude between equation 52 and the expression of $4 / a$ could be as large as $27.5 \%$. The overestimation of the specific surface of solid, $S_{v}$, would have a direct influence on the calculation of hydraulic diameter, $d_{h}$ and therefore result in an underestimating of Reynold number, $R e$. To be exact, lower the model porosity value, larger the extent of undervaluing of Reynold number. As models with dissimilar porosity would have different discrepancy extent, curves in figure 42 apart from each other now can be well explained.

Hence, in order to obtain an accurate expression for relating friction factor, $C_{f}$ to Reynold number, $R e$, and make precise prediction of pressure drop $\Delta P$ as fluid flow through porous media, it's necessary to redefine hydraulic diameter, $d_{h}$, first as:

$$
\begin{equation*}
d_{h}^{*}=\frac{4 \beta}{S_{v}^{*}(1-\beta)} \tag{53}
\end{equation*}
$$

then Reynold number, $R e$, as

$$
\begin{equation*}
R e^{*}=\frac{\rho}{\mu} u_{m} d_{h}^{*} \tag{54}
\end{equation*}
$$

### 2.5.6.4 Simulation results analysis based on modified

The friction factor of all 114 simulation, together with the correlations from Gedeon's [83] work were recounted, by applying the modified value of hydraulic diameter, $d_{h}^{*}$, through equation as below, and the new relationship between $C_{f}^{*}$ and $R e^{*}$ were shown in figure 45 and figure 46:

$$
\begin{equation*}
C_{f}^{*}=\frac{2 \Delta P d_{h}^{*}}{\rho L u_{m}^{2}} \tag{55}
\end{equation*}
$$



Figure 45. Plot illustrating the characteristic relationship between the modified friction factor $C_{f}^{*}$ and the modified Reynold number $R e^{*}$ based on Gedeon's research [83] on regenerators comprised of woven screens.


Figure 46. Plots illustrating the characteristic relationship between the modified friction factor $C_{f}^{*}$ and the modified Reynold number $R e^{*}$ as predicted by 4-layer numerical models with dissimilar fiber thickness and divergent porosity.

Although in Gedeon's work [83], crossing section of fibers is circular, equation 52 is still applied for the purpose of simplification as the similarity between square fiber and circular fiber, which has been mentioned before. Obviously, in figure 45, curves represent regenerators with $78.1 \%$ and $71.02 \%$ porosity become closer when compared to figure 43 . But the situation that black line $(62.32 \%)$ turn to be the top curve in the whole region of $R e^{*}$ was not expected. Such phenomenon can be explained as in Gedeon's work, $R e^{*}$ could be as large 7500, therefore the
value of all parameters ( $a_{1}, a_{2}$ and $a_{3}$ ) obtained through fitting $C_{f}^{*}$ with $R e^{*}$ might not fully captured the characteristic rules in this small and marginal section.

In figure 46, the tendency of converging for three curves, represent divergent porosity ( $45 \%, 60 \%$ and $75 \%$ ), is not only apparent but also precise. When $R e^{*}$ gradually increases, curves represent different porosity start to fork off the convergent results in sequence, from low porosity to high porosity. Such phenomenon indicated a possibility that the critical value of $R e^{*}$, at which flow behavior transferred from distinct laminar to distinct turbulent, might vary for different models, i.e., depend on the value of model porosity. Generally speaking, the low value of $\beta$ means high density of sudden boundary change in a woven structure which could become a great improvement to the formation of turbulence, as fluid flow through porous media. On the contrary, models with high porosity would hinder the process of turbulence development. Another possible explanation for crossover between curves is due to the failure of the assumption of laminar flow under high inlet velocity in simulations. Even so, the variance among the values of $R e^{*}$, at which, crossover had been detected, still can provide the conclusion that porosity, $\beta$, of woven matrix model would determinate at what time flow behavior would transfer from distinct laminar to distinct turbulent. For example, the critical value of $R e^{*}$ equals to 35 and 80 for models with the porosity of $45 \%$ and $60 \%$ respectively, as its value become about 200 for models with the porosity of $75 \%$ or higher.

The remaining deviation of curves for high porosity, i.e., $75 \%$, at very small $R e^{*}$ is probably due to the computational accuracy of finite element method. For example, as the thickness of square fibers is as large as 140 um , the model pressure loss, $\Delta P_{m o}$, of a 4-layer woven matrix model with porosity $\beta$ equals to $75 \%$, is only 1.1 Pa and 2.3 Pa when $R e^{*}$ is 5 and 10 , respectively. At the same time, the absolute pressure value for the whole model is over $1 \times 10^{5}$. Such problem might be solved by using a precise model with large size in the direction of flow. Otherwise, a treatment of excluding results with the pressure drop less than 10 Pa is suggested if those data would be used for creating a new correlation.

In present work, all 114 results were included into one plot as shown in figure 47. In figure 47 (a), 114 data points could have been grouped easily based on the porosity value. After the modification of hydraulic diameter, $d_{h}$, all data points now can distribute along a single curve.

First, it's very clear that fiber thickness, $a$ has little influence on the relationship between weather $C_{f}$ and $R e$ or $C_{f}^{*}$ and $R e^{*}$. Second, there is no necessity to create individual expressions for $C_{f}^{*}$ case by case, if the modified Reynold number, $R e^{*}$, would be applied, at least when $R e^{*}$ no more than 200. Above all, this rule meets people's expectation in this area, that Reynold number, $R e^{*}$ plays a decisive role for laminar flow as long as a suitable definition is found.
(a)

(b)


Figure 47 Plots displayed the simulation results from 114 models: (a) from the view of $C_{f}$ and $R e$; (b) from the view of $C_{f}^{*}$ and $R e^{*}$.

Miyabe [60] claimed that the area can be used for heat transfer between the fluid and solid, i.e., the contacted area, is considered as the total surface of fibers substrate the overlapped surface. Thus, the precise modification of specific area for circular crossing sectional fibers can be obtained through a careful derivation by following Miyabe's work. The expression of modified specific area, $S_{v}^{*}$, for a woven structure made up by circular fibers is as:

$$
\begin{equation*}
S_{v}^{*}=\frac{4}{d}\left(1-\frac{1}{2 \sqrt{\left(\frac{w}{d}+1\right)^{2}+1}}\right) \tag{56}
\end{equation*}
$$

where $\frac{w}{d}$ is the function of porosity, $\beta$. A general equation of calculating specific area, $S_{v}^{*}$, is given before ending of this section:

$$
\begin{equation*}
S_{v}^{*}=\frac{4}{d_{t}}(1-F(\beta)) \quad(0<F<0.5) \tag{57}
\end{equation*}
$$

where $F$ is an expression about porosity, $\beta$, can be determined by theoretical derivation or experiment test. Mehta [90] also tried to modify the calculation of hydraulic diameter earlier by including in the surface of walls and proved its effectiveness. In a large size model, compared to the effects of wall boundary, the impact from overlapped surface is at least an order of magnitude greater, which should not be ignored anymore.

In short, when $R e^{*}$ is less than 200 , if the woven configure is decided, there is a single expression which can depicture the relationship between friction factor and Reynold number, regardless of fiber thickness and mode porosity. When $R e^{*}$ is greater than 200 , a serial of equations is required to give an accurate prediction of pressure loss as fluid flow through a woven matrix model.

### 2.5.7 Study of feasibility as applying small-scale models on random stacked media

Gedeon [83] at the same time carried out a series of experiments to research the pressure loss as fluid flow through metal-felt matrices, which was made from circular crossing sectional fibers. In metal-felt matrices, fibers mainly lie transverse to the out-of-plane direction, also the flow direction. As in the plane, fibers were distributed with random angle and random spacing. From a practical perspective, the metal-felt matrices were of interest in the present study since the cost for processing metal-felt matrices was much cheaper than processing woven matrices. In order to simplify finite element models, all fibers were restricted in the plane perpendicular to the flow direction. So as to better understand how the degree of randomness would affect the pressure loss, 20 models were separated into two groups, i.e., model \#1 to \#10 belonged to group I, while model \#11 to \#20 belonged to group II. In group II, the angle of fiber and the spacing between two fibers were totally random, except that all fibers were required to parallel to each other for an individual layer, as shown in figure 19. As in group I, there was one more restriction to fibers angle, so as to make sure fibers could only extend in the direction $0^{\circ}, 45^{\circ}, 90^{\circ}$ and $135^{\circ}$, which were also the directions for fibers in woven matrices. Thus, hereafter, models of group II were referred to highly random models, as models of group I were referred to lowly random models.

(a)
(b)


Figure 48. Plots displayed pressure loss per unit length, $\Delta P / \Delta L$, for two groups of random stacked models.

The simulation outcomes of pressure loss per unit length, $\Delta P / \Delta L$, for two groups of random stacked models are shown in figure 48 . For both groups, the value of $\Delta P / \Delta L$ at different
frontal velocities show dispersion to various extents. In figure 48(a), the $\Delta P / \Delta L$ obtained from model \#10 is $11.5 \%$ smaller than that from model \#5. In figure 48(b), the value of pressure loss per unit length for model \#17 is $11.1 \%$ smaller than that from model \#14. In order to have a direct impression of dispersion degrees of simulation results from 8-layer (fiber layer) random stacked models, the value of $\Delta P / \Delta L$, under different frontal velocities, are normalized to the average pressure loss per unit length of each group. The comparison among 20 models is demonstrated in figure 49.


Figure 49. Plots illustrating the dispersiosn of simualtion results for two groups of random stacked models.

First, results dispersion degree for two groups looks close to each other. The largest dispersion degree in both groups is less than $\pm 10 \%$. More than half results have the dispersion degree less than $\pm 2.5 \%$. Nevertheless, it's still unsafe to simulate fluid flow metal-felt matrices with a thin 8-layer model. Up to $7.5 \%$ (this value might further increase since only 10 models of each group were studied) deviation might happen. Second, Reynold number, Re, has little influence on results dispersion of lowly random models, which could be seen in figure 49(a). However, after introducing the freedom of fiber angle, the phenomenon of results dispersion for models with low $R e$ becomes greater markedly in figure 49(b). It might due to that fluid flow at low $R e$ is more sensitive to the structure changes. For pressure drop at high Reynold number, $R e$, the main influence factor would be the density of sudden boundary or structure changes, which is a function of porosity, $\beta$, and have a minor connection with the angle change or spacing change.

The averaged pressure loss of either group was then substituted into equation 40 to calculate the average friction factor, $C_{f}$. The consequences were then compared to Gedeon's [83] work, experimental and simulation results of the present study as shown in figure 50.


Figure 50. The average friction factor, $C_{f}$, obtained from two groups of random stacked models were compared with the consequence of Gedeon's work [83], and experimental and simulation results of present study.

In Gedeon's work, the value of friction factor, $C_{f}$, for metal-felt matrices is about $50 \%$ higher than it of woven matrices. The friction factor, $C_{f}$, obtained through simulating fluid flow through a random stacked model is $25 \%$ greater than the value got from woven matrix model, as two model have the same size of out-of-plane direction. The big deviation between simulations (8layer random stacked model of the present study) and experiments (Gedeon experiment on metalfelt matrices) might due to the large range of $R e$ (up to 2500) in experiments. More careful and concentrated (focus on Re region of interesting) experiments will be required. Another possible reason could be that oscillating flow was applied in Gedeon's work which might be more sensitive to a random structure.

In summary, larger size in the out-of-plane direction is preferred when simulating fluid flow through random stack models. Woven matrix structure can greatly reduce the pressure loss when compared to the random stacked structure.

### 2.6 Conclusion

Due to a multitude of application areas where flow characteristics through a porous medium need to be accurately predicted, there exists a strong motivation for a detailed analysis of flow behavior of fluids through a porous medium. Due to the complexity of the flow patterns in a porous medium that features a woven matrix configuration, a number prior studies have focused on developing numerical models. Due to the computational cost considerations, many of the numerical studies reported thus far have been able to model only relatively small regions of a physically large woven matrix. Little insight on the flow behavior in large model systems obtained through computational methods are at present available. Hence, present study focused on developing numerical models that capture the flow behavior of a fluid over large regions of a woven matrix open porous structure. The phenomenon of scatter of $C_{f}$ values under same $R e$ value was also studied in present work. In addition, a series of systematical simulations were carried out for small scale random stacked models. The main conclusions obtained from the present study are given below.

1. A finite element model that captures the geometric characteristics of a real woven matrix comprised of circular cross-section fibers and curvature due to fiber bending is developed with an equivalent model system comprised of fibers with square cross-section.
2. Changes in the in-plane size of the finite element model, lateral to the fluid flow direction, had relatively minor effects on the pressure drops predicted by the models. Significant boundary effects were observed only in the case of models that were very small, e.g., with $2 \times 2$ open porous section.
3. Changes in the thickness of the finite element model in the fluid flow direction had significant effects on the pressure drops. In simulations with very thin models, the boundary effects had a greater influence and caused the predicted pressure drops to increase proportionately. On the other hand, simulations with thick models indicated that the flows were fully developed and the boundary effects were minimized, resulting in relatively smaller pressure drops.
4. Defects in the lay-up of the woven matrix layers were also shown to have a significant
impact on the pressure drops predicted by the simulations. Higher defect densities resulted in greater pressure drops as they disrupted the steady flow of fluid in the through-thickness direction.
5. Higher defect intensities also resulted in greater pressure drops. Such tendency receded dramatically as defect size over $3 / 8^{\text {th }}$ of the open porous size.
6. The pressure drops obtained in the finite element model simulations of thick models that contained some defective layers matched very well with experimental observations.
7. Scatter of $C_{f}$ values at low $R e$ (less than 200) can be well explained and amended by carefully recalculating specific surface of solid, $S_{v}$.
8. The value of pressure loss obtained from small scale random stacked models scattered intensively. Maxima and minima deviated from average value by $\pm 7.5 \%$.
9. The averaged pressure loss obtained from small scale random stacked models was about $20 \%$ greater than it from small scale woven matrix model.

## Future Works

Although many works have been done and many interesting conclusions have been obtained in the present study, it's still too early to say everything is going to be clear soon. More dedicated and challenging problems are waiting for explanations. In order to have a deeper understanding in both areas, there are some urgent topics for future works as:
(Computational Modeling of Indentation of Thin Films)

1. To condense the dimensionless equations with empirical formulae or theoretical derivations, so as to use them for practical purposes and on much more complex systems;
2. To extract more information from the substrate, e.g., plasticities and anisotropy, as penetration ratio is high;
3. To create dimensionless equations for "thick" film system, i.e., penetration ratio is restricted at a low level intentionally;
(Computational Modeling of Flow Through Porous Media)
4. To build a perfect woven medium with three-dimensional printing technic and verify the phenomenon of lower pressure drop in such media as fluid flowing through them;
5. To study the heat transmission in large size and perfect woven structures;
6. To extend the conclusions from the present study to the region of the compressible fluid and turbulent flow.

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## Appendix A

The critical values of $y_{i 4}$ and $y_{i 5}$ for grouping in present study, i.e., $b_{i 1}$ and $b_{i 2}$, and $c_{i 1}, c_{i 2}$ and $c_{i 3}$

| $\Theta$ | $b_{i 1}$ | $b_{i 2}$ | $C_{i 1}$ | $C_{i 2}$ | $C_{i 3}$ |
| :---: | :---: | :---: | :---: | :---: | :---: |
| $60^{\circ}$ | 1302.5 | 4570 | 724400 | 532350 | 485275 |
| $70.3^{\circ}$ | 5906.25 | 18057.5 | 1208400 | 846250 | 825250 |
| $80^{\circ}$ | 50486.25 | 94876.25 | 2207750 | 1199750 | 1817500 |

## Appendix B

The specific expression of $y_{i j k}$

$$
\begin{aligned}
y_{i j k}=a_{i j k 1}+ & a_{i j k 2} x_{1}+a_{i j k 3} x_{2}+a_{i j k 4} x_{3}+a_{i j k 5} x_{4}+a_{i j k 6} x_{1} x_{2}+a_{i j k 7} x_{1} x_{3}+a_{i j k 8} x_{1} x_{4} \\
& +a_{i j k 9} x_{2} x_{3}+a_{i j k 10} x_{2} x_{4}+a_{i j k 11} x_{3} x_{4}+a_{i j k 12} x_{1}^{2}+a_{i j k 13} x_{2}^{2}+a_{i j k 14} x_{3}^{2} \\
& +a_{i j k 15} x_{4}^{2}+a_{i j k 16} x_{1} x_{2} x_{3}+a_{i j k 17} x_{1} x_{2} x_{4}+a_{i j k 18} x_{2} x_{3} x_{4}+a_{i j k 19} x_{1}^{2} x_{2} \\
& +a_{i j k 20} x_{2}^{2} x_{1}+a_{i j k 21} x_{1}^{2} x_{3}+a_{i j k 22} x_{3}^{2} x_{1}+a_{i j k 23} x_{1}^{2} x_{4}+a_{i j k 24} x_{4}^{2} x_{1} \\
& +a_{i j k 25} x_{2}^{2} x_{3}+a_{i j k 26} x_{3}^{2} x_{2}+a_{i j k 27} x_{2}^{2} x_{4}+a_{i j k 28} x_{4}^{2} x_{2}+a_{i j k 29} x_{3}^{2} x_{4} \\
& +a_{i j k 30} x_{4}^{2} x_{3}+a_{i j k 31} x_{1}^{3}+a_{i j k 32} x_{2}^{3}+a_{i j k 33} x_{3}^{3}+a_{i j k 34} x_{4}^{3}+a_{i j k 35} x_{1} x_{2} x_{3} x_{4} \\
& +a_{i j k 36} x_{1}^{2} x_{4}^{2}+a_{i j k 37} x_{2}^{2} x_{4}^{2}+a_{i j k 38} x_{3}^{2} x_{4}^{2}+a_{i j k 39} x_{1}^{3} x_{4}+a_{i j k 40} x_{4}^{3} x_{1} \\
& +a_{i j k 41} x_{4}^{3} x_{2}+a_{i j k 42} x_{1} x_{2}^{-1}+a_{i j k 43} x_{2} x_{1}^{-1}+a_{i j k 44} x_{1} x_{3}^{-1}+a_{i j k 45} x_{3} x_{1}^{-1} \\
& +a_{i j k 46} x_{2} x_{3}^{-1}+a_{i j k 47} x_{3} x_{2}^{-1}+a_{i j k 48} x_{4} x_{1}^{-1}+a_{i j k 49} x_{4} x_{2}^{-1}+a_{i j k 50} x_{4} x_{3}^{-1} \\
& +a_{i j k 51} x_{2} x_{4} x_{1}^{-1}+a_{i j k 52} x_{3} x_{4} x_{1}^{-1}+a_{i j k 53} x_{1} x_{3} x_{2}^{-1}+a_{i j k 54} x_{3} x_{4} x_{2}^{-1} \\
& +a_{i j k 55} x_{4}^{2} x_{1}^{-1}+a_{i j k 56} x_{4}^{2} x_{2}^{-1}+a_{i j k 57} x_{2} x_{3} x_{4} x_{1}^{-1}+a_{i j k 58} x_{1} x_{3} x_{4} x_{2}^{-1} \\
& +a_{i j k 59} x_{4}^{2} x_{2} x_{1}^{-1}+a_{i j k 60} x_{4}^{2} x_{3} x_{1}^{-1}+a_{i j k 61} x_{1}^{2} x_{3} x_{2}^{-1}+a_{i j k 62} x_{1}^{2} x_{4} x_{2}^{-1} \\
& +a_{i j k 63} x_{4}^{2} x_{1} x_{2}^{-1}+a_{i j k 64} x_{4}^{2} x_{1} x_{3}^{-1}+a_{i j k 65} x_{4}^{2} x_{2} x_{3}^{-1}+a_{i j k 66} x_{2} x_{3} x_{1}^{-2} \\
& +a_{i j k 67} x_{2} x_{4} x_{1}^{-2}+a_{i j k 68} x_{1} x_{4} x_{2}^{-2}+a_{i j k 69} x_{3} x_{4} x_{2}^{-2}+a_{i j k 70} x_{1} x_{3} x_{2}^{-2} \\
& +a_{i j k 71} x_{1} x_{4} x_{3}^{-2}+a_{i j k 72} x_{4}^{2} x_{1}^{-1} x_{2}^{-1}+a_{i j k 73} x_{4} x_{1}^{-1} x_{2}^{-1} x_{3}^{-1}+a_{i j k 74} x_{4} x_{2}^{-2} x_{1}^{-1} \\
& +a_{i j k 75} x_{1} x_{2}^{-2} x_{3}^{-1}
\end{aligned}
$$

Values of $a_{i j k l}$ as $i=1$ and $k=1$

| $l$ | $j=1$ | $j=2$ | $j=3$ | $j=4$ | $j=5$ |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 1 | 28.23 | 35.99402 | 12.56647 | 1.32374 | 104.8189 |
| 2 | -40.9077 | -32.6324 | -11.7312 | 0.708423 | 1665.616 |
| 3 | 7.63468 | 3.866561 | 1.99634 | -0.49542 | -72.545 |
| 4 | -3.0944 | -9.24931 | -4.36252 | -1.71739 | -99.0788 |
| 5 | 29.78426 | -136.197 | -44.257 | -20.3448 | 2945.061 |
| 6 | -0.21917 | -0.82237 | -0.20339 | 0.140804 | -404.767 |
| 7 | 35.34071 | 26.41394 | 9.635051 | -1.54527 | 1335.106 |
| 8 | 51.36851 | 175.7436 | 58.14958 | 11.79565 | -6140.3 |
| 9 | -4.52572 | -2.04475 | -1.26359 | 0.569787 | -210.871 |
| 10 | 2.087082 | -12.1232 | -5.65112 | -1.23853 | -423.583 |
| 11 | 55.27897 | 195.8938 | 64.48311 | 18.16965 | -791.511 |
| 12 | -5.46501 | 2.668212 | -0.04625 | 0.916841 | -1854.57 |
| 13 | -0.70347 | -0.22078 | -0.04832 | -0.03446 | 78.87212 |
| 14 | 9.233851 | 11.29939 | 4.645505 | 1.148927 | 100.3261 |
| 15 | -12.394 | 245.0451 | 89.50229 | -4.53348 | 87.66724 |
| 16 | -1.58369 | -0.28534 | -0.11357 | 0.029615 | 73.92053 |
| 17 | 3.085067 | -4.13019 | -0.98236 | -0.00752 | -77.6218 |
| 18 | 16.35797 | 8.8335 | 3.778076 | -0.03804 | 266.5998 |
| 19 | 1.066942 | 0.300541 | 0.095688 | -0.058 | 48.18782 |
| 20 | -0.01093 | 0.057957 | 0.015715 | -0.00146 | 26.52366 |
| 21 | 4.453466 | -1.52604 | -0.05741 | -0.12247 | -46.2529 |
| 22 | -7.61603 | -4.7898 | -1.91222 | 0.240206 | -515.293 |
| 23 | -39.5626 | -101.202 | -32.641 | -11.6083 | 6833.593 |
| 24 | 529.3335 | 486.1176 | 149.9796 | -10.7394 | -8155.13 |
| 25 | 0.341427 | 0.16848 | 0.043641 | -0.02194 | -3.92168 |
| 26 | 0.616941 | -0.03119 | 0.16161 | -0.08744 | 51.7641 |
| 27 | -1.47261 | -0.10787 | 0.014523 | 0.159816 | 52.89483 |
| 28 | -41.9395 | -19.5701 | 0.853835 | 7.400561 | 5462.971 |
| 29 | -9.22032 | -47.4212 | -15.6989 | -4.5309 | 293.8602 |
| 30 | 35.74728 | -173.873 | -58.6381 | -17.6124 | -768.533 |
| 31 | -1.00824 | -0.28852 | -0.00654 | -0.22615 | 529.9474 |
| 32 | 0.02997 | -0.00037 | -0.00097 | 0.004633 | -6.46757 |
| 33 | -2.34934 | -2.41878 | -1.02833 | -0.25363 | -1.67825 |
| 34 | -32.503 | 85.39849 | 26.45575 | 78.03599 | 1726.547 |
| 35 | -8.27196 | -1.49208 | -1.14913 | 0.377395 | -172.113 |
| 36 | -184.567 | -131.314 | -51.727 | 1.379123 | -1086.32 |
| 37 | 8.292805 | 5.963372 | 1.935167 | -1.55795 | -264.677 |
| 38 | 12.19523 | 90.05132 | 31.88809 | 11.33279 | 496.4793 |
| 39 | 22.81911 | 34.5481 | 12.25171 | 3.618047 | -2130.87 |
|  |  |  |  |  |  |


| 40 | 970.0411 | 1522.557 | 568.5461 | 94.91025 | 5046.887 |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 41 | -140.678 | -351.424 | -135.497 | 41.06342 | -5355.96 |
| 42 | 10.53858 | 5.326543 | 1.828653 | 0.08469 | 3165.68 |
| 43 | -0.31691 | -0.38451 | -0.11779 | 0.052748 | 19.37514 |
| 44 | 20.14166 | 13.83045 | 5.750368 | -0.81444 | 781.2651 |
| 45 | -0.28819 | -0.23104 | -0.07197 | 0.021448 | -6.00839 |
| 46 | -1.50837 | -1.10029 | -0.76555 | 0.234897 | -126.703 |
| 47 | -0.10014 | 0.051452 | 0.016063 | -0.02019 | 9.445314 |
| 48 | -5.58392 | -2.11445 | -0.89122 | 0.409096 | -277.424 |
| 49 | 3.295729 | 5.859249 | 1.696565 | 0.06288 | -139.92 |
| 50 | 25.00267 | 78.4847 | 27.60681 | 8.099292 | -396.917 |
| 51 | -1.38719 | -0.01241 | 0.119531 | 0.590716 | 132.4424 |
| 52 | 2.692792 | 0.554518 | 0.253069 | -0.16105 | 114.0217 |
| 53 | -4.49526 | -3.25888 | -0.95101 | -0.08737 | -301.71 |
| 54 | -10.9922 | -8.0981 | -2.34397 | 0.043445 | -462.363 |
| 55 | -6.6669 | -14.7467 | -4.24534 | 0.295378 | 533.3494 |
| 56 | -1.49719 | -28.9307 | -10.3794 | -0.77007 | -545.06 |
| 57 | -6.26864 | -3.31576 | -1.10057 | 0.190218 | -158.111 |
| 58 | 25.49023 | 23.93235 | 6.530369 | -2.27537 | 381.1511 |
| 59 | 15.14217 | 4.259784 | -0.87479 | 1.259734 | -1391.18 |
| 60 | -2.55253 | -1.62555 | -0.83867 | 0.074818 | 64.45783 |
| 61 | -4.60883 | -1.20248 | -0.3125 | 0.650703 | 750.2297 |
| 62 | -16.6978 | -54.0811 | -16.6718 | -2.64468 | -701.043 |
| 63 | -9.94864 | 77.47064 | 30.14456 | 1.690289 | 111.4427 |
| 64 | -108.477 | -270.701 | -89.7889 | -20.3701 | 4565.428 |
| 65 | -6.36083 | 39.24178 | 14.09528 | -0.92074 | -1003.83 |
| 66 | 0.004771 | 0.003884 | 0.000999 | -0.00011 | -0.35506 |
| 67 | 0.331522 | 0.103213 | 0.038135 | -0.04719 | 11.06315 |
| 68 | -0.9275 | -0.49023 | -0.05543 | 0.277181 | 45.67191 |
| 69 | 0.48258 | 0.187837 | 0.050691 | 0.018287 | 62.57093 |
| 70 | 0.217092 | 0.139941 | 0.037678 | 0.012317 | -88.4372 |
| 71 | 7.876726 | 16.15563 | 4.475112 | 2.804682 | 17.47823 |
| 72 | 0.722164 | 2.488496 | 0.799264 | -0.12852 | -33.7303 |
| 73 | -0.07965 | -0.17531 | -0.04679 | -0.00605 | 22.04512 |
| 74 | 0.031984 | 0.013472 | 0.004548 | -3.55E-05 | -0.86483 |
| 75 | -0.75203 | -0.39738 | -0.13306 | -0.01879 | -141.709 |

Values of $a_{i j k l}$ as $i=1$ and $k=2$

| $l$ | $j=1$ | $j=2$ | $j=3$ | $j=4$ | $j=5$ |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 1 | 36.86408 | 43.5854 | 13.88885 | 0.698291 | 757.1208 |
| 2 | 1.122146 | -21.1108 | -5.01303 | 0.92941 | 1936.946 |
| 3 | 8.237129 | 16.10938 | 0.621224 | -0.85892 | 542.6326 |
| 4 | -12.9307 | -22.6645 | -5.93815 | -1.11227 | -1434.1 |
| 5 | 6.808913 | -99.4752 | -48.0558 | -19.4127 | 3782.904 |
| 6 | -2.32757 | -3.37986 | -0.65859 | -0.01346 | -2107.2 |
| 7 | -1.27131 | 23.02414 | 5.516689 | -1.00733 | 1830.376 |
| 8 | 92.68097 | 67.79404 | 24.74579 | -8.87789 | -5801.27 |
| 9 | -0.00145 | -12.4398 | -0.24025 | 0.915931 | -441.728 |
| 10 | -23.9077 | -2.83413 | -4.32022 | 4.655952 | 889.0756 |
| 11 | 87.22948 | 210.8328 | 84.27828 | 20.49507 | -5956.74 |
| 12 | 2.422019 | 1.352431 | 0.200125 | 1.32676 | 197.3057 |
| 13 | -3.8091 | 0.603618 | 0.136124 | -0.14456 | 186.9761 |
| 14 | 14.39132 | 21.07138 | 5.587403 | 0.553288 | 928.3904 |
| 15 | 214.5595 | 187.1196 | 77.68017 | 13.93824 | -5566.52 |
| 16 | -0.5076 | -1.42125 | -0.3352 | -0.24091 | -362.389 |
| 17 | -8.1473 | -13.3361 | -6.45604 | 0.25548 | 2674.753 |
| 18 | 28.99698 | 9.749945 | 2.190639 | -0.76221 | 744.3483 |
| 19 | 1.930444 | 0.243792 | 0.098938 | -0.24601 | -249.547 |
| 20 | -0.25097 | 1.019683 | 0.202941 | 0.15917 | 623.3091 |
| 21 | -0.78139 | 0.39603 | -0.06187 | 0.290823 | 501.6903 |
| 22 | 1.715351 | -5.22229 | -1.06915 | 0.104082 | -423.319 |
| 23 | -64.4817 | -52.9345 | -11.3003 | -2.22152 | -258.824 |
| 24 | 589.2675 | 487.6364 | 131.7514 | 19.03335 | -1411.74 |
| 25 | 1.017367 | 0.551195 | 0.197523 | -0.03168 | 26.06381 |
| 26 | -1.34555 | 2.480126 | -0.25317 | -0.10814 | 88.91014 |
| 27 | -0.8886 | -1.89523 | 0.92278 | -0.49215 | -719.672 |
| 28 | -133.916 | 2.998466 | 33.7118 | -3.65897 | 17439.33 |
| 29 | -20.9438 | -52.6391 | -20.9292 | -5.51298 | 1667.654 |
| 30 | -144.576 | -167.169 | -78.5495 | -34.0728 | 193.4561 |
| 31 | -1.29048 | -0.32594 | -0.04813 | -0.31802 | 40.45531 |
| 32 | 0.591861 | -0.335 | -0.10021 | 0.017771 | -68.8156 |
| 33 | -3.17853 | -4.73227 | -1.16921 | -0.10567 | -197.864 |
| 34 | 45.67663 | 185.8699 | 90.92057 | 77.21417 | 6282.738 |
| 35 | -2.05919 | 7.714543 | 1.858256 | 1.033978 | -434.556 |
| 36 | -110.33 | -161.217 | -47.0354 | -4.15884 | 278.2759 |
| 37 | 12.6594 | 5.346666 | -2.53705 | -1.70713 | -2850.95 |
| 38 | 59.69611 | 113.0206 | 45.40844 | 16.4772 | -1823.15 |
| 39 | 23.80789 | 21.16259 | 5.341529 | 0.722254 | -669.054 |
|  |  |  |  |  |  |


| 40 | 717.5634 | 1427.869 | 498.6612 | 29.5213 | 15604.52 |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 41 | -28.8503 | -388.243 | -144.204 | 75.14435 | -18200.3 |
| 42 | 5.075161 | 0.221491 | 0.909754 | 0.39199 | 3229.457 |
| 43 | -1.24052 | -0.48297 | -0.15689 | 0.073156 | -10.0729 |
| 44 | -3.43744 | 9.845731 | 2.628276 | -1.15241 | 521.6984 |
| 45 | -0.73105 | -0.24287 | -0.08647 | 0.040255 | -1.58552 |
| 46 | 0.912897 | -6.86877 | -0.35866 | 0.553678 | -346.995 |
| 47 | 0.62654 | -0.1356 | -0.02399 | -0.01059 | -8.12201 |
| 48 | -1.40574 | -10.699 | -2.42531 | 0.271089 | 199.4184 |
| 49 | 8.567681 | 10.93099 | 2.956618 | -0.11862 | 460.1588 |
| 50 | -9.60759 | 66.61773 | 29.71416 | 7.804618 | -1825.68 |
| 51 | 8.200975 | -0.13996 | -0.3302 | -0.12668 | 246.1998 |
| 52 | 1.365429 | 3.315668 | 0.395903 | 0.04086 | -42.6699 |
| 53 | -3.57801 | -1.56614 | -0.80667 | 0.327066 | -71.0718 |
| 54 | -10.2922 | -13.0712 | -4.59266 | -0.66753 | 122.9349 |
| 55 | -35.4792 | 16.21102 | 7.720064 | -0.16547 | 941.0548 |
| 56 | -31.1932 | -24.4414 | -9.55596 | 0.788294 | 4011.353 |
| 57 | -8.66778 | -5.90469 | -1.32009 | 0.172407 | -93.0801 |
| 58 | 12.7444 | 18.89714 | 6.16845 | 0.61562 | 1736.494 |
| 59 | 8.664251 | -20.8891 | -13.7026 | 2.168546 | -2436.25 |
| 60 | 7.482441 | -17.3182 | -5.39917 | -0.67983 | 778.2772 |
| 61 | 0.494273 | -0.83321 | -0.04937 | -0.36453 | -574.05 |
| 62 | -18.3336 | -13.6309 | -7.05498 | 1.826218 | 2246.269 |
| 63 | 38.49673 | 42.68334 | 23.72827 | -10.198 | -7962.38 |
| 64 | -335.348 | -237.476 | -80.3895 | -10.5881 | -1662.63 |
| 65 | 82.45436 | 71.22686 | 17.52674 | -7.67133 | -2430.2 |
| 66 | 0.087959 | 0.006823 | 0.003372 | -0.0017 | 0.66021 |
| 67 | -0.31333 | 0.799874 | 0.237152 | -0.01642 | -14.7146 |
| 68 | 0.194336 | -2.53274 | -0.70136 | 0.168854 | -146.547 |
| 69 | 0.535104 | 0.664426 | 0.30513 | 0.122209 | -177.394 |
| 70 | -0.04643 | 0.339608 | 0.08381 | -0.07354 | -75.9305 |
| 71 | 49.63276 | 30.50927 | 9.753522 | 3.23829 | 913.0959 |
| 72 | 6.501381 | -0.05367 | -0.45589 | 0.041592 | -672.268 |
| 73 | 0.307986 | -0.07473 | -0.26794 | -0.00336 | -77.728 |
| 74 | -0.27016 | 0.114285 | 0.089439 | -5.47E-03 | 27.72625 |
| 75 | -0.34941 | -0.05009 | -0.074 | -0.01785 | -193.189 |

Values of $a_{i j k l}$ as $i=1$ and $k=3$

| $l$ | $j=1$ | $j=2$ | $j=3$ | $j=4$ | $j=5$ |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 1 | 18.07127 | 25.02971 | 9.616134 | 4.160998 | -4.03223 |
| 2 | -157.907 | -216.934 | -79.6912 | 19.76385 | 902.4583 |
| 3 | 9.895936 | 9.030124 | 4.935832 | 0.578186 | -47.7689 |
| 4 | 13.51133 | 10.3986 | 0.809594 | -7.67301 | 125.7869 |
| 5 | 75.94297 | 33.53006 | 12.72815 | -1.37914 | -316.005 |
| 6 | -2.39674 | -4.39583 | -1.52341 | -1.94567 | -375.451 |
| 7 | 112.0982 | 154.6691 | 56.58033 | -9.87768 | 808.504 |
| 8 | 94.58635 | 88.68298 | 27.97284 | -57.7296 | -3029.85 |
| 9 | -4.53811 | -5.64882 | -3.42161 | -0.16014 | -67.9081 |
| 10 | -2.07836 | 2.122603 | 0.815855 | -0.80869 | -17.2165 |
| 11 | 24.60806 | 70.39745 | 25.42362 | 21.58584 | 1112.163 |
| 12 | -16.9895 | -32.0159 | -10.5211 | 5.224867 | -413.365 |
| 13 | -0.64402 | 0.172031 | 0.079185 | 0.044532 | 32.85925 |
| 14 | -5.85953 | -5.60976 | 0.113242 | 4.975212 | -96.2862 |
| 15 | 81.07676 | 129.1863 | 53.17686 | -2.06514 | 1809.521 |
| 16 | -1.31872 | -1.39934 | -0.51961 | 0.570616 | 46.87527 |
| 17 | -0.08295 | 2.355686 | 0.812549 | 2.20367 | 119.3403 |
| 18 | 13.93341 | 17.01427 | 6.045673 | 1.064092 | 109.9345 |
| 19 | -0.376 | -0.11741 | 0.024357 | -0.62264 | 39.31706 |
| 20 | 0.541589 | 0.819753 | 0.279408 | 0.204727 | 19.57408 |
| 21 | 11.48162 | 25.67062 | 9.491001 | -3.20658 | 54.57655 |
| 22 | -21.6039 | -28.99 | -10.8775 | 1.723706 | -296.003 |
| 23 | 51.8792 | 169.8718 | 54.43747 | 65.5585 | 1716.395 |
| 24 | 669.7018 | 725.4401 | 271.1229 | 24.92623 | 3025.597 |
| 25 | 0.076059 | 0.036859 | 0.010664 | -0.07234 | -3.33807 |
| 26 | 0.724795 | 0.798268 | 0.666104 | 0.102748 | 23.00142 |
| 27 | -1.33524 | -2.25342 | -0.77867 | -0.14147 | -2.91924 |
| 28 | -43.0235 | -83.6098 | -31.3468 | 27.92367 | -541.659 |
| 29 | -18.0374 | -31.5503 | -11.7312 | -7.41256 | -546.134 |
| 30 | -50.6641 | -44.2117 | -27.1294 | -18.0849 | -2952.26 |
| 31 | -8.77869 | -23.6279 | -8.55814 | -9.29682 | -82.2821 |
| 32 | 0.040322 | -0.02492 | -0.01019 | 0.001787 | -2.30945 |
| 33 | 1.410066 | 1.657418 | 0.06376 | -1.05523 | 28.57041 |
| 34 | -42.6355 | -26.9097 | -11.452 | 10.17504 | 2.847336 |
| 35 | -11.4581 | -20.9218 | -7.48335 | 1.178424 | -82.9801 |
| 36 | -424.159 | -774.718 | -261.532 | -226.625 | -2272.66 |
| 37 | 4.875998 | 9.767018 | 3.59819 | -2.30589 | 62.48544 |
| 38 | 37.75387 | 35.18873 | 17.10641 | 15.1671 | 1391.885 |
| 39 | 62.56366 | 105.3057 | 36.80196 | 31.94514 | 17.86671 |
|  |  |  |  |  |  |


| 40 | 1026.115 | 1394.611 | 436.7217 | 604.1047 | -3166.61 |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 41 | -68.1418 | -140.193 | -47.6585 | -35.9751 | 225.9109 |
| 42 | 2.46895 | -27.6812 | -10.0231 | -1.25396 | 3025.649 |
| 43 | -0.17175 | -0.1159 | -0.03601 | 0.051748 | 4.490649 |
| 44 | 85.5847 | 126.1852 | 46.22406 | -10.117 | 854.5826 |
| 45 | -0.23039 | -0.22608 | -0.07499 | 0.058099 | -3.67047 |
| 46 | -3.36332 | -5.39798 | -2.67435 | -0.3019 | -47.37 |
| 47 | 0.248788 | 0.673348 | 0.216112 | -0.24016 | 5.110019 |
| 48 | -4.86005 | -5.95157 | -2.08956 | -0.2622 | -78.0301 |
| 49 | -0.04128 | 8.182362 | 3.209028 | -1.268 | -202.268 |
| 50 | -15.7411 | 15.1576 | 3.917473 | 2.119562 | 199.9156 |
| 51 | -0.90037 | -1.82482 | -0.62294 | -0.17715 | 20.70215 |
| 52 | 2.692826 | 3.323522 | 1.182358 | 0.02239 | 57.65665 |
| 53 | 23.99773 | 41.36793 | 14.25765 | -1.64548 | 469.0433 |
| 54 | -10.1064 | -18.3036 | -6.40481 | 0.876116 | -185.692 |
| 55 | -0.06706 | -4.00025 | -1.42943 | 1.968361 | 43.88096 |
| 56 | -20.924 | -33.5036 | -11.7769 | 1.724351 | 62.90646 |
| 57 | -1.41518 | -1.90336 | -0.68495 | 0.032677 | -24.9466 |
| 58 | 30.1893 | 72.22127 | 24.4012 | -7.40667 | -238.629 |
| 59 | 0.497066 | 1.238329 | 0.520047 | -1.27044 | -19.633 |
| 60 | -2.55709 | -1.03278 | -0.39677 | -1.15941 | -17.5735 |
| 61 | -43.6161 | -83.0247 | -29.3341 | 7.441716 | 143.6946 |
| 62 | 5.092311 | 99.8803 | 36.52632 | -26.7797 | 597.5182 |
| 63 | 24.92125 | 120.2923 | 54.38853 | 11.61977 | -1542.43 |
| 64 | -390.102 | -312.987 | -104.218 | -111.926 | 1844.554 |
| 65 | 22.41094 | 35.98604 | 12.24406 | 3.685402 | -8.76916 |
| 66 | 0.002555 | -5.77E-05 | -5.43E-05 | -0.00138 | 0.008433 |
| 67 | 0.082399 | 0.124987 | 0.043343 | 0.004995 | 0.219622 |
| 68 | 1.980883 | -7.39875 | -3.35066 | -1.31344 | 162.9126 |
| 69 | 1.576791 | 2.527064 | 0.885242 | -0.07926 | 45.84766 |
| 70 | -4.04805 | -5.65826 | -1.87832 | 0.69341 | -182.34 |
| 71 | 14.80963 | -20.4037 | -8.31403 | 26.42803 | -637.418 |
| 72 | 0.960613 | 1.300443 | 0.456837 | -0.13605 | -8.36214 |
| 73 | 0.333695 | 0.477206 | 0.168678 | -0.02687 | 13.11555 |
| 74 | -0.06192 | -0.10966 | -0.03933 | $1.02 \mathrm{E}-02$ | -1.31802 |
| 75 | 0.126745 | 2.609027 | 0.9806 | 0.447009 | -142.398 |

Values of $a_{i j k l}$ as $i=1$ and $k=4$

| $l$ | $j=1$ | $j=2$ | $j=3$ | $j=4$ | $j=5$ |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 1 | 39.82987 | 52.48462 | 19.3183 | 8.478699 | 405.1424 |
| 2 | -34.9334 | -73.418 | -38.8422 | 74.70213 | 2150.118 |
| 3 | 11.30915 | 17.15405 | 8.217191 | 1.881927 | -484.296 |
| 4 | -13.3135 | -19.6451 | -10.4029 | -12.0053 | -738.332 |
| 5 | 49.35719 | -41.3291 | -3.30491 | -74.4787 | 5549.748 |
| 6 | -20.2013 | -36.3983 | -13.7928 | -1.39927 | -1026.21 |
| 7 | 34.6554 | 80.70037 | 38.42618 | -52.4677 | 16.22862 |
| 8 | 12.03014 | 185.5228 | 48.46564 | 94.26647 | -2043.81 |
| 9 | -3.89794 | -12.9244 | -6.68539 | 1.045489 | 246.9179 |
| 10 | -7.5997 | -66.212 | -19.5446 | -24.7379 | -315.347 |
| 11 | 64.71826 | 161.1955 | 50.88903 | 60.76207 | -3129.47 |
| 12 | 29.34701 | 14.86882 | 6.434619 | 2.244969 | 167.5182 |
| 13 | 0.726471 | 5.161353 | 2.065507 | -0.94209 | 125.4517 |
| 14 | 11.54332 | 13.13172 | 7.253112 | 7.113312 | 559.6755 |
| 15 | 145.3617 | 365.9742 | 128.7272 | 131.0426 | -998.647 |
| 16 | -1.11036 | -1.16385 | -0.25022 | 1.068445 | 95.10265 |
| 17 | 29.25984 | 59.12439 | 21.06662 | -1.10704 | -66.8575 |
| 18 | 19.96626 | 28.50486 | 8.618476 | 1.157484 | 187.6231 |
| 19 | 1.539681 | -3.1314 | -0.95797 | -0.2662 | 119.3777 |
| 20 | 2.516523 | 6.180149 | 2.288268 | 0.192462 | 104.9174 |
| 21 | 8.081755 | 12.50422 | 4.012765 | -1.34682 | 170.3408 |
| 22 | -4.68992 | -14.5089 | -7.22418 | 11.09071 | -70.3999 |
| 23 | -93.4272 | -115.703 | -40.5614 | 67.30781 | 1115.838 |
| 24 | 566.4118 | 66.92813 | 68.30293 | -497.717 | 3175.601 |
| 25 | 0.051883 | -0.03084 | -0.01439 | 0.086917 | -14.9378 |
| 26 | 0.159645 | 2.149671 | 1.307304 | -0.42753 | -57.2585 |
| 27 | -2.73848 | 3.5841 | 1.160585 | 2.136613 | 53.70083 |
| 28 | 56.87068 | 193.3241 | 66.65922 | 97.24947 | -743.762 |
| 29 | -24.1789 | -62.4003 | -21.0126 | -18.7058 | 590.8181 |
| 30 | -82.5728 | -232.173 | -94.5545 | -65.9399 | -1903.11 |
| 31 | -19.8201 | -3.78211 | -1.8869 | -7.74209 | -581.078 |
| 32 | -0.2006 | -1.09275 | -0.42581 | 0.079558 | -17.7825 |
| 33 | -2.17763 | -1.83581 | -1.33869 | -1.19474 | -142.248 |
| 34 | -88.3595 | -113.555 | -37.5683 | -55.7633 | -757.023 |
| 35 | -16.0941 | -27.4677 | -9.82361 | -0.2286 | -207.764 |
| 36 | -298.079 | -224.81 | -78.5042 | -173.237 | -5174.76 |
| 37 | -1.22712 | -23.3786 | -8.11769 | -8.02245 | 247.9222 |
| 38 | 43.15532 | 109.5839 | 44.36843 | 27.9579 | 1301.574 |
| 39 | 76.56112 | 46.70914 | 16.98937 | 20.78243 | 1306.685 |
|  |  |  |  |  |  |


| 40 | 1400.261 | 2299.007 | 737.1123 | 1073.073 | -1389.85 |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 41 | -196.135 | -418.576 | -147.535 | -34.0096 | 76.77042 |
| 42 | 3.252918 | -30.1735 | -11.1654 | 5.449804 | 2684.995 |
| 43 | -0.99035 | -1.03679 | -0.3059 | -0.42218 | 6.984278 |
| 44 | 13.09606 | 38.12571 | 21.55152 | -53.6102 | 837.9632 |
| 45 | -0.81714 | -0.32191 | -0.07729 | -0.17048 | -0.2723 |
| 46 | -3.41193 | -9.28415 | -4.52474 | 2.099466 | 49.90395 |
| 47 | 1.918573 | 0.591155 | 0.1226 | 0.482045 | -15.8518 |
| 48 | -7.01442 | -11.2303 | -4.18999 | 2.04797 | -161.424 |
| 49 | -1.1469 | 9.577415 | 3.833097 | -3.01041 | -331.051 |
| 50 | -13.7486 | 18.6821 | 3.190477 | 9.685795 | -1775.78 |
| 51 | -0.5335 | 1.079569 | -0.30715 | 3.022768 | 24.75432 |
| 52 | 2.862489 | 2.709803 | 0.937199 | -0.28036 | 94.43346 |
| 53 | -2.89756 | 7.884431 | 2.57988 | -3.46642 | 520.3489 |
| 54 | -6.97373 | -2.90145 | -0.672 | -0.70446 | -215.565 |
| 55 | 4.409954 | 5.588265 | 2.538581 | 0.006425 | 388.8223 |
| 56 | -15.4604 | -45.3444 | -16.3561 | 2.907709 | 572.692 |
| 57 | -3.79454 | -2.13441 | -0.47174 | 0.444081 | -46.4812 |
| 58 | 44.97289 | 45.57204 | 14.34165 | 5.359798 | -416.725 |
| 59 | -8.08189 | -22.4659 | -7.53507 | -8.00566 | -164.346 |
| 60 | -4.65382 | -5.90763 | -2.2603 | -1.1278 | -105.409 |
| 61 | -14.8146 | -22.0302 | -7.60377 | 0.842923 | -116.594 |
| 62 | 23.40391 | 20.49046 | 6.913082 | -6.93147 | 660.528 |
| 63 | -57.2219 | 116.4302 | 53.37559 | -19.5073 | -1163.01 |
| 64 | -450.284 | -553.312 | -177.746 | -194.501 | 3219.586 |
| 65 | 1.994232 | 32.65584 | 8.690272 | -8.35108 | 517.0117 |
| 66 | 0.062591 | $2.82 \mathrm{E}-03$ | -2.86E-03 | 0.02269 | -0.27168 |
| 67 | 0.404394 | 0.800989 | 0.306744 | -0.13063 | 2.328679 |
| 68 | 8.290659 | -0.47047 | -0.78849 | 1.368475 | -0.90013 |
| 69 | 0.112276 | -0.05207 | -0.03882 | 0.109751 | 47.08958 |
| 70 | -2.0505 | -1.14351 | -0.27896 | -0.38553 | -122.398 |
| 71 | 83.17833 | 93.32475 | 26.68184 | 68.5245 | -1404.24 |
| 72 | 0.858971 | 1.635329 | 0.549622 | -0.04405 | -54.6214 |
| 73 | -0.04001 | -0.04724 | -0.02166 | -0.06803 | 12.13317 |
| 74 | 0.033351 | 0.023718 | 0.009422 | $2.38 \mathrm{E}-03$ | 0.33256 |
| 75 | -0.89968 | 1.520791 | 0.588433 | -0.15663 | -94.4888 |

Values of $a_{i j k l}$ as $i=1$ and $k=5$

| $l$ | $j=1$ | $j=2$ | $j=3$ | $j=4$ | $j=5$ |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 1 | 19.84074 | 32.39815 | 11.74727 | 5.447394 | 119.5417 |
| 2 | -167.469 | -172.508 | -58.089 | -62.5306 | 1851.276 |
| 3 | 5.427721 | -9.31008 | -3.04996 | 2.198809 | -86.932 |
| 4 | 13.21274 | 1.171337 | -1.05009 | -14.1345 | -100.395 |
| 5 | 74.33839 | 185.4033 | 66.32599 | 20.4493 | 291.1677 |
| 6 | -13.0996 | -6.66109 | -2.925 | 3.174256 | -275.445 |
| 7 | 154.0666 | 227.0923 | 79.95193 | 54.43851 | 66.05346 |
| 8 | 139.5144 | 43.79244 | 7.569063 | -148.364 | -2128.06 |
| 9 | -8.73245 | -10.0384 | -3.95307 | -3.1255 | -33.5266 |
| 10 | 8.997433 | 48.7804 | 18.25897 | 17.4811 | 45.86463 |
| 11 | 32.74686 | -92.0629 | -35.6309 | 18.9186 | -243.184 |
| 12 | -121.522 | -310.845 | -105.02 | -62.7205 | -492.6 |
| 13 | 1.525768 | 4.007537 | 1.445283 | 0.132726 | 29.84675 |
| 14 | -2.42898 | 11.46572 | 5.211132 | 11.17041 | 43.4071 |
| 15 | -3.55265 | -88.3165 | -23.218 | 6.342839 | -316.79 |
| 16 | 2.648622 | -1.55081 | -0.50719 | 0.609508 | 1.195028 |
| 17 | -17.6967 | -74.2891 | -24.7072 | -8.49732 | 131.2431 |
| 18 | 15.34442 | 12.50332 | 4.038069 | 2.667391 | 41.81792 |
| 19 | -0.55371 | 0.395959 | 0.018946 | 0.405622 | 15.43242 |
| 20 | 1.945257 | 3.405858 | 1.216305 | 0.041942 | 15.48407 |
| 21 | 10.73975 | 20.49193 | 7.433546 | -0.25561 | 119.6481 |
| 22 | -25.5188 | -42.0012 | -15.2578 | -8.09347 | -93.4632 |
| 23 | 251.1057 | 972.6941 | 321.0823 | 337.1123 | 274.0776 |
| 24 | 487.83 | 649.3719 | 249.8915 | 206.3852 | -252.481 |
| 25 | -0.0072 | 0.264766 | 0.093769 | 0.05025 | -4.08618 |
| 26 | 0.913781 | 1.063257 | 0.531029 | 0.160203 | 19.48256 |
| 27 | -2.70712 | -6.10951 | -2.11456 | -2.28596 | -2.74353 |
| 28 | -30.9319 | -116.943 | -44.2685 | -10.666 | -224.536 |
| 29 | -27.9163 | 11.18354 | 5.659582 | -14.3126 | -34.1592 |
| 30 | -41.5501 | 118.9893 | 37.78268 | -25.352 | 693.641 |
| 31 | 35.07895 | 35.61885 | 13.56671 | -13.1069 | 135.6095 |
| 32 | -0.12431 | -0.36396 | -0.13129 | 0.00837 | -1.84802 |
| 33 | 1.561424 | -2.74749 | -1.33936 | -1.7864 | -10.9624 |
| 34 | 43.08213 | 15.92684 | -1.52044 | 36.64037 | -393.336 |
| 35 | -16.89 | -13.4058 | -4.71699 | -3.97602 | 30.82217 |
| 36 | -217.94 | -871.138 | -276.139 | -473.933 | 510.2516 |
| 37 | 4.999037 | 14.63565 | 5.030172 | 3.672417 | -7.94765 |
| 38 | 26.83978 | -27.27 | -9.49235 | 18.0252 | -44.2877 |
| 39 | -42.2179 | -50.3968 | -20.4095 | 33.82031 | -319.852 |
|  |  |  |  |  |  |


| 40 | 715.4577 | 337.684 | 52.4631 | 496.8071 | -1315.2 |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 41 | -56.2672 | -51.7968 | -11.2658 | -52.4211 | 318.6196 |
| 42 | -43.2618 | -131.745 | -45.6082 | 3.827817 | 3083.335 |
| 43 | 0.523359 | 1.077947 | 0.372788 | 0.427772 | 3.997095 |
| 44 | 144.5738 | 190.2301 | 64.81265 | 47.2271 | 317.2888 |
| 45 | -0.82802 | -1.11751 | -0.38072 | -0.25927 | -0.57888 |
| 46 | -6.54711 | -6.62343 | -2.41929 | -2.70398 | -37.6131 |
| 47 | 4.204818 | 5.393185 | 1.886617 | 1.301147 | -15.3259 |
| 48 | -4.11526 | -5.09513 | -1.72158 | 0.560542 | -52.5307 |
| 49 | 11.29168 | 26.37372 | 9.247135 | -3.02715 | -173.536 |
| 50 | -16.3512 | -25.3022 | -10.6346 | -6.32027 | -15.9637 |
| 51 | -3.56886 | -6.87172 | -2.32059 | -2.61204 | 15.6699 |
| 52 | 3.297558 | 4.328771 | 1.437832 | 0.909724 | 39.33394 |
| 53 | -15.1346 | -4.28057 | -2.64269 | -32.0683 | 801.9216 |
| 54 | -18.6737 | -26.0179 | -9.09711 | -3.58061 | -56.3524 |
| 55 | -0.73683 | -2.5807 | -0.89374 | -2.31392 | 64.76789 |
| 56 | -43.868 | -73.6251 | -24.8932 | 2.881102 | 127.1112 |
| 57 | -0.757 | -0.73222 | -0.26175 | -0.01316 | -8.46061 |
| 58 | 157.7653 | 201.8565 | 72.40885 | 61.71479 | -710.523 |
| 59 | 3.763465 | 9.206854 | 3.148958 | 2.868375 | -19.6992 |
| 60 | -2.26786 | -2.56818 | -0.77837 | -0.99489 | -37.4938 |
| 61 | -64.4857 | -68.1859 | -23.8777 | -11.4154 | -129.39 |
| 62 | -57.7975 | -151.014 | -51.681 | -4.63245 | 1137.691 |
| 63 | 172.584 | 638.094 | 237.5298 | -61.1947 | -517.133 |
| 64 | -365.979 | -24.5197 | -6.69333 | -81.3084 | 963.229 |
| 65 | 23.13317 | 21.08895 | 6.245276 | 7.059053 | 62.43948 |
| 66 | 0.002551 | $4.21 \mathrm{E}-03$ | $1.53 \mathrm{E}-03$ | -0.00074 | -0.06823 |
| 67 | 0.080147 | 0.088992 | 0.029264 | 0.016812 | -0.04438 |
| 68 | 0.119557 | -29.096 | -11.7193 | -2.87793 | 3.602539 |
| 69 | 1.725124 | 2.490168 | 0.877733 | -0.01216 | 31.58673 |
| 70 | -6.42603 | -5.65638 | -1.91152 | 1.838373 | -201.984 |
| 71 | -9.5009 | -82.6123 | -26.9522 | -5.94437 | -262.838 |
| 72 | 1.927897 | 2.850983 | 0.953238 | 0.420998 | -7.21699 |
| 73 | 0.15208 | 0.122275 | 0.043852 | -0.03693 | 7.278131 |
| 74 | -0.11016 | -0.16294 | -0.05592 | -1.21E-02 | -0.67359 |
| 75 | 2.855444 | 10.11215 | 3.721186 | 1.986668 | -114.085 |

Values of $a_{i j k l}$ as $i=1$ and $k=6$

| $l$ | $j=1$ | $j=2$ | $j=3$ | $j=4$ | $j=5$ |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 1 | 48.4471 | 3.289265 | 2.474363 | 11.57418 | -46.6053 |
| 2 | 123.8078 | -16.4851 | -19.9127 | 70.86355 | 1076.191 |
| 3 | -1.97083 | -39.5595 | -15.6001 | 12.49456 | -355.44 |
| 4 | -29.9215 | 15.46719 | 3.303958 | -26.6595 | 206.165 |
| 5 | -11.5416 | 72.16892 | 32.73848 | -4.86812 | 2674.453 |
| 6 | 12.68362 | 23.90474 | 7.831505 | -3.83952 | -905.037 |
| 7 | 16.46896 | 227.4591 | 92.4862 | -39.4699 | 1015.942 |
| 8 | -102.885 | -612.379 | -221.083 | -195.766 | -103.898 |
| 9 | -5.59734 | -15.7467 | -6.17919 | -1.10627 | 34.84063 |
| 10 | 33.2988 | 256.2497 | 103.6123 | -49.3471 | 362.89 |
| 11 | 103.4602 | 111.2433 | 24.00863 | 64.36499 | -2169.82 |
| 12 | -265.534 | -587.607 | -213.864 | 47.52657 | -584.421 |
| 13 | 1.225871 | 12.76985 | 5.203621 | -2.46993 | 140.8611 |
| 14 | 17.87416 | -4.31544 | 0.384766 | 16.30405 | -70.1776 |
| 15 | 221.2872 | -1.82528 | 21.00567 | -12.7127 | -2348.19 |
| 16 | -13.48 | -10.9805 | -3.68739 | -1.24645 | 30.95652 |
| 17 | -35.6574 | -139.2 | -50.555 | 19.56505 | -152.679 |
| 18 | 7.518109 | 0.722682 | 0.322658 | 2.242278 | 18.48624 |
| 19 | 6.365367 | 6.563523 | 2.387256 | -0.69798 | 78.42567 |
| 20 | 0.781167 | 5.083522 | 1.945399 | -0.26046 | 125.529 |
| 21 | 6.899849 | -2.01124 | -0.33056 | -1.80595 | 84.98486 |
| 22 | 9.678782 | -33.2765 | -14.7925 | 10.79551 | -254.385 |
| 23 | 282.5861 | 1568.916 | 585.4667 | -44.3059 | 656.3896 |
| 24 | 692.5782 | 999.8489 | 339.4673 | 326.2268 | -1533.76 |
| 25 | 0.925283 | 1.204984 | 0.402776 | 0.179429 | -1.22812 |
| 26 | 0.421367 | 1.392757 | 0.607447 | -0.31343 | -9.20404 |
| 27 | -5.73789 | -48.4033 | -19.1852 | 7.198182 | -91.9822 |
| 28 | 7.358227 | -245.779 | -114.386 | 158.5989 | -977.845 |
| 29 | -40.1439 | -46.2892 | -11.935 | -28.6703 | 501.4929 |
| 30 | -122.254 | -26.73 | -3.66314 | -71.9008 | 1203.236 |
| 31 | 212.0197 | 254.5237 | 88.74095 | -29.7051 | 215.6928 |
| 32 | -0.21042 | -1.20876 | -0.4802 | 0.207935 | -21.2808 |
| 33 | -2.49844 | 3.166634 | 0.539003 | -2.45398 | 6.604598 |
| 34 | -14.0675 | 116.4108 | 12.44232 | 98.68894 | 785.4682 |
| 35 | -3.0047 | -0.19139 | -0.3618 | 1.363305 | -50.6047 |
| 36 | 63.40227 | -1222.41 | -463.578 | -149.669 | -174.682 |
| 37 | 11.26723 | 82.23344 | 32.3103 | -15.253 | 224.639 |
| 38 | 41.18144 | 17.59205 | 4.547068 | 37.51346 | -171.845 |
| 39 | -377.561 | -451.848 | -158.145 | 66.95714 | -413.153 |
|  |  |  | 12 |  |  |


| 40 | 874.0497 | 1495.829 | 518.7821 | 450.0894 | -1543.83 |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 41 | -135.986 | -326.213 | -99.6037 | -162.176 | 659.2775 |
| 42 | -0.22197 | -5.45526 | -4.56787 | 10.46467 | 2865.635 |
| 43 | 0.429368 | 6.246041 | 2.365654 | -0.42538 | 10.48474 |
| 44 | -23.0134 | 152.8015 | 63.75495 | -40.8216 | 1222.058 |
| 45 | -1.01342 | -2.07791 | -0.65975 | -0.03803 | -10.7853 |
| 46 | -3.95983 | -11.4867 | -4.73407 | -1.60177 | -9.21521 |
| 47 | 4.123743 | 7.823722 | 2.395952 | -0.42458 | -10.2518 |
| 48 | -5.18239 | -1.10722 | -0.56132 | 0.26837 | -8.1464 |
| 49 | 3.816743 | 1.939772 | 1.271621 | -1.21125 | -437.982 |
| 50 | -19.9313 | 35.22723 | 7.525406 | -4.50462 | -588.68 |
| 51 | -6.28571 | -37.9401 | -14.131 | 1.699657 | -17.8088 |
| 52 | 2.528914 | 4.875621 | 1.713275 | 0.750364 | 69.9844 |
| 53 | -24.287 | -72.0193 | -24.1357 | -9.24247 | 259.1764 |
| 54 | -5.83651 | -16.6242 | -5.25973 | -0.4291 | -19.0908 |
| 55 | -4.75923 | -20.4757 | -6.93168 | -1.24449 | 60.1661 |
| 56 | -28.526 | -21.3271 | -7.57073 | 3.097943 | 645.2413 |
| 57 | -0.87796 | -0.21535 | -0.02055 | -0.31952 | -10.2615 |
| 58 | 96.33755 | 199.1223 | 66.36466 | 10.94873 | -242.175 |
| 59 | 3.459128 | 42.82872 | 16.05831 | -3.65409 | 44.89201 |
| 60 | -2.30875 | -2.68084 | -1.11218 | -1.58307 | -84.497 |
| 61 | -27.8644 | -15.2295 | -5.75364 | -1.52552 | -34.6963 |
| 62 | 51.74896 | -17.5916 | -5.94312 | 5.266354 | 423.5252 |
| 63 | -73.9337 | 7.897402 | 22.10302 | -46.1193 | -1256.88 |
| 64 | -535.491 | -150.2 | -42.1126 | -211.786 | 2526.912 |
| 65 | -7.56158 | 3.7845 | 2.161664 | 3.482756 | 3.106267 |
| 66 | 0.03575 | $3.59 \mathrm{E}-03$ | -3.31E-03 | -0.00542 | 0.198483 |
| 67 | 0.413656 | 0.632868 | 0.230409 | 0.058681 | -2.33031 |
| 68 | 13.09682 | -3.27119 | -3.24912 | -2.86066 | -126.824 |
| 69 | 0.069282 | 0.059318 | -0.00687 | 0.023986 | 16.23662 |
| 70 | -4.66641 | -1.45845 | -0.03509 | 1.101372 | -82.5214 |
| 71 | 123.0778 | -20.5035 | -12.3047 | 66.85362 | -1067.71 |
| 72 | 1.837215 | 3.288998 | 1.148169 | 0.146074 | -19.511 |
| 73 | 0.109104 | -0.06322 | -0.03146 | -0.04274 | 4.038054 |
| 74 | -0.04485 | -0.08104 | -0.02705 | $3.12 \mathrm{E}-05$ | 0.602433 |
| 75 | -1.30703 | 2.352394 | 1.269008 | 0.796669 | -42.3428 |

Values of $a_{i j k l}$ as $i=2$ and $k=1$

| $l$ | $j=1$ | $j=2$ | $j=3$ | $j=4$ | $j=5$ |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 1 | 66.56627 | 101.6981 | 33.67868 | 5.290572 | 263.5964 |
| 2 | -190.598 | 7.393353 | -49.3396 | 0.607344 | 1924.034 |
| 3 | 53.47089 | 23.0294 | 8.498964 | 0.368753 | 81.89793 |
| 4 | 12.74035 | -27.8902 | -8.30025 | -5.01055 | -680.559 |
| 5 | 45.18042 | -521.797 | -37.3591 | -111.563 | 6215.397 |
| 6 | -5.15662 | -2.0598 | -0.59457 | 0.497123 | -420.539 |
| 7 | 154.1281 | -4.15222 | 40.22782 | -7.7979 | 829.7971 |
| 8 | 315.3973 | 384.2145 | 138.2385 | 14.81604 | -5382.26 |
| 9 | -38.7555 | -17.8222 | -6.37882 | 1.051681 | -259.435 |
| 10 | 13.33473 | -40.2008 | -15.6455 | 1.042801 | -786.518 |
| 11 | 69.86974 | 697.0086 | 121.0843 | 115.3213 | -3948.6 |
| 12 | -12.165 | 3.461987 | -0.411 | 5.463071 | -1745.19 |
| 13 | -1.55415 | 0.025529 | -0.02578 | -0.30344 | 71.14845 |
| 14 | 8.70773 | 34.89843 | 10.3857 | 3.655751 | 485.8344 |
| 15 | 23.96361 | 624.2946 | 251.8862 | -31.1537 | 2170.215 |
| 16 | -0.69132 | -1.29411 | -0.50215 | 0.475186 | 124.6533 |
| 17 | 20.78306 | -2.85763 | -1.43914 | -3.2457 | 182.255 |
| 18 | 16.13307 | 32.56285 | 9.85576 | 0.099176 | 144.0579 |
| 19 | 1.384709 | 0.369368 | 0.153885 | -0.14867 | 49.76455 |
| 20 | 0.264803 | 0.219226 | 0.058613 | -0.0196 | 18.75026 |
| 21 | 5.179622 | -4.34754 | -0.33554 | -1.30718 | -96.3383 |
| 22 | -33.9376 | 10.09999 | -7.8498 | 1.176197 | -381.446 |
| 23 | -85.6475 | -204.023 | -62.153 | -23.253 | 6207.135 |
| 24 | 980.4376 | 1763.164 | 506.2174 | 48.56657 | -15243 |
| 25 | 0.808244 | 0.481854 | 0.160707 | -0.08843 | -9.33947 |
| 26 | 8.303521 | 2.906501 | 1.086518 | -0.10643 | 62.0952 |
| 27 | -2.26833 | -1.30275 | -0.20611 | 0.106147 | 83.37455 |
| 28 | -121.534 | -238.624 | -43.6283 | 46.96081 | 6392.332 |
| 29 | -19.0939 | -184.2 | -31.3955 | -28.7704 | 1143.654 |
| 30 | -14.1177 | -300.592 | -125.805 | -36.2541 | 145.9062 |
| 31 | 3.028865 | 1.460061 | 0.609112 | -1.76186 | 522.5568 |
| 32 | 0.046622 | -0.05499 | -0.01482 | 0.032537 | -4.82179 |
| 33 | -2.37525 | -8.39174 | -2.43371 | -0.74109 | -94.9504 |
| 34 | 126.3272 | -38.5704 | -24.7022 | 306.9854 | -2448.83 |
| 35 | -28.6588 | -6.35962 | -2.91716 | 1.412526 | -168.062 |
| 36 | -157.707 | -397.512 | -131.167 | -71.3487 | 2285.079 |
| 37 | 21.24627 | 25.6907 | 9.169274 | -4.32377 | -466.009 |
| 38 | 59.65472 | 207.8429 | 80.5014 | 42.54838 | 786.9151 |
| 39 | 16.55637 | 68.82253 | 20.79457 | 15.32303 | -2278.34 |
|  |  |  | 12 |  |  |


| 40 | 1257.045 | 3263.231 | 1264.942 | 705.0854 | 433.3496 |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 41 | -265.156 | -1193.77 | -487.407 | 11.09277 | -3047.85 |
| 42 | 15.23953 | 10.47747 | 4.395029 | -0.0025 | 6717.361 |
| 43 | -1.4181 | -1.49862 | -0.50694 | 0.160488 | 19.80756 |
| 44 | 92.22183 | -1.01242 | 21.8127 | -3.42369 | 856.2535 |
| 45 | -1.06764 | -0.76362 | -0.24686 | 0.101328 | -3.77889 |
| 46 | -17.7707 | -9.27181 | -3.4728 | 0.239347 | -206.224 |
| 47 | 0.237378 | 0.534369 | 0.209261 | -0.28077 | 1.127539 |
| 48 | -7.84644 | -16.7945 | -5.834 | 1.496096 | -350.262 |
| 49 | 11.9613 | 18.70488 | 7.123212 | -1.6018 | -257.067 |
| 50 | 101.7609 | 285.6179 | 43.60003 | 51.48126 | -2068.67 |
| 51 | -13.4533 | -1.59307 | -0.4295 | -0.53074 | 322.3641 |
| 52 | 4.729971 | 6.559465 | 2.268866 | -0.89617 | 174.6625 |
| 53 | 2.432952 | -3.69622 | -0.77428 | -1.72227 | -129.149 |
| 54 | -29.6538 | -22.6863 | -8.31956 | 0.954462 | -607.381 |
| 55 | -38.5543 | -29.9371 | -9.71485 | 3.134159 | 1107.286 |
| 56 | -38.7486 | -102.665 | -38.8092 | -9.77519 | -1001.87 |
| 57 | -3.53033 | -12.193 | -3.76768 | 0.84628 | -152.257 |
| 58 | 101.2306 | 79.83111 | 26.62059 | -17.0986 | -510.882 |
| 59 | 58.79701 | 6.370585 | -3.48913 | 0.320662 | -2039.23 |
| 60 | 5.959124 | -12.2388 | -3.73916 | -2.8848 | -207.074 |
| 61 | -19.4131 | -12.141 | -4.92585 | 5.670814 | 756.1258 |
| 62 | -89.0563 | -107.433 | -38.5308 | -17.9689 | -451.645 |
| 63 | 47.45986 | 412.1615 | 163.9448 | 5.013535 | 2024.446 |
| 64 | 31.63301 | -715.431 | -218.49 | -106.891 | 4646.866 |
| 65 | -118.908 | 206.7328 | 57.15116 | 5.458461 | -638.13 |
| 66 | 0.025428 | $1.01 \mathrm{E}-02$ | 3.65E-03 | -0.00259 | -0.36744 |
| 67 | 0.449433 | 0.687812 | 0.226613 | -0.05463 | 3.38807 |
| 68 | -2.5916 | -8.38982 | -3.0113 | 2.551375 | 8.303211 |
| 69 | 1.116823 | 0.970615 | 0.440887 | -0.01482 | 128.2253 |
| 70 | -1.17749 | -0.09557 | -0.14515 | 0.338941 | -187.824 |
| 71 | -79.7883 | 18.417 | 4.499892 | 12.44272 | -368.436 |
| 72 | 4.913806 | 7.681183 | $2.591281$ | 0.26952 | -77.3036 |
| 73 | -1.04727 | 0.500957 | $0.194854$ | -0.15401 | 43.69029 |
| 74 | 0.141008 | -0.06165 | -0.02691 | $1.59 \mathrm{E}-02$ | -4.16398 |
| 75 | -0.70319 | -0.81806 | -0.3275 | -0.13246 | -237.211 |

Values of $a_{i j k l}$ as $i=2$ and $k=2$

| $l$ | $j=1$ | $j=2$ | $j=3$ | $j=4$ | $j=5$ |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 1 | 119.2682 | 122.4639 | 42.262 | 2.34039 | -5.98622 |
| 2 | 60.57097 | -103.265 | -32.4321 | 3.023934 | -4156.86 |
| 3 | 76.73049 | 16.29997 | 22.62491 | -5.8697 | 1174.782 |
| 4 | -91.8414 | -66.3721 | -24.2769 | -1.67035 | 244.4165 |
| 5 | 33.58607 | 4.645917 | -57.5223 | -84.5374 | 5237.993 |
| 6 | 7.169061 | -4.50428 | -2.09615 | 0.985611 | -2604.72 |
| 7 | -18.0465 | 80.25065 | 26.77205 | -3.80558 | 6721.734 |
| 8 | 114.8864 | 227.833 | 89.9513 | -6.31236 | -1462.36 |
| 9 | -47.7436 | -8.47953 | -16.9525 | 5.190195 | -753.819 |
| 10 | -16.4192 | -84.1687 | -23.6432 | 2.259214 | 1086.115 |
| 11 | 206.9404 | 375.5322 | 170.1047 | 89.63116 | -5276.31 |
| 12 | 5.585289 | 3.781906 | 2.120505 | 1.493655 | 41.35942 |
| 13 | -13.9003 | 1.240058 | 0.031031 | -0.22493 | 129.098 |
| 14 | 84.80828 | 65.7082 | 22.8927 | 0.61765 | -421.851 |
| 15 | -244.047 | 716.324 | 277.8906 | 92.27567 | 300.3392 |
| 16 | -21.1378 | -3.91653 | -1.58266 | 0.191849 | 73.43496 |
| 17 | -107.653 | -27.3254 | -11.8232 | -2.99363 | 626.6996 |
| 18 | 47.1829 | 69.96917 | 19.39032 | 0.029486 | -4.80271 |
| 19 | -1.80006 | 1.192075 | -0.20178 | -0.06282 | 7.564053 |
| 20 | 4.823559 | 1.658739 | 1.032872 | -0.19315 | 561.7323 |
| 21 | 0.684848 | -1.76607 | -0.0741 | 0.092534 | -48.503 |
| 22 | 12.86314 | -13.7628 | -5.0286 | 0.127041 | -1772.51 |
| 23 | -62.4461 | -114.207 | -39.4493 | -7.94823 | 605.8873 |
| 24 | 966.3743 | 1453.283 | 488.8577 | -39.7415 | 11441.02 |
| 25 | 8.806881 | 1.866679 | 0.729747 | -0.32922 | 9.125743 |
| 26 | 5.970941 | -1.61527 | 3.423406 | -0.95436 | 112.0471 |
| 27 | -6.32269 | -4.68104 | -1.08855 | 0.766231 | -120.4 |
| 28 | -73.631 | 116.226 | 21.92894 | 72.70131 | -11442.1 |
| 29 | -55.3626 | -104.46 | -45.556 | -25.6933 | 1550.309 |
| 30 | 488.3896 | -452.908 | -183.06 | -132.159 | -1915.79 |
| 31 | 2.057026 | -0.81989 | -0.2252 | -0.69968 | 16.8841 |
| 32 | 0.394388 | -0.90091 | -0.28822 | 0.126002 | -41.952 |
| 33 | -19.6846 | -15.9319 | -5.49126 | 0.035747 | 178.4985 |
| 34 | -16.6702 | 229.4072 | 119.0769 | 268.4895 | -2030.6 |
| 35 | 66.00195 | -2.11688 | -0.6765 | 3.098906 | -410.675 |
| 36 | -246.292 | -501.559 | -169.34 | -33.236 | 2257.262 |
| 37 | -16.0457 | 36.44516 | 18.77307 | -21.0317 | 1277.457 |
| 38 | -93.0486 | 313.7953 | 115.5303 | 74.42121 | 1408.313 |
| 39 | 23.63438 | 56.96346 | 19.14499 | 5.090121 | -327.094 |
|  |  |  |  |  |  |


| 40 | 2463.654 | 4718.767 | 1609.361 | 517.0335 | -27323.9 |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 41 | -675.906 | -2364.49 | -818.6 | 205.0992 | 8770.973 |
| 42 | 29.84501 | 6.728473 | 2.733652 | -0.08721 | 6193.44 |
| 43 | -2.82359 | -2.10666 | -0.77369 | 0.41645 | -6.40519 |
| 44 | -54.0651 | 43.85166 | 12.79355 | -2.7549 | 3757.164 |
| 45 | -1.43381 | -0.73432 | -0.25298 | 0.091473 | -2.19715 |
| 46 | -19.2655 | -8.49312 | -9.09662 | 2.450216 | -668.142 |
| 47 | 0.244847 | -0.11973 | -0.00169 | -0.07685 | -11.838 |
| 48 | -22.5462 | -30.6194 | -8.73198 | 1.899971 | 41.7358 |
| 49 | 27.39865 | 18.28013 | 5.940247 | -2.54719 | -90.7245 |
| 50 | 3.140711 | 67.35295 | 51.71151 | 31.91432 | -2163.78 |
| 51 | 11.00282 | 8.209723 | 0.066285 | 2.575591 | -190.671 |
| 52 | 10.62556 | 10.96863 | 2.719208 | -0.2635 | 46.28029 |
| 53 | -21.5701 | -7.65017 | -2.09055 | 0.155201 | -130.289 |
| 54 | -37.3384 | -26.0051 | -7.5973 | -0.09827 | -40.6897 |
| 55 | 89.47518 | 36.06259 | 5.60292 | 5.92274 | -1023.73 |
| 56 | -20.0016 | -68.1122 | -27.828 | -3.31476 | -144.123 |
| 57 | -21.255 | -24.5866 | -6.73334 | 1.215764 | -80.3161 |
| 58 | 82.55718 | 60.75314 | 17.61563 | -2.96724 | -700.203 |
| 59 | -126.875 | -126.141 | -35.832 | -15.0496 | 3207.73 |
| 60 | -46.7861 | -35.386 | -10.5564 | -4.42537 | -225 |
| 61 | -6.67053 | -1.33602 | -1.24355 | 0.600739 | 107.5556 |
| 62 | -51.6085 | -61.3314 | -23.2648 | 4.21105 | -1343.65 |
| 63 | 4.655095 | 186.043 | 79.42624 | -32.6324 | 1308.215 |
| 64 | -723.251 | -869.183 | -271.911 | -82.1516 | 1596.234 |
| 65 | 428.0929 | 376.9083 | 108.0509 | -5.77827 | -167.632 |
| 66 | 0.102739 | $1.55 \mathrm{E}-02$ | $9.22 \mathrm{E}-03$ | -0.00532 | 0.281318 |
| 67 | 1.402037 | 2.064184 | 0.722737 | -0.27858 | 3.786491 |
| 68 | -5.45777 | -6.11087 | -1.64044 | 1.685243 | -49.1488 |
| 69 | 0.86961 | 1.174441 | 0.314959 | 0.032989 | 25.27048 |
| 70 | 1.160526 | 0.8164 | 0.18682 | 0.011703 | -125.92 |
| 71 | 162.4555 | 74.66074 | 22.18579 | 12.8434 | -949.954 |
| 72 | -4.66636 | 0.130885 | 0.941555 | 0.024817 | 176.5069 |
| 73 | -1.04419 | 0.534672 | 0.042951 | -0.1015 | 9.206418 |
| 74 | 0.236292 | 0.06068 | 0.025773 | $9.98 \mathrm{E}-03$ | -3.83087 |
| 75 | -1.8817 | -0.68265 | -0.22343 | -0.00867 | -197.163 |

Values of $a_{i j k l}$ as $i=2$ and $k=3$

| $l$ | $j=1$ | $j=2$ | $j=3$ | $j=4$ | $j=5$ |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 1 | 28.36362 | 38.00989 | 13.27718 | 3.319324 | 151.2625 |
| 2 | -781.932 | -561.802 | -224.261 | -105.603 | 1653.619 |
| 3 | 30.27556 | 15.01958 | 6.322657 | 1.762615 | -1.59243 |
| 4 | 60.04928 | 88.50135 | 27.67189 | 3.56497 | -366.544 |
| 5 | 395.5943 | -13.0445 | 22.52135 | 41.90993 | 941.111 |
| 6 | -8.85894 | -4.22009 | -1.17925 | -11.2724 | -298.713 |
| 7 | 601.642 | 416.2318 | 167.3938 | 77.357 | -164.236 |
| 8 | 160.4804 | 106.1773 | 47.6564 | -342.397 | -1509.47 |
| 9 | -24.9837 | -18.2773 | -7.10793 | -1.45017 | -58.1741 |
| 10 | 12.27432 | 26.14097 | 9.721429 | 17.27675 | -250.999 |
| 11 | -208.913 | 230.5849 | 57.24042 | 72.25996 | -303.812 |
| 12 | -80.8638 | -95.0531 | -31.5144 | 19.15198 | -452.558 |
| 13 | 0.011543 | 1.846329 | 0.575456 | 0.894402 | 24.39321 |
| 14 | -23.5797 | -55.1258 | -16.7388 | -3.46958 | 255.6616 |
| 15 | -114.695 | 140.6762 | 68.40811 | -100.465 | 2525.441 |
| 16 | -7.80646 | -15.0998 | -5.54751 | 2.285082 | 85.17182 |
| 17 | 1.30597 | -17.1819 | -6.92067 | 17.57015 | 169.1318 |
| 18 | 27.21237 | 41.32249 | 13.34745 | 8.133574 | 64.40491 |
| 19 | -1.85702 | -1.0432 | -0.22848 | -2.48532 | 23.2507 |
| 20 | 2.103476 | 2.570749 | 0.879495 | 1.036396 | 12.32049 |
| 21 | 53.70935 | 108.3247 | 37.22416 | -0.13439 | -39.7832 |
| 22 | -128.957 | -72.4489 | -30.6605 | -14.0488 | -98.7802 |
| 23 | 196.9969 | 121.347 | 20.44145 | 167.353 | 1924.343 |
| 24 | 1618.315 | 3281.473 | 1105.83 | 1226.949 | -4262.38 |
| 25 | 0.26965 | 0.708344 | 0.285734 | -0.23339 | -5.90382 |
| 26 | 6.369099 | 3.336489 | 1.29386 | 0.077481 | 23.25601 |
| 27 | -3.14805 | -5.20591 | -1.69242 | -2.02644 | 13.56375 |
| 28 | -189.968 | -472.026 | -164.922 | -42.7504 | 734.7813 |
| 29 | 36.76748 | -89.8676 | -25.3168 | -28.513 | -86.8517 |
| 30 | 154.2662 | -24.3909 | -31.6574 | -6.56642 | -3833.71 |
| 31 | -5.02407 | -25.5551 | -6.20529 | -30.665 | -17.504 |
| 32 | -0.05589 | -0.25883 | -0.08895 | -0.05514 | -1.43939 |
| 33 | 2.525367 | 12.56082 | 3.790081 | 1.245554 | -59.3525 |
| 34 | 71.60179 | 153.4185 | 40.75404 | 66.87542 | -288.473 |
| 35 | -33.1605 | -52.8051 | -17.6436 | -7.62031 | -81.9017 |
| 36 | -768.873 | -863.623 | -234.098 | -640.718 | -2236.58 |
| 37 | 16.22186 | 40.40906 | 14.03456 | 0.129599 | -26.6146 |
| 38 | -8.49053 | 18.25199 | 13.30516 | 39.65477 | 1534.729 |
| 39 | 71.29848 | 129.9461 | 35.55438 | 97.89866 | -23.1929 |
|  |  |  |  |  |  |


| 40 | 423.0773 | -1683.66 | -690.457 | 250.823 | 4925.173 |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 41 | -121.765 | -154.254 | -37.7113 | -127.727 | -37.8431 |
| 42 | -22.7036 | -129.039 | -36.3902 | -58.1291 | 7196.072 |
| 43 | -0.404 | -0.20063 | -0.06614 | 0.379312 | 1.770017 |
| 44 | 434.6671 | 381.405 | 143.1538 | 88.12037 | 245.9121 |
| 45 | -0.75515 | -0.80206 | -0.27983 | 0.255894 | -5.45948 |
| 46 | -13.3311 | -13.7661 | -5.0748 | -4.10925 | -47.6898 |
| 47 | 1.707805 | 4.452983 | 1.723883 | -1.84924 | 18.91207 |
| 48 | -16.3518 | -22.2016 | -7.34043 | -4.01366 | -47.706 |
| 49 | 8.507343 | 60.66037 | 20.72973 | 4.446486 | -462.164 |
| 50 | -46.8904 | 120.493 | 28.23627 | 14.15403 | -615.635 |
| 51 | -4.4168 | -7.73708 | -2.70752 | -3.16517 | 57.11013 |
| 52 | 9.330991 | 10.75871 | 3.523633 | 1.663238 | 47.51993 |
| 53 | 86.73748 | 162.2158 | 48.18535 | 14.17388 | 286.1168 |
| 54 | -34.1836 | -67.592 | -22.5485 | -4.00268 | -154.399 |
| 55 | -2.20426 | -23.2466 | -8.18137 | 4.341882 | 85.81952 |
| 56 | -84.2393 | -195.723 | -65.7463 | -16.5791 | 481.3756 |
| 57 | -3.59626 | -5.33944 | -1.72285 | -0.75558 | -23.2949 |
| 58 | 149.3369 | 344.0559 | 130.4885 | -31.5065 | -330.727 |
| 59 | 10.46605 | 22.53654 | 7.834974 | 1.56865 | -73.2157 |
| 60 | -6.10603 | 3.081599 | 1.174524 | -4.41856 | -9.03871 |
| 61 | -187.173 | -324.226 | -113.127 | -13.307 | 477.2938 |
| 62 | 146.4683 | 374.4672 | 135.9129 | -33.819 | 38.73004 |
| 63 | 147.8411 | 839.1747 | 301.3496 | 297.8771 | -4632.68 |
| 64 | 38.33855 | 58.66051 | 38.15899 | -199.276 | 728.5345 |
| 65 | 39.83504 | 90.73224 | 27.41077 | 47.82257 | -208.007 |
| 66 | 0.004313 | -5.64E-03 | -1.57E-03 | -0.01214 | 0.131535 |
| 67 | 0.286316 | 0.47368 | 0.159135 | 0.101619 | -0.99386 |
| 68 | -0.81862 | -43.0577 | -15.2548 | -29.4836 | 454.1127 |
| 69 | 6.038978 | 11.7958 | 3.962439 | 0.942531 | 59.95557 |
| 70 | -16.6387 | -30.4695 | -10.7208 | 5.859674 | -342.602 |
| 71 | -225.903 | -282.932 | -98.9111 | -40.7662 | -158.668 |
| 72 | 4.528052 | 8.655871 | 2.908727 | 0.856827 | $-29.852$ |
| 73 | 1.271537 | 2.362704 | 0.79315 | 0.149901 | 16.36006 |
| 74 | -0.31835 | -0.7486 | -0.25365 | -5.10E-02 | -1.15464 |
| 75 | 2.269072 | 10.631 | 3.424216 | 6.795004 | -334.548 |

Values of $a_{i j k l}$ as $i=2$ and $k=4$

| $l$ | $j=1$ | $j=2$ | $j=3$ | $j=4$ | $j=5$ |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 1 | 63.54933 | 94.40837 | 36.47133 | 18.66981 | 79.01635 |
| 2 | -196.642 | -315.723 | -149.011 | 208.6868 | 815.1846 |
| 3 | -36.3444 | 25.98999 | 17.98194 | -0.53772 | -186.859 |
| 4 | 33.78924 | 21.48121 | -3.30942 | -21.8127 | 84.70566 |
| 5 | 250.9765 | -246.861 | -30.0298 | -95.0671 | 339.493 |
| 6 | -20.7843 | -78.8536 | -29.1934 | -30.0161 | -1339.67 |
| 7 | 226.0599 | 294.4711 | 143.724 | -110.3 | 694.9239 |
| 8 | -111.807 | -198.563 | -132.81 | -47.2478 | 885.1247 |
| 9 | 44.38347 | -24.4157 | -17.5369 | -0.90649 | 130.0338 |
| 10 | -120.31 | -117.83 | -29.607 | -45.7684 | -362.532 |
| 11 | 128.4857 | 701.5377 | 210.6497 | 103.6049 | 864.2317 |
| 12 | -33.1578 | -53.7029 | -14.6017 | 23.39098 | 312.0661 |
| 13 | -1.93407 | 14.83692 | 5.645277 | 2.578907 | 111.4777 |
| 14 | -31.4746 | -24.0355 | -0.75161 | 14.40715 | -180.859 |
| 15 | -65.8611 | 378.3563 | 149.2125 | 171.4194 | 5085.001 |
| 16 | -10.5121 | 0.179355 | -0.45465 | 0.929214 | 323.5687 |
| 17 | -12.714 | 45.05648 | 18.25757 | 56.17532 | 837.1872 |
| 18 | 52.27106 | 83.01338 | 31.17727 | 14.96918 | 260.9932 |
| 19 | -1.18138 | 7.509593 | 2.161707 | -8.5424 | 51.93012 |
| 20 | 5.9641 | 12.62925 | 4.742374 | 6.336416 | 109.4735 |
| 21 | -1.27512 | 32.038 | 11.69834 | 1.095256 | 227.3373 |
| 22 | -56.4621 | -46.1365 | -27.5161 | 24.07277 | -278.157 |
| 23 | -14.0703 | -133.151 | -66.9319 | 61.52836 | -1543.93 |
| 24 | 1698.577 | 3544.374 | 1462.962 | -419.639 | -1193.95 |
| 25 | 0.11985 | 0.705785 | 0.323283 | 0.435861 | -24.2083 |
| 26 | -13.0397 | 0.049874 | 2.389038 | -0.53312 | -42.6127 |
| 27 | 18.9513 | 16.15439 | 3.284081 | 0.474821 | -53.4271 |
| 28 | 485.4409 | 307.788 | 48.47935 | 264.1557 | -3927.66 |
| 29 | -67.5703 | -241.802 | -79.2974 | -34.7015 | -555.735 |
| 30 | -135.633 | -536.914 | -218.963 | -39.4147 | -6260.25 |
| 31 | 14.34075 | 12.66548 | 4.748905 | -8.18409 | -238.465 |
| 32 | -0.14409 | -3.0343 | -1.12735 | -0.78006 | -13.2235 |
| 33 | 11.08673 | 10.33655 | 1.852003 | -2.47475 | 72.61649 |
| 34 | 142.5391 | 158.7801 | 42.54997 | -85.1681 | -2312.02 |
| 35 | 31.03443 | -81.3331 | -28.3711 | -6.62341 | -912.857 |
| 36 | -412.802 | -440.334 | -120.086 | -289.096 | -241.142 |
| 37 | -108.808 | -54.1522 | -9.53925 | -39.1188 | 1290.452 |
| 38 | 120.12 | 252.547 | 102.9737 | 39.67869 | 2661.318 |
| 39 | 32.0453 | 56.2634 | 17.97993 | 30.85379 | 607.5422 |
|  |  |  |  |  |  |


| 40 | 1825.534 | 945.4077 | 26.69633 | 2222.685 | -985.093 |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 41 | -1037 | -1653.02 | -543.285 | -191.254 | 4380.919 |
| 42 | -11.99 | -123.151 | -42.5558 | -13.9233 | 6110.568 |
| 43 | -1.49503 | -0.38292 | 0.00732 | -0.48612 | -4.33912 |
| 44 | 101.7451 | 286.0449 | 115.4895 | -117.401 | 1506.563 |
| 45 | -0.27275 | 0.407931 | 0.155146 | -0.26746 | -14.9631 |
| 46 | 15.7625 | -32.0647 | -15.0489 | 3.144633 | -55.5632 |
| 47 | -0.26683 | -1.46223 | -0.51735 | 0.290518 | 16.19181 |
| 48 | -27.227 | -35.6049 | -12.4628 | 1.512562 | -36.9865 |
| 49 | 18.14222 | 52.09413 | 17.94774 | -2.16669 | -650.676 |
| 50 | -51.5351 | 175.2881 | 38.39952 | 20.37056 | -660.311 |
| 51 | -1.86916 | -15.6808 | -7.46456 | 5.188321 | 167.2047 |
| 52 | 7.180553 | 4.500288 | 1.593348 | 0.44575 | 117.2479 |
| 53 | 31.91433 | 64.60545 | 20.65274 | -5.83927 | 525.3109 |
| 54 | -15.4019 | -14.8964 | -4.00914 | -2.28108 | -335.765 |
| 55 | 36.69831 | 26.84192 | 8.337921 | 2.852014 | 184.1147 |
| 56 | -73.597 | -159.053 | -55.6893 | -2.65664 | 912.7849 |
| 57 | -7.61372 | -4.98379 | -1.19372 | 0.111651 | -90.9199 |
| 58 | 18.95355 | 91.21082 | 29.8049 | 10.86698 | -483.486 |
| 59 | -54.0661 | -59.557 | -16.5754 | -22.3487 | -89.1549 |
| 60 | -11.9957 | -5.85748 | -2.21499 | -4.16681 | -73.3862 |
| 61 | -11.9335 | -66.3467 | -23.4337 | -3.19321 | -128.108 |
| 62 | -19.9121 | 44.62086 | 16.05319 | -25.8973 | 788.2178 |
| 63 | 125.4184 | 644.5135 | 258.8239 | 89.32673 | -3382.49 |
| 64 | -792.604 | -855.636 | -314.777 | -602.606 | 12.19406 |
| 65 | 169.9448 | 276.4051 | 93.90329 | -5.86725 | -408.062 |
| 66 | -0.02642 | -1.22E-01 | -4.47E-02 | 0.035197 | 1.273651 |
| 67 | 1.579685 | 2.66575 | 0.948187 | -0.12004 | -10.0382 |
| 68 | 5.812266 | -17.8738 | -7.42056 | -6.76427 | 103.1289 |
| 69 | 1.278992 | 1.019102 | 0.284436 | 0.197692 | 65.97762 |
| 70 | -3.75983 | -2.51769 | -0.67492 | 1.085311 | -202.474 |
| 71 | 134.1715 | -66.9452 | -12.6597 | 182.3807 | -844.273 |
| 72 | 0.492664 | 3.229711 | 1.255421 | 0.205865 | -52.2118 |
| 73 | $0.172228$ | -0.02004 | 0.009085 | -0.17123 | 18.1943 |
| 74 | 0.035346 | 0.008903 | $9.66 \mathrm{E}-05$ | $6.65 \mathrm{E}-03$ | -0.81285 |
| 75 | -0.33719 | 6.473197 | 2.362633 | 1.884306 | -182.953 |

Values of $a_{i j k l}$ as $i=2$ and $k=5$

| $l$ | $j=1$ | $j=2$ | $j=3$ | $j=4$ | $j=5$ |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 1 | -91.1466 | 41.88163 | 11.78747 | -20.5501 | 296.962 |
| 2 | 284.8035 | 559.8033 | 174.4235 | 27.77461 | 2509.356 |
| 3 | -35.2662 | -46.5451 | -13.9244 | -21.1279 | -56.5467 |
| 4 | 251.633 | 13.81974 | 8.047224 | 33.45825 | -418.356 |
| 5 | -39.8797 | 415.761 | 147.3547 | 123.8376 | -1027.4 |
| 6 | -8.30846 | -49.0111 | -19.6824 | 14.13027 | -333.775 |
| 7 | 241.9392 | 320.7208 | 97.51562 | 189.3129 | -922.675 |
| 8 | -1232.71 | -1206.79 | -301.42 | -1034.15 | 547.6515 |
| 9 | -14.8628 | -30.1689 | -9.09392 | -10.5981 | 38.4016 |
| 10 | 130.727 | 128.8995 | 28.79317 | 144.773 | -414.835 |
| 11 | 118.1197 | -138.785 | -54.9018 | -85.0006 | 978.9585 |
| 12 | -1083.32 | -1212.84 | -369.514 | -415.459 | 729.6148 |
| 13 | 6.303139 | 12.83971 | 4.114748 | 2.832261 | 16.41613 |
| 14 | -155.854 | 11.10473 | 1.655323 | -6.763 | 173.965 |
| 15 | -152.675 | -146.535 | -75.978 | 5.00983 | 487.6932 |
| 16 | 2.324309 | -10.8303 | -2.87126 | -5.52094 | 9.375278 |
| 17 | -118.789 | -187.326 | -57.1791 | -58.8807 | 398.4444 |
| 18 | 15.88279 | 10.64914 | 3.655614 | 0.960771 | 17.9773 |
| 19 | 0.230045 | -9.81578 | -3.80903 | 1.30357 | 3.741436 |
| 20 | 5.11285 | 13.37289 | 4.623704 | 1.16332 | 13.37416 |
| 21 | 9.999075 | 55.59926 | 19.16798 | 10.67983 | 50.60761 |
| 22 | -28.6078 | -71.9119 | -21.9122 | -32.2152 | 177.9767 |
| 23 | 3345.983 | 3650.985 | 1076.154 | 1464.251 | -4390.34 |
| 24 | 2333.687 | 2841.14 | 843.6693 | 1850.879 | -6707.8 |
| 25 | 0.561226 | 1.22755 | 0.383507 | 0.63272 | -3.03847 |
| 26 | 1.64772 | 6.41693 | 1.933155 | 1.451509 | -6.09796 |
| 27 | -11.129 | -7.82515 | -1.31217 | -10.6819 | 30.34675 |
| 28 | -223.375 | -337.296 | -94.8207 | -241.057 | 970.6622 |
| 29 | -17.6198 | 42.06124 | 19.06222 | 28.57562 | -450.596 |
| 30 | 244.7725 | 141.2181 | 56.36262 | 109.2637 | -677.115 |
| 31 | 235.4309 | 269.6067 | 87.9781 | 60.46358 | 229.3098 |
| 32 | -0.52022 | -1.3495 | -0.45427 | -0.18777 | -1.10629 |
| 33 | 34.77208 | -4.83371 | -1.43816 | 0.168521 | -13.5241 |
| 34 | -37.0751 | -170.312 | -48.6425 | -67.8763 | 503.6025 |
| 35 | -27.066 | -15.8178 | -6.46383 | -3.93048 | 25.47744 |
| 36 | -2603.06 | -2263.62 | -593.11 | -1388.02 | 5429.122 |
| 37 | 22.57788 | 23.47423 | 5.788906 | 18.86089 | -79.7149 |
| 38 | -103.394 | -79.3888 | -31.7442 | -48.1844 | 407.1825 |
| 39 | -426.725 | -544.587 | -179.344 | -106.909 | -520.618 |
|  |  |  |  |  |  |


| 40 | 301.9714 | -2264.17 | -824.048 | -525.957 | 3280.422 |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 41 | -35.1758 | 177.2715 | 65.60788 | 31.95405 | -288.427 |
| 42 | -43.1156 | -266.997 | -86.0219 | -68.3958 | 7093.703 |
| 43 | 3.324733 | 2.820844 | 0.734558 | 2.408097 | -2.25118 |
| 44 | 239.7981 | 124.9931 | 41.52361 | 114.9343 | -482.606 |
| 45 | -1.0672 | -0.31637 | -0.00307 | -0.88429 | -1.44381 |
| 46 | -4.98189 | -6.47737 | -1.71768 | -2.78709 | -6.20757 |
| 47 | 10.38479 | 8.193655 | 2.163112 | 7.287927 | -21.4178 |
| 48 | -9.07087 | -14.2268 | -5.02933 | 1.285491 | -29.3546 |
| 49 | 21.99701 | 96.10596 | 34.42229 | -2.0894 | -374.881 |
| 50 | 158.5642 | -37.7395 | -11.3371 | 2.320152 | 34.9147 |
| 51 | -16.2151 | -13.4065 | -3.41505 | -12.9652 | 46.9621 |
| 52 | 4.116676 | 0.625365 | -0.11991 | 1.638032 | 42.07865 |
| 53 | -90.5584 | 9.236947 | 11.40788 | -135.852 | 200.4564 |
| 54 | -42.6534 | -58.3003 | -18.3473 | -19.981 | -7.84953 |
| 55 | -5.83632 | -1.16029 | 0.473035 | -9.66589 | 63.35701 |
| 56 | -106.584 | -216.474 | -75.1656 | -7.11467 | 478.298 |
| 57 | -0.16466 | -0.41285 | -0.17687 | 0.165692 | -3.67497 |
| 58 | 459.5353 | 543.7279 | 167.2488 | 356.273 | -389.167 |
| 59 | 16.27025 | 15.68163 | 3.996329 | 15.50289 | -70.7618 |
| 60 | -0.64224 | 5.315552 | 2.073186 | 0.030435 | -48.4671 |
| 61 | -94.6 | -146.544 | -49.7567 | -47.5651 | -42.1576 |
| 62 | -255.434 | -548.177 | -187.534 | -36.4812 | 1390.153 |
| 63 | 421.2633 | 2178.774 | 787.2571 | 109.7932 | -3955.02 |
| 64 | -167.874 | 259.0884 | 81.18981 | 158.2088 | -369.193 |
| 65 | 23.56999 | -2.41084 | -1.4711 | 0.331426 | 23.78449 |
| 66 | -0.01968 | -1.40E-02 | -4.38E-03 | -0.00338 | -0.09453 |
| 67 | 0.229434 | 0.22634 | 0.072866 | 0.065969 | -0.26047 |
| 68 | -25.5316 | -152.195 | -52.2079 | -39.0247 | 17.05542 |
| 69 | 4.850251 | 7.485818 | 2.566449 | 0.694149 | 29.40385 |
| 70 | -13.8702 | -12.8357 | -4.98366 | 5.002014 | -227.796 |
| 71 | -132.322 | $-65.2568$ | -23.4863 | -77.3621 | 304.1115 |
| 72 | 5.555799 | 6.997878 | 2.191217 | 2.340212 | -16.8139 |
| 73 | 0.524578 | 0.24118 | 0.093791 | -0.10203 | 7.221367 |
| 74 | -0.36652 | -0.48261 | -1.58E-01 | -1.12E-01 | -0.09825 |
| 75 | 11.87842 | 31.0456 | 9.914145 | 15.56078 | -222.427 |

Values of $a_{i j k l}$ as $i=2$ and $k=6$

| $l$ | $j=1$ | $j=2$ | $j=3$ | $j=4$ | $j=5$ |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 1 | 36.28515 | -11.0843 | 6.109857 | -14.9814 | 350.4956 |
| 2 | -326.841 | -632.017 | -205.191 | -132.265 | 3509.863 |
| 3 | -21.7348 | -60.8535 | -21.6878 | -20.1166 | 608.5373 |
| 4 | 100.3004 | 124.5875 | 22.6295 | -13.2702 | -1189.08 |
| 5 | -76.6034 | 499.9122 | 204.6522 | 88.86051 | 2608.978 |
| 6 | 36.09173 | 30.67715 | 1.737552 | 36.0295 | -1985.17 |
| 7 | 338.4193 | 912.206 | 300.764 | 217.2562 | 635.4033 |
| 8 | 468.5169 | 245.9778 | 165.1024 | -1210.62 | -2571.93 |
| 9 | 4.65077 | -88.1667 | -28.8615 | -25.252 | -695.663 |
| 10 | 55.39747 | 345.1582 | 101.482 | 342.206 | -1000.16 |
| 11 | 146.2218 | -44.0699 | -38.538 | 81.89257 | -843.956 |
| 12 | -583.164 | -1554.32 | -524.469 | -191.445 | 490.9358 |
| 13 | 3.476364 | 70.88302 | 25.65857 | 12.47765 | 182.5924 |
| 14 | -68.1597 | -65.0769 | -7.35638 | 13.08597 | 1016.793 |
| 15 | 202.7619 | -486.073 | -162.65 | -45.2706 | -4280.13 |
| 16 | -25.8416 | -16.9042 | -5.85947 | -14.6129 | -54.885 |
| 17 | -99.3106 | -569.571 | -195.441 | -45.8935 | 567.1278 |
| 18 | 61.36607 | 47.67608 | 14.52982 | 16.69075 | 38.40325 |
| 19 | 9.458142 | 8.816732 | 1.678885 | 5.059596 | 11.32516 |
| 20 | 9.647418 | 48.13281 | 18.70961 | -3.26591 | 327.4737 |
| 21 | 41.91501 | 26.60384 | 10.34505 | 2.412163 | 30.56245 |
| 22 | -31.185 | -173.316 | -56.8724 | -31.7694 | -326.552 |
| 23 | 357.4185 | 3126.526 | 1005.907 | 1296.138 | -1429.67 |
| 24 | 973.7342 | 912.5732 | 203.4193 | 1315.898 | -4548.2 |
| 25 | 5.177508 | 3.492883 | 1.164529 | 1.208209 | 6.261901 |
| 26 | -10.7964 | 13.7258 | 4.496752 | 1.801647 | 189.5316 |
| 27 | -19.2672 | -115.55 | -34.5838 | -99.4346 | 284.8922 |
| 28 | -230.892 | -542.326 | -177.667 | -244.54 | 1518.413 |
| 29 | -67.1271 | -0.13963 | 8.255735 | -42.6205 | -268.809 |
| 30 | 47.73467 | 282.2406 | 85.05237 | 6.8467 | 226.0594 |
| 31 | 472.5557 | 868.3733 | 306.2772 | -104.635 | -269.726 |
| 32 | -2.37271 | -15.9248 | -6.01166 | 0.80026 | -51.2478 |
| 33 | 19.24249 | 16.05116 | 1.630684 | -0.31794 | -234.648 |
| 34 | 49.05696 | 367.6328 | 107.7478 | 52.31295 | 2113.341 |
| 35 | -87.2013 | -73.7476 | -22.9571 | 3.660892 | 523.3445 |
| 36 | 665.1982 | -857.695 | -158.924 | -2131.29 | 1702.446 |
| 37 | 66.83327 | 288.1529 | 91.37465 | 141.9118 | -579.304 |
| 38 | -0.13261 | -109.676 | -37.0843 | 18.09027 | 679.3265 |
| 39 | -848.109 | -1624.07 | -577.808 | 244.3864 | 332.5435 |
|  |  |  |  |  |  |


| 40 | 926.5613 | 936.0841 | 157.2396 | 2111.643 | -4252.72 |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 41 | -318.329 | -853.144 | -242.819 | -644.977 | 880.6432 |
| 42 | -4.99143 | -153.938 | -54.6064 | 30.4368 | 5889.415 |
| 43 | 2.108694 | 17.26342 | 5.602971 | 9.701039 | -3.08509 |
| 44 | 264.7365 | 642.61 | 216.4063 | 174.2135 | 565.6809 |
| 45 | -0.27134 | -3.42128 | -1.1314 | -1.40123 | -14.4359 |
| 46 | -6.62698 | -51.9003 | -17.1918 | -24.683 | -290.128 |
| 47 | -0.80778 | 14.02359 | 4.820755 | 3.385636 | -28.4943 |
| 48 | -20.6626 | -18.2835 | -6.62111 | 4.260054 | 37.42463 |
| 49 | 33.70994 | 40.11081 | 14.90981 | -18.767 | -552.855 |
| 50 | 53.86612 | 75.59775 | 9.241135 | 15.58304 | -562.223 |
| 51 | -22.6006 | -105.763 | -34.4184 | -55.2551 | 58.52189 |
| 52 | 8.26111 | 10.59669 | 3.305904 | 5.839726 | 94.24878 |
| 53 | -25.9609 | -63.9554 | -26.7757 | -82.0618 | 14.76349 |
| 54 | -6.21651 | -31.9478 | -10.872 | -8.16293 | -82.147 |
| 55 | -4.57349 | -32.3182 | -9.81144 | -18.489 | 40.93036 |
| 56 | -92.6939 | -130.407 | -45.448 | 34.76539 | 1020.975 |
| 57 | -1.55383 | 1.341242 | 0.337718 | -0.45699 | -22.4043 |
| 58 | 149.5621 | 403.051 | 146.5921 | 163.8135 | -8.63257 |
| 59 | 24.39052 | 130.5158 | 42.6367 | 68.41628 | -32.783 |
| 60 | -8.80313 | -4.02765 | -1.1438 | -6.46816 | -132.663 |
| 61 | -78.5364 | -57.9042 | -20.177 | -19.4299 | 168.8323 |
| 62 | 94.57244 | -109.851 | -43.5663 | 39.47403 | 82.42797 |
| 63 | -0.36741 | 626.2228 | 269.5954 | -242.952 | -1812.67 |
| 64 | -533.421 | 391.8477 | 161.8612 | -88.7881 | 5874.339 |
| 65 | 78.73201 | 86.07125 | 25.11338 | 65.12043 | 12.25228 |
| 66 | -0.1214 | -7.04E-02 | -1.95E-02 | -0.03788 | 0.416545 |
| 67 | 1.398844 | 1.888147 | 0.642888 | 0.330816 | -3.97112 |
| 68 | 9.054928 | 2.552595 | -2.44523 | -12.2585 | -536.036 |
| 69 | 0.701729 | 1.064579 | 0.375203 | -0.30818 | 21.09021 |
| 70 | -4.39627 | -12.0002 | -3.85191 | 4.053924 | -33.7857 |
| 71 | -41.8677 | -330.602 | -117.196 | -58.089 | -1322.13 |
| 72 | 4.172794 | 7.66277 | 2.543442 | 1.570501 | -15.7311 |
| 73 | 0.452525 | 0.384319 | 0.119053 | -0.14586 | 0.599946 |
| 74 | -0.12974 | -0.22271 | -7.35E-02 | -2.25E-02 | 0.404764 |
| 75 | -2.62347 | 2.160051 | 1.329271 | 5.015072 | -4.49507 |

Values of $a_{i j k l}$ as $i=3$ and $k=1$

| $l$ | $j=1$ | $j=2$ | $j=3$ | $j=4$ | $j=5$ |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 1 | 218.25 | 303.0773 | 100.8584 | 44.64509 | -257.231 |
| 2 | -52.7917 | 76.23853 | 41.58524 | -195.105 | 4807.331 |
| 3 | 24.74669 | -53.7252 | -19.5373 | 21.53868 | -514.531 |
| 4 | 157.1421 | 168.951 | 53.78774 | 10.745 | -437.368 |
| 5 | -258.509 | 988.3755 | 222.3491 | 340.6676 | 4211.01 |
| 6 | -11.6952 | -40.1156 | -15.1724 | 5.966876 | -309.36 |
| 7 | 325.9142 | 362.6444 | 122.4495 | 104.8477 | -2894.02 |
| 8 | 972.8215 | 1536.197 | 548.3815 | 165.0483 | -7905.96 |
| 9 | -57.4249 | -38.7372 | -14.5182 | -12.5222 | 518.9638 |
| 10 | -217.902 | -334.617 | -103.682 | 32.90281 | -4.1395 |
| 11 | 810.1781 | 159.2567 | 108.1827 | 83.06012 | -3795.76 |
| 12 | -289.647 | -337.663 | -119.673 | -5.87794 | -348.792 |
| 13 | 4.705723 | 19.94366 | 7.051665 | -1.09941 | 0.08499 |
| 14 | -40.8544 | -58.6096 | -14.5421 | 7.416047 | 81.50176 |
| 15 | 381.4816 | 289.031 | 280.0819 | 39.89343 | -716.832 |
| 16 | -24.0076 | -34.2148 | -12.4512 | 11.29469 | 116.4543 |
| 17 | 18.26424 | -57.5105 | -16.7658 | -68.7145 | 86.82515 |
| 18 | 93.49547 | 116.4338 | 27.82037 | -0.97056 | 115.7929 |
| 19 | 6.7933 | 4.533286 | 1.496067 | 2.408662 | 26.30263 |
| 20 | 1.506191 | 4.970431 | 1.918416 | -1.5083 | 12.58271 |
| 21 | 49.05149 | 90.33387 | 32.20277 | -53.6527 | -95.0838 |
| 22 | -34.0492 | -34.3369 | -10.5278 | -20.4288 | 589.1172 |
| 23 | 365.8818 | 484.5593 | 133.3077 | 310.7154 | 6412.88 |
| 24 | 5650.786 | 7976.689 | 2450.826 | 2770.212 | -9239.86 |
| 25 | 5.010016 | 4.357545 | 1.61489 | 0.02436 | -8.92705 |
| 26 | 5.545759 | 1.655262 | 1.016114 | 0.932718 | -146.809 |
| 27 | 13.06435 | 28.37006 | 10.37044 | -1.76072 | 13.01389 |
| 28 | -1183.81 | -2701.89 | -825.239 | -622.686 | 2408.378 |
| 29 | -264.919 | -139.318 | -54.3873 | -0.93466 | 1082.549 |
| 30 | -426.299 | -267.387 | -343.61 | 244.576 | 641.9583 |
| 31 | 91.36751 | 103.0262 | 36.93386 | 13.22653 | 51.25779 |
| 32 | -0.87235 | -2.1659 | -0.77922 | 0.102254 | 1.031283 |
| 33 | 5.89014 | 10.07159 | 1.513826 | -1.42623 | 33.33969 |
| 34 | 968.5847 | 408.2492 | -14.5917 | 601.5585 | -3525.97 |
| 35 | -70.0672 | -69.1346 | -23.5673 | 6.928884 | -134.914 |
| 36 | -487.788 | -16.9571 | 41.27175 | -269.905 | -806.774 |
| 37 | 61.35597 | 133.1443 | 40.02091 | 47.58405 | -187.617 |
| 38 | 163.9864 | 29.61264 | 49.78934 | 25.07638 | 763.3233 |
| 39 | -244.258 | -331.757 | -109.484 | -71.3327 | -1695.09 |
|  |  |  |  |  |  |


| 40 | -2792.14 | -15007.4 | -5501.57 | -546.539 | 7351.822 |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 41 | -512.153 | 1490.532 | 703.2819 | -856.083 | 1895.392 |
| 42 | 46.44546 | -99.9174 | -26.5766 | -34.7222 | 16740.18 |
| 43 | -5.94149 | -6.10779 | -1.99735 | -0.4911 | 11.44316 |
| 44 | 159.5854 | 197.7668 | 62.88335 | 75.83098 | -1065.3 |
| 45 | -6.55135 | -7.39812 | -2.5238 | 0.670159 | 1.531972 |
| 46 | -12.5732 | -8.22696 | -1.91221 | -7.97109 | 152.482 |
| 47 | 9.307288 | 21.62614 | 7.66782 | -4.6756 | -21.2326 |
| 48 | -83.5276 | -149.96 | -47.0233 | -18.9187 | -23.0421 |
| 49 | -5.73604 | 177.3101 | 73.68531 | -11.9793 | -884.989 |
| 50 | 506.8943 | 253.7736 | 125.2569 | -23.1978 | -1721.33 |
| 51 | -10.6879 | -22.2321 | -8.06437 | -14.7783 | 89.08929 |
| 52 | 29.10243 | 44.13888 | 12.37613 | 0.409488 | 100.4808 |
| 53 | 5.276971 | 119.8441 | 35.42992 | -60.1607 | 140.5748 |
| 54 | -31.1857 | -12.0125 | -8.03705 | 47.7766 | -779.764 |
| 55 | -171.755 | -330.82 | -115.765 | -43.3838 | 723.8636 |
| 56 | -578.637 | -1417.78 | -448.054 | -436.812 | 663.0574 |
| 57 | -16.0151 | -19.7719 | -5.07542 | -6.02225 | -38.6085 |
| 58 | 559.7436 | 1075.801 | 417.0869 | -143.799 | -2173.97 |
| 59 | 137.4734 | 295.6694 | 99.62044 | 42.34005 | -519.191 |
| 60 | 12.16209 | 25.52999 | 17.51545 | -29.471 | -150.22 |
| 61 | -177.358 | -305.6 | -110.227 | 111.6169 | 712.2724 |
| 62 | -295.539 | -526.03 | -126.917 | -804.77 | -1789.07 |
| 63 | 1734.776 | 7305.717 | 2466.937 | 2175.105 | -6183.53 |
| 64 | -968.297 | -59.8161 | 183.9409 | -468.886 | 476.9404 |
| 65 | 301.5485 | 380.13 | 24.11426 | 252.5157 | 114.4765 |
| 66 | 0.173485 | $1.52 \mathrm{E}-01$ | 5.16E-02 | -0.03936 | -0.02789 |
| 67 | 2.081299 | 3.562093 | 1.153899 | 0.950925 | -3.73232 |
| 68 | -26.1457 | -188.985 | -70.2823 | -27.4967 | 217.5876 |
| 69 | -4.78284 | -10.0167 | -3.90586 | -2.90155 | 227.4633 |
| 70 | -6.2671 | -20.4069 | -6.79231 | 12.12949 | -400.748 |
| 71 | -214.503 | -322.054 | -118.819 | -68.298 | 38.3429 |
| 72 | 74.0594 | 151.1852 | 47.48064 | 41.74895 | -134.014 |
| 73 | 6.778128 | 12.6607 | 3.545759 | 2.229545 | 26.11342 |
| 74 | -0.78719 | -2.50379 | -0.72442 | -4.91E-01 | -3.8614 |
| 75 | -0.10831 | 16.0464 | 5.155077 | 1.361957 | -471.929 |

Values of $a_{i j k l}$ as $i=3$ and $k=2$

| $l$ | $j=1$ | $j=2$ | $j=3$ | $j=4$ | $j=5$ |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 1 | 286.2932 | 530.445 | 166.6531 | 38.51684 | 430.5148 |
| 2 | -1096.07 | -51.3552 | -112.949 | 20.00701 | 2400.809 |
| 3 | 148.1235 | 119.1137 | 55.65609 | 3.475428 | 1782.218 |
| 4 | 126.7489 | -162.853 | -24.5327 | -38.4976 | -2217.11 |
| 5 | 563.4732 | -760.083 | -209.409 | -383.381 | 17701.37 |
| 6 | -32.9997 | -84.4536 | -29.3839 | 10.3092 | -2700.43 |
| 7 | 872.9705 | 182.1376 | 140.4726 | -62.781 | 1907.648 |
| 8 | 1076.406 | 1697.956 | 502.3687 | 38.77606 | -10842.7 |
| 9 | -25.0257 | -121.562 | -51.9473 | 13.13538 | -1134.61 |
| 10 | 158.0291 | -382.09 | -72.0672 | -186.629 | -1140.91 |
| 11 | 95.15232 | 2280.69 | 738.9587 | 551.0778 | -13295.7 |
| 12 | 118.0992 | 24.78266 | 4.813531 | 15.69072 | -799.848 |
| 13 | -83.237 | 17.00532 | 4.315526 | -4.73321 | 67.8578 |
| 14 | -32.7444 | 206.9363 | 45.79321 | 33.41202 | 1331.75 |
| 15 | 578.1324 | 3544.186 | 1574.51 | 1637.818 | -1400 |
| 16 | -62.441 | -38.084 | -15.8054 | 19.28002 | 341.4637 |
| 17 | 23.363 | -403.695 | -134.847 | -42.0147 | 827.914 |
| 18 | 54.43178 | 202.5611 | 64.66949 | 8.499899 | -988.185 |
| 19 | -13.1947 | -11.8362 | -3.03765 | -4.04236 | 68.10479 |
| 20 | 26.58493 | 34.26436 | 12.084 | -3.85028 | 460.674 |
| 21 | 10.0323 | -18.3122 | -2.36608 | -11.5988 | -99.1029 |
| 22 | -174.287 | 11.54189 | -16.5516 | 6.630192 | -599.924 |
| 23 | -477.099 | -105.646 | -7.2245 | 106.1284 | 9192.042 |
| 24 | 4863.979 | 9314.153 | 3834.548 | -2514.62 | -32643.4 |
| 25 | 34.8361 | 21.52625 | 7.896838 | -6.4107 | -20.3515 |
| 26 | -21.7215 | 6.003039 | 4.848872 | -0.29819 | 189.0376 |
| 27 | -38.0727 | 42.39394 | 7.449276 | 10.82593 | 1186.452 |
| 28 | -554.226 | -1684.74 | -951.578 | 3794.181 | 20268.39 |
| 29 | 46.73216 | -601.377 | -186.737 | -107.704 | 3651.903 |
| 30 | 462.5063 | -1379.44 | -758.032 | -177 | -10832.3 |
| 31 | -13.9408 | 19.50795 | 6.042342 | -2.79795 | 275.4519 |
| 32 | 8.140945 | -11.3164 | -3.57508 | 2.392327 | -1.89721 |
| 33 | 10.23277 | -52.3222 | -11.4062 | -7.76795 | -251.286 |
| 34 | -987.476 | -3173.5 | -1570.49 | 827.5935 | -13709.9 |
| 35 | -67.1169 | -8.92526 | -7.92017 | 2.13245 | -69.8971 |
| 36 | -949.985 | -1820.27 | -676.989 | -98.1465 | 12357.9 |
| 37 | 26.69159 | 528.2737 | 214.6947 | -256.388 | -9733.77 |
| 38 | -70.6701 | 681.9571 | 243.9935 | 233.4099 | 4657.402 |
| 39 | 138.2029 | 104.3515 | 29.76793 | 1.545415 | -3422.05 |
|  |  |  |  |  |  |


| 40 | 5534.531 | 13420.59 | 3364.378 | 12549.57 | -65977.1 |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 41 | -3358.19 | -15426.7 | -4737.73 | -8793.6 | 74128.22 |
| 42 | 99.2132 | 25.10454 | 11.58001 | -6.94231 | 15073.93 |
| 43 | -23.2649 | -22.0955 | -7.81569 | 2.537928 | 41.41763 |
| 44 | 356.2758 | 19.91573 | 48.87001 | -16.9867 | 1721.812 |
| 45 | -7.12592 | -6.2502 | -2.16698 | 1.367712 | 8.137522 |
| 46 | 19.92511 | -43.5182 | -20.6078 | -1.36509 | -1037.8 |
| 47 | 3.224682 | 5.61836 | 2.124147 | -2.34383 | -35.5899 |
| 48 | -24.1405 | -124.173 | -37.349 | 10.56467 | -365.636 |
| 49 | 35.72 | 97.6085 | 35.02665 | -8.81151 | -95.188 |
| 50 | 140.5902 | 822.868 | 245.612 | 147.6605 | -7727.26 |
| 51 | -141.847 | -123.696 | -50.6239 | 57.3245 | 792.1327 |
| 52 | 1.842556 | 21.78025 | 5.101034 | 2.242917 | 212.1518 |
| 53 | -23.5553 | -15.8232 | -7.28413 | -0.45541 | 159.6351 |
| 54 | -80.1913 | -53.757 | -16.2864 | -47.5209 | -1258.5 |
| 55 | 18.46678 | -119.392 | -74.0626 | 95.95609 | 3240.848 |
| 56 | -212.253 | -574.355 | -175.115 | -146.327 | -669.603 |
| 57 | -16.7663 | -91.6087 | -32.4802 | -10.5229 | -102.075 |
| 58 | 282.845 | 411.7926 | 169.6225 | -6.18184 | -690.788 |
| 59 | -167.391 | -189.303 | -21.8162 | -734.79 | -5838.68 |
| 60 | -48.7222 | -56.4492 | -7.98184 | -27.1197 | 256.9391 |
| 61 | -70.7638 | -55.2691 | -20.0099 | 17.87759 | 165.4736 |
| 62 | -120.616 | -647.297 | -228.165 | -90.2809 | -1076.18 |
| 63 | 528.9031 | 2950.016 | 1085.019 | 100.953 | 9779.486 |
| 64 | -1126.46 | -3582.4 | -1227.99 | -1152.45 | 5978.749 |
| 65 | 352.9063 | 1937.142 | 586.8217 | 179.6073 | -3175.87 |
| 66 | 0.666407 | $3.86 \mathrm{E}-01$ | $1.47 \mathrm{E}-01$ | -0.15989 | -1.17498 |
| 67 | 8.692291 | 16.15565 | 5.556477 | -2.5561 | -22.2991 |
| 68 | -10.3747 | -39.0423 | -16.9355 | 2.876602 | -397.034 |
| 69 | 2.407267 | -1.49894 | -0.83288 | 5.963574 | 184.6632 |
| 70 | -2.06546 | -0.25498 | -0.11417 | 0.833093 | -238.306 |
| 71 | -104.323 | 134.1542 | 47.79632 | 107.0194 | -465.648 |
| 72 | 14.68279 | 43.15271 | 15.78948 | -0.83418 | -364.777 |
| 73 | -5.83938 | 0.269501 | -0.04666 | 0.055402 | 51.76334 |
| 74 | 0.564276 | 0.146343 | 0.040339 | -1.83E-01 | -4.41737 |
| 75 | -4.41408 | -1.87758 | -0.65838 | 0.136536 | -315.53 |

Values of $a_{i j k l}$ as $i=3$ and $k=3$

| $l$ | $j=1$ | $j=2$ | $j=3$ | $j=4$ | $j=5$ |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 1 | 256.6162 | 463.2325 | 153.9968 | 105.0543 | -792.788 |
| 2 | 40.588 | 1556.698 | 558.4326 | 151.5778 | -2318.11 |
| 3 | -12.0307 | -124.496 | -41.6259 | -24.7641 | 41.09128 |
| 4 | 46.91667 | -140.17 | -53.6138 | -3.63083 | 347.8436 |
| 5 | 160.1489 | 157.2549 | 41.93267 | -8.07298 | -65.2596 |
| 6 | -80.1329 | -138.049 | -45.2889 | -23.2822 | -166.455 |
| 7 | 686.5398 | -211.274 | -113.608 | 126.0022 | 1127.582 |
| 8 | 3159.19 | 6452.573 | 2141.889 | 1789.243 | -13157.4 |
| 9 | -39.8301 | 35.43012 | 13.85889 | -1.50465 | 77.06305 |
| 10 | -295.762 | -757.134 | -253.819 | -132.88 | 1021.815 |
| 11 | -223.312 | -534.902 | -165.838 | -16.9638 | 789.6319 |
| 12 | -1356.86 | -1663.5 | -539.304 | -420.462 | 2411.829 |
| 13 | 14.49986 | 28.8837 | 9.2965 | 5.995358 | -34.0445 |
| 14 | -36.0218 | 17.66327 | 10.39495 | 3.060883 | -245.563 |
| 15 | -161.846 | 333.0862 | 125.823 | 270.269 | -349.917 |
| 16 | -21.9587 | -35.9949 | -11.336 | -15.8847 | 133.8275 |
| 17 | -181.222 | -229.958 | -75.5267 | -83.1707 | 774.0592 |
| 18 | 49.81008 | 61.70283 | 20.39771 | 6.122823 | -18.5547 |
| 19 | 1.956655 | 16.45078 | 5.913173 | 3.358711 | -36.4352 |
| 20 | 13.53996 | 20.26123 | 6.562035 | 5.591674 | -13.2129 |
| 21 | 177.2531 | 285.0325 | 90.18969 | 73.11291 | -453.615 |
| 22 | -156.441 | -8.69675 | 9.078679 | -24.0134 | -228.78 |
| 23 | 1730.113 | -77.4598 | -55.6933 | 92.88388 | 1702.783 |
| 24 | -4358.34 | -11220.1 | -3731.33 | -2420.77 | 16997.63 |
| 25 | -0.8409 | -5.10038 | -1.69752 | -0.01347 | 5.466827 |
| 26 | 14.26024 | 8.166181 | 1.967289 | 2.177501 | -52.7728 |
| 27 | 19.97171 | 51.86126 | 17.38864 | 9.807774 | -78.9398 |
| 28 | 220.5381 | 899.7828 | 306.5778 | 109.7223 | -892.564 |
| 29 | -28.5631 | 27.31454 | 5.889659 | -24.0466 | -125.609 |
| 30 | 373.5908 | 637.8955 | 201.8087 | -175.899 | -1092.38 |
| 31 | 523.128 | 790.9577 | 259.1567 | 182.6661 | -1124.35 |
| 32 | -1.35594 | -2.45525 | -0.79265 | -0.63997 | 3.341018 |
| 33 | 6.356264 | -5.77723 | -2.7278 | -0.96403 | 83.34026 |
| 34 | 11.84376 | 367.6545 | 130.8268 | 10.1418 | -629.432 |
| 35 | -53.2112 | -18.266 | -5.39539 | -13.147 | -101.337 |
| 36 | 1781.926 | 5607.152 | 1855.722 | 1247.332 | -10048 |
| 37 | -15.7295 | -57.9026 | -19.5308 | -5.51657 | 51.48461 |
| 38 | -30.5154 | -164.61 | -52.266 | 36.71561 | 544.5323 |
| 39 | -1130.52 | -1684.67 | -550.079 | -383.615 | 2201.229 |
|  |  |  |  |  |  |


| 40 | -2081.12 | -2783.77 | -779.702 | -1387.7 | 9332.187 |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 41 | 179.4047 | 289.1881 | 83.44323 | 159.3816 | -1022.19 |
| 42 | -462.408 | -804.27 | -210.638 | -447.5 | 18460.9 |
| 43 | -4.08277 | -8.48174 | -2.82256 | -1.81588 | 13.76694 |
| 44 | 472.6321 | -233.508 | -101.522 | -10.9267 | 1605.582 |
| 45 | -5.74185 | -11.057 | -3.6444 | -1.75632 | 10.73372 |
| 46 | -23.8553 | 4.935572 | 2.337915 | 3.775559 | 11.45841 |
| 47 | 40.21388 | 136.1523 | 45.646 | 22.8069 | -188.615 |
| 48 | -65.3521 | -90.8318 | -28.9604 | -21.7801 | 146.7051 |
| 49 | 224.5951 | 79.67034 | 18.16736 | 117.335 | -830.061 |
| 50 | 42.45943 | 1.173628 | -0.60373 | 17.72925 | -210.385 |
| 51 | 32.79662 | 88.96359 | 29.55171 | 13.82795 | -111.422 |
| 52 | 21.43217 | 26.20236 | 8.200424 | 0.590246 | -2.31389 |
| 53 | 504.1079 | 1150.751 | 367.0016 | 35.27511 | -788.868 |
| 54 | 72.81179 | 216.638 | 72.83541 | 64.30192 | -489.354 |
| 55 | 70.38893 | 119.2864 | 38.65117 | 15.4145 | -114.365 |
| 56 | -723.932 | -1327.78 | -434.945 | -376.257 | 2798.873 |
| 57 | -3.07067 | -3.61755 | -1.13587 | 0.061252 | -0.49707 |
| 58 | 480.1145 | 755.9097 | 245.9201 | 632.6732 | -3204.59 |
| 59 | -46.6701 | -134.424 | -44.8298 | -20.6138 | 159.5645 |
| 60 | -23.7092 | -21.0463 | -6.49282 | 3.161974 | -16.4231 |
| 61 | -468.341 | -715.592 | -228.177 | -207.152 | 1280.498 |
| 62 | -7.0401 | 2064.879 | 693.1952 | 327.951 | -3263.16 |
| 63 | 4847.55 | 2454.355 | 628.9487 | 1655.126 | -6627.98 |
| 64 | 981.6209 | 1254.897 | 369.4115 | 482.3844 | -2764.54 |
| 65 | 25.7293 | -66.3244 | -20.4973 | -54.0011 | 343.1691 |
| 66 | 0.140875 | $3.16 \mathrm{E}-01$ | $1.05 \mathrm{E}-01$ | 0.044203 | -0.29932 |
| 67 | 0.328846 | -0.15598 | -0.06344 | 0.193872 | -1.04648 |
| 68 | -38.827 | 2306.071 | 808.3894 | 237.9857 | -3354.55 |
| 69 | -35.2111 | -86.7885 | -29.033 | -29.7097 | 244.9435 |
| 70 | -176.619 | -631.175 | -211.652 | -78.9218 | 258.9839 |
| 71 | -417.888 | -14.5196 | 8.410208 | -45.8855 | -532.362 |
| 72 | 11.36503 | 19.19053 | 6.317366 | 6.655038 | -52.5853 |
| 73 | 1.015014 | 3.738867 | 1.230175 | -0.2038 | 2.230453 |
| 74 | 0.71663 | 0.852648 | 0.286055 | $4.94 \mathrm{E}-01$ | -2.93284 |
| 75 | -4.50904 | -42.7491 | -18.5414 | -7.25421 | -403.836 |

Values of $a_{i j k l}$ as $i=3$ and $k=4$

| $l$ | $j=1$ | $j=2$ | $j=3$ | $j=4$ | $j=5$ |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 1 | 130.7408 | 205.6688 | 63.60501 | 59.10683 | -146.279 |
| 2 | 66.20939 | -2710.64 | -726.578 | -184.023 | -205.462 |
| 3 | -334.606 | 0.094728 | -4.50159 | 25.40892 | -400.713 |
| 4 | 411.3077 | 440.7219 | 145.651 | -15.4511 | -509.598 |
| 5 | -968.971 | 705.7318 | 194.8802 | 221.5443 | 4357.873 |
| 6 | -93.6842 | -183.254 | -49.9811 | -47.7512 | -1319.12 |
| 7 | 68.25424 | 2431.622 | 692.7048 | 228.1446 | 1044.361 |
| 8 | 22.11455 | -21.9955 | 4.802865 | -1261.11 | 3431.496 |
| 9 | 238.9509 | -195.852 | -63.3681 | -41.5888 | 439.1533 |
| 10 | -60.9416 | 665.8487 | 250.0875 | 306.8986 | -3566.5 |
| 11 | 1668.64 | 291.809 | 112.0324 | 204.404 | -3644.6 |
| 12 | -506.417 | -658.156 | -227.801 | 68.66266 | -579.849 |
| 13 | 31.5657 | 101.3058 | 30.84944 | 12.42782 | 88.46077 |
| 14 | -283.002 | -237.014 | -76.5199 | 19.53904 | 259.758 |
| 15 | 14.80728 | 1190.025 | 364.0817 | 2.139081 | -4043.15 |
| 16 | -42.0052 | -105.231 | -37.8452 | -12.3235 | 582.4437 |
| 17 | 216.451 | 326.0933 | 98.74309 | 58.24749 | 665.8342 |
| 18 | 267.0724 | 209.4846 | 63.64586 | 11.66653 | -68.041 |
| 19 | -17.3585 | -49.6963 | -15.2023 | -18.2489 | 340.1619 |
| 20 | 20.03036 | 43.62907 | 12.85593 | 11.60541 | 50.66537 |
| 21 | 113.5176 | 278.5556 | 91.1526 | 3.873903 | 51.41966 |
| 22 | 73.74016 | -497.612 | -134.32 | -48.7299 | -474.39 |
| 23 | -350.187 | -3495.57 | -1233.44 | -684.2 | 11760.67 |
| 24 | 8538.205 | 19404.93 | 6709.648 | 7276.09 | -47548.1 |
| 25 | -1.20005 | 7.645839 | 3.75494 | -0.06933 | -40.103 |
| 26 | -67.4858 | 47.73283 | 14.66451 | 10.73952 | -118.914 |
| 27 | -45.1514 | -156.31 | -54.9973 | -45.2784 | 697.1597 |
| 28 | -1541.39 | -5713.48 | -2070.44 | -872.945 | 15372.62 |
| 29 | -551.956 | -95.1426 | -41.2275 | -39.4908 | 702.9475 |
| 30 | -101.375 | -455.827 | -108.667 | 170.8926 | -2420.56 |
| 31 | 451.6526 | 967.9085 | 332.23 | 126.6862 | -2329.18 |
| 32 | -4.63515 | -17.0692 | -5.38209 | -1.90101 | -12.0364 |
| 33 | 70.74892 | 45.94641 | 14.60151 | -5.55172 | -46.0811 |
| 34 | 874.3758 | 2351.03 | 807.4652 | -292.357 | -1790.49 |
| 35 | -179.004 | -278.208 | -93.7908 | 31.59733 | -1218.85 |
| 36 | 1920.05 | 10629.73 | 3762.606 | 723.0519 | -31428.4 |
| 37 | 237.4797 | 993.3464 | 362.9107 | 82.88244 | -1939.54 |
| 38 | 147.1831 | -223.414 | -88.2813 | -12.5797 | 2453.098 |
| 39 | -1130.85 | -2600.35 | -891.436 | -305.154 | 6605.413 |
|  |  |  |  |  |  |


| 40 | -5022.37 | -32473.6 | -11592.6 | -3727.34 | 66710.11 |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 41 | -2589.65 | -1616.35 | -286.332 | -1048.43 | -969.542 |
| 42 | -69.2442 | -250.178 | -60.299 | -156.986 | 14949.86 |
| 43 | -6.10694 | -1.42754 | -0.22542 | 2.451425 | -3.09714 |
| 44 | 450.4199 | 2006.876 | 563.0474 | 152.2295 | 2085.156 |
| 45 | -5.36113 | -6.44843 | -1.94848 | 0.704313 | 4.296171 |
| 46 | 89.99407 | -125.706 | -34.9101 | -30.2172 | 73.30468 |
| 47 | 11.68051 | 22.80749 | 7.154055 | -1.53576 | -81.7947 |
| 48 | -127.295 | -167.816 | -53.4945 | -24.2827 | -15.667 |
| 49 | 45.42026 | 261.9329 | 86.61698 | 30.92975 | -1197.99 |
| 50 | 775.594 | 339.3305 | 151.6973 | 48.17912 | -1550.85 |
| 51 | -6.17965 | -75.7551 | -35.4452 | -35.5025 | 417.9945 |
| 52 | 53.3391 | 51.12946 | 14.93157 | 7.461431 | 105.4645 |
| 53 | 135.4208 | 177.0778 | 58.73625 | 26.63667 | 730.716 |
| 54 | -97.0185 | -134.185 | -34.7548 | -25.0411 | -65.1891 |
| 55 | 64.67963 | -143.662 | -58.4397 | 24.85274 | 811.476 |
| 56 | -345.333 | -957.277 | -330.641 | -68.0853 | 2259.944 |
| 57 | -58.2471 | -75.1912 | -19.5828 | -10.5515 | -20.396 |
| 58 | 444.2433 | 1342.957 | 431.7463 | 91.70507 | -3107.85 |
| 59 | -39.9455 | 276.4444 | 115.3702 | 3.674087 | -993.597 |
| 60 | -17.2859 | 64.79581 | 22.2395 | -18.4211 | -244.396 |
| 61 | -192.564 | -415.903 | -138.068 | -25.7601 | 412.1888 |
| 62 | 81.00381 | 296.3504 | 113.0416 | -114.615 | 1432.34 |
| 63 | 407.993 | 2697.591 | 945.9167 | 963.085 | -8239.23 |
| 64 | 864.2456 | 4410.033 | 1344.741 | -579.207 | 3807.267 |
| 65 | 682.6015 | 171.1483 | -19.2605 | 218.705 | -2051.89 |
| 66 | 0.143234 | -1.36E-02 | -2.13E-02 | -0.12074 | 0.282264 |
| 67 | 5.484153 | 9.584523 | 3.280085 | 1.691773 | -16.8224 |
| 68 | 48.65529 | -2.65015 | 0.947305 | -38.1792 | -82.0862 |
| 69 | 7.301707 | 8.622702 | 2.158534 | 2.388438 | 64.66247 |
| 70 | -26.733 | -45.9368 | -15.461 | 1.840558 | -221.301 |
| 71 | -866.125 | -1897.95 | -547.732 | 3.247849 | -1632.67 |
| 72 | 7.03555 | 35.14847 | 13.27183 | 0.336557 | -120.377 |
| 73 | 4.800761 | 7.042649 | 2.166646 | 0.457988 | 14.24306 |
| 74 | -0.23531 | -0.97686 | -0.32533 | -4.71E-02 | 1.227495 |
| 75 | -4.52768 | 6.968797 | 2.301609 | 6.908468 | -265.907 |

Values of $a_{i j k l}$ as $i=3$ and $k=5$

| $l$ | $j=1$ | $j=2$ | $j=3$ | $j=4$ | $j=5$ |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 1 | 222.2768 | 435.0211 | 141.369 | 52.96851 | -725.973 |
| 2 | -562.126 | -207.655 | -102.547 | -302.499 | -257.684 |
| 3 | 99.26563 | 174.9375 | 67.34943 | 68.2906 | -82.9429 |
| 4 | 74.24806 | -38.4995 | -13.6539 | 55.24309 | 606.305 |
| 5 | 983.6725 | 630.2193 | 157.5492 | 246.9502 | -3884.84 |
| 6 | -134.663 | -334.714 | -109.458 | -81.8393 | -625.058 |
| 7 | 1011.209 | 19.11491 | -2.00874 | 258.5951 | 2433.061 |
| 8 | 4627.735 | 14771.48 | 5032.474 | 2502.027 | -21007 |
| 9 | -161.492 | -63.0284 | -19.158 | -68.9178 | -233.465 |
| 10 | -574.308 | -2511.5 | -865.996 | -330.239 | 3547.615 |
| 11 | -671.347 | -637.097 | -167.2 | -16.1695 | 1835.998 |
| 12 | -1301.97 | -859.375 | -264.989 | -201.383 | 196.1417 |
| 13 | 34.2489 | 26.91994 | 6.666177 | 8.024872 | 104.4562 |
| 14 | 34.4743 | 63.42538 | 18.79425 | -6.93846 | -493.802 |
| 15 | -739.438 | 129.6537 | 84.46612 | -1.96844 | 682.204 |
| 16 | -61.7705 | -12.9491 | -1.32604 | 2.582131 | -12.1765 |
| 17 | -645.617 | -415.793 | -117.792 | -151.44 | 964.5782 |
| 18 | 176.0915 | 120.3016 | 35.89223 | 51.80738 | -231.218 |
| 19 | -23.9011 | -48.6671 | -15.4928 | -18.162 | 67.81376 |
| 20 | 81.5539 | 103.5859 | 32.3894 | 32.16815 | -10.7137 |
| 21 | 207.6758 | 303.4923 | 95.94376 | 93.58941 | -284.774 |
| 22 | -169.184 | -18.3399 | -4.74014 | -44.4383 | -597.553 |
| 23 | 121.0953 | -5221.61 | -1754.22 | -1188.11 | 13144.36 |
| 24 | -1200.41 | -17308.2 | -6187.43 | 239.172 | 3149.372 |
| 25 | 12.06069 | 10.97797 | 3.404984 | 6.193232 | -34.7874 |
| 26 | 25.93312 | 15.16445 | 4.675156 | 11.73185 | 87.71116 |
| 27 | 68.01041 | 370.569 | 126.5323 | 65.24156 | -642.982 |
| 28 | -165.987 | 2191.5 | 815.1445 | -246.974 | -696.168 |
| 29 | 68.98291 | 152.4831 | 42.24417 | 18.9078 | -863.917 |
| 30 | 581.8633 | -36.22 | -56.1098 | -205.039 | 1668.774 |
| 31 | 758.0643 | 1309.604 | 425.5961 | 298.0612 | -2130.73 |
| 32 | -12.4646 | -19.3162 | -6.04352 | -5.69948 | 14.03742 |
| 33 | -15.7511 | -29.1551 | -8.68955 | -2.84312 | 151.3445 |
| 34 | 714.9871 | 288.6299 | 93.42444 | 100.5071 | -1634.28 |
| 35 | -186.719 | -289.574 | -96.684 | -137.67 | 590.8913 |
| 36 | 4274.919 | 13895.94 | 4566.899 | 3128.981 | -27110.5 |
| 37 | 40.27203 | -369.258 | -130.253 | -25.7594 | 553.2382 |
| 38 | -132.209 | -60.9142 | -9.7331 | -32.8219 | 504.784 |
| 39 | -1443.33 | -2665.97 | -865.652 | -586.308 | 4127.735 |
|  |  |  |  |  |  |


| 40 | -4526.53 | -5998.95 | -1499.42 | -4742.16 | 31475.19 |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 41 | 40.27587 | 1237.169 | 368.6553 | 660.2086 | -4442.41 |
| 42 | -257.905 | -146.113 | 21.29111 | -379.18 | 16060.68 |
| 43 | -2.72555 | -37.204 | -12.8407 | -4.23784 | 57.18577 |
| 44 | 643.9719 | -227.266 | -81.3855 | 99.8151 | 3081.279 |
| 45 | -5.11902 | 6.813714 | 2.50181 | 0.283883 | -15.5675 |
| 46 | -50.3249 | 40.94519 | 13.34099 | -4.80323 | -342.22 |
| 47 | 4.857448 | -32.513 | -11.1429 | 1.755152 | 5.015922 |
| 48 | -49.564 | -96.5439 | -31.4462 | -13.4294 | 70.66957 |
| 49 | 84.38457 | 372.84 | 123.4843 | 80.30981 | -1222.15 |
| 50 | -149.536 | -267.165 | -81.0533 | -21.0031 | 1689.716 |
| 51 | -1.06876 | 202.8083 | 70.32979 | 11.18133 | -214.903 |
| 52 | 32.36853 | -5.70646 | -2.94775 | -6.06 | 135.2429 |
| 53 | 180.8686 | 754.1269 | 235.7179 | 33.4874 | -615.633 |
| 54 | -44.9939 | -8.24177 | -2.33531 | -12.725 | 21.34565 |
| 55 | -1.01374 | 71.33082 | 23.9819 | -11.777 | 30.57447 |
| 56 | -320.629 | -795.555 | -261.082 | -187.477 | 2117.294 |
| 57 | 0.777537 | 0.10838 | -0.20943 | 0.957544 | -13.4495 |
| 58 | 523.5839 | -119.756 | -41.0826 | 401.8076 | -644.764 |
| 59 | 6.456549 | -249.583 | -86.4984 | -7.28357 | 218.9559 |
| 60 | -24.5837 | 18.93035 | 7.331877 | 15.85494 | -199.476 |
| 61 | -388.002 | -456.222 | -143.271 | -177.485 | 640.674 |
| 62 | -294.505 | -214.65 | -64.786 | -95.7588 | 1246.708 |
| 63 | 2587.941 | 6884.874 | 2186.827 | 2004.276 | -15571.1 |
| 64 | 399.9112 | -72.3881 | -85.3204 | -133.336 | 6651.123 |
| 65 | 226.846 | -61.4829 | -27.5459 | -7.75553 | 37.54057 |
| 66 | -0.09914 | -1.72E-01 | -5.73E-02 | -0.01081 | -0.26582 |
| 67 | 1.380651 | 1.600583 | 0.51189 | 0.528714 | -2.23224 |
| 68 | -153.7 | -303.644 | -92.7819 | -110.034 | 206.378 |
| 69 | 8.271952 | 14.71245 | 4.851862 | 2.797616 | 2.249926 |
| 70 | -24.0828 | -48.5725 | -16.5708 | -5.64739 | -192.329 |
| 71 | -471.073 | 100.1316 | 53.70238 | -31.343 | -2739.83 |
| 72 | 12.99458 | 12.38207 | 3.873968 | 5.704016 | -39.0264 |
| 73 | 2.195322 | 3.148127 | 1.040078 | 0.470717 | 3.669553 |
| 74 | -0.71306 | -0.97 | -0.31401 | -3.06E-01 | 1.810905 |
| 75 | 35.16493 | 16.62524 | 2.232005 | 26.64922 | -321.656 |

Values of $a_{i j k l}$ as $i=3$ and $k=6$

| $l$ | $j=1$ | $j=2$ | $j=3$ | $j=4$ | $j=5$ |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 1 | 295.0208 | 415.5822 | 153.7631 | -35.0702 | 57.08297 |
| 2 | -1306.78 | -3012.67 | -949.566 | -985.844 | 4264.388 |
| 3 | 78.60355 | 351.1901 | 109.0579 | 118.08 | 44.40504 |
| 4 | -73.0342 | 174.1173 | 17.4897 | 38.1489 | -2014.76 |
| 5 | 1980.111 | 1724.671 | 608.9249 | 1193.795 | 4865.141 |
| 6 | 98.57575 | 266.2051 | 81.4566 | -48.7209 | -1775.63 |
| 7 | 1143.218 | 1904.234 | 579.224 | 1086.326 | -595.38 |
| 8 | 2828.139 | 9766.869 | 3263.693 | -2728.91 | -3252.62 |
| 9 | -59.3734 | -332.717 | -96.4245 | -225.883 | 479.4748 |
| 10 | -728.979 | -1637.1 | -534.088 | 713.3469 | -2923.05 |
| 11 | -503.958 | -461.702 | -175.806 | -248.69 | -3222.77 |
| 12 | -1305.99 | -1465.16 | -442.849 | -332.835 | -489.329 |
| 13 | 21.35793 | 43.68381 | 10.71445 | 43.65318 | -218.261 |
| 14 | 92.26312 | -89.5178 | -2.39518 | 22.46928 | 1432.387 |
| 15 | -1640.6 | -1704.4 | -504.138 | -1120.21 | -1014.35 |
| 16 | 12.61647 | 3.503251 | 3.209872 | -7.09768 | -143.546 |
| 17 | -1112.64 | -1976.21 | -621.118 | -334.408 | 3970.29 |
| 18 | 498.8135 | 465.124 | 136.8629 | 247.8655 | -768.224 |
| 19 | 97.58753 | 180.246 | 55.99548 | 27.51191 | 279.2139 |
| 20 | 41.08898 | 42.13288 | 13.84037 | 39.49591 | -90.2194 |
| 21 | 164.0418 | 331.5845 | 106.0416 | 80.06464 | -0.03726 |
| 22 | -190.336 | -421.782 | -127.659 | -209.51 | 147.9391 |
| 23 | 1707.704 | -1823.97 | -757.231 | 315.0778 | 13947.3 |
| 24 | 4818.974 | 3403.083 | 1193.739 | 11621.25 | -57628.3 |
| 25 | -5.94013 | 7.887949 | 2.28265 | 6.617147 | 85.3186 |
| 26 | -13.1347 | 76.32933 | 22.34438 | 36.27484 | -194.528 |
| 27 | 123.7578 | 595.7824 | 212.7256 | -52.1595 | 128.3416 |
| 28 | -1475.36 | -3881.25 | -1331.33 | -3176.25 | 18456.76 |
| 29 | -66.9374 | 57.34469 | 28.20476 | 9.97124 | 420.5312 |
| 30 | 300.2521 | 7.842626 | -87.0063 | 345.3163 | -410.998 |
| 31 | 307.8706 | 768.248 | 256.4051 | 77.97361 | -2254.19 |
| 32 | -12.1391 | -39.7182 | -12.7649 | -12.4205 | 82.23912 |
| 33 | -19.6625 | 10.15445 | -3.30709 | -6.55172 | -299.853 |
| 34 | 592.0423 | 374.0739 | 102.078 | 42.6731 | -1005.14 |
| 35 | -454.158 | -733.63 | -239.147 | -202.373 | 580.6569 |
| 36 | -561.654 | 7586.542 | 2711.451 | -93.9156 | -30180.1 |
| 37 | 358.8378 | 128.1079 | 5.047393 | 300.4977 | -1244.65 |
| 38 | 32.09708 | -106.23 | -10.9795 | -135.471 | 1345.774 |
| 39 | -451.475 | -1584.06 | -531.972 | -99.6537 | 4569.922 |
|  |  |  |  |  |  |
| 1 |  |  |  |  |  |


| 40 | -2423.57 | -20382.5 | -7105.68 | -9780.91 | 86098.04 |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 41 | -857.844 | 5591.722 | 2102.573 | 1270.88 | -17272.2 |
| 42 | -241.679 | -408.238 | -83.6309 | -331.802 | 15412.99 |
| 43 | 10.92104 | -28.2924 | -10.7549 | 24.50628 | 59.58249 |
| 44 | 1057.313 | 1431.399 | 427.8497 | 917.0895 | 239.8886 |
| 45 | -19.6998 | -20.1616 | -5.95269 | -7.24964 | 53.07756 |
| 46 | -84.7091 | -143.79 | -40.5836 | -133.814 | 24.84329 |
| 47 | 51.54035 | 60.40987 | 18.26097 | 23.84253 | -231.963 |
| 48 | -48.3508 | -118.208 | -39.8931 | -2.26656 | 101.6701 |
| 49 | -95.9905 | 139.5608 | 51.28153 | -47.7094 | -1017.54 |
| 50 | -237.323 | -9.69843 | -40.2454 | -23.192 | -2080.49 |
| 51 | -162.796 | 50.17782 | 24.9674 | -177.67 | -167.851 |
| 52 | 49.30644 | 19.69908 | 4.383268 | 12.89847 | 80.95124 |
| 53 | 153.5906 | 494.9164 | 158.2765 | -60.7918 | -235.317 |
| 54 | -69.1633 | -72.5459 | -20.6584 | -28.9212 | 56.81897 |
| 55 | -34.3586 | 49.48454 | 20.82711 | -58.895 | 36.02258 |
| 56 | -3.68746 | -468.673 | -162.115 | 49.08297 | 2346.056 |
| 57 | -12.7924 | -16.0525 | -4.93887 | 1.468222 | -13.1458 |
| 58 | 371.643 | 236.0849 | 67.57551 | 461.3836 | -1374.02 |
| 59 | 277.8733 | 38.7194 | -0.04995 | 250.6502 | 190.345 |
| 60 | -19.5701 | 43.97451 | 15.97693 | 1.934103 | -227.625 |
| 61 | -188.855 | -296.257 | -96.1745 | -117.085 | 315.2062 |
| 62 | 2.513111 | 43.84181 | 14.4557 | 64.84794 | 1962.695 |
| 63 | 1006.217 | 4854.465 | 1629.612 | 1037.728 | -11590.4 |
| 64 | -65.5563 | 1473.823 | 385.2499 | 587.3212 | 3281.789 |
| 65 | 846.2421 | 298.7504 | 60.23684 | 486.2506 | -1087.7 |
| 66 | 1.130004 | $2.06 \mathrm{E}+00$ | $6.59 \mathrm{E}-01$ | 0.176271 | -5.92821 |
| 67 | 1.808594 | 2.012121 | 0.667238 | 1.601805 | -1.04017 |
| 68 | 31.07702 | 24.9518 | 8.490489 | -62.7042 | -656.015 |
| 69 | 0.591916 | 2.071061 | 0.714534 | -1.08235 | 28.26171 |
| 70 | -34.7246 | -70.2912 | -23.4174 | 0.7341 | 8.326148 |
| 71 | -688.422 | -985.397 | -276.944 | -608.247 | -417.991 |
| 72 | 6.820427 | 5.290224 | 1.399855 | 5.458983 | -41.8803 |
| 73 | 2.413397 | 4.132303 | 1.352063 | 0.215362 | 0.141965 |
| 74 | -0.01003 | 0.024617 | 0.014275 | -5.53E-02 | 1.028359 |
| 75 | 2.626346 | -28.088 | -10.7327 | 25.86615 | -121.259 |

## Appendix C

The input properties of 100 test samples

| sample \# | $E_{f}(\mathrm{GPa})$ | $\sigma_{f}(\mathrm{MPa})$ | $E_{S}(\mathrm{GPa})$ | $\sigma_{f L} / \sigma_{f T}$ | $n$ |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 1 | 125 | 300 | 75 | 1.15 | 0.05 |
| 2 | 125 | 300 | 175 | 1.15 | 0.05 |
| 3 | 125 | 300 | 75 | 1.35 | 0.05 |
| 4 | 125 | 300 | 175 | 1.35 | 0.05 |
| 5 | 125 | 300 | 75 | 1.15 | 0.25 |
| 6 | 125 | 300 | 175 | 1.15 | 0.25 |
| 7 | 125 | 300 | 75 | 1.35 | 0.25 |
| 8 | 125 | 300 | 175 | 1.35 | 0.25 |
| 9 | 125 | 300 | 75 | 1.15 | 0.35 |
| 10 | 125 | 300 | 175 | 1.15 | 0.35 |
| 11 | 125 | 300 | 75 | 1.35 | 0.35 |
| 12 | 125 | 300 | 175 | 1.35 | 0.35 |
| 13 | 125 | 300 | 75 | 1.15 | 0.45 |
| 14 | 125 | 300 | 175 | 1.15 | 0.45 |
| 15 | 125 | 300 | 75 | 1.35 | 0.45 |
| 16 | 125 | 300 | 175 | 1.35 | 0.45 |
| 17 | 225 | 300 | 75 | 1.15 | 0.05 |
| 18 | 225 | 300 | 175 | 1.15 | 0.05 |
| 19 | 225 | 300 | 75 | 1.35 | 0.05 |
| 20 | 225 | 300 | 175 | 1.35 | 0.05 |
| 21 | 225 | 300 | 75 | 1.15 | 0.25 |
| 22 | 225 | 300 | 175 | 1.15 | 0.25 |
| 23 | 225 | 300 | 75 | 1.35 | 0.25 |
| 24 | 225 | 300 | 175 | 1.35 | 0.25 |
| 25 | 225 | 300 | 75 | 1.15 | 0.35 |
| 26 | 225 | 300 | 175 | 1.15 | 0.35 |
| 27 | 225 | 300 | 75 | 1.35 | 0.35 |
| 28 | 225 | 300 | 175 | 1.35 | 0.35 |
| 29 | 225 | 300 | 75 | 1.15 | 0.45 |
| 30 | 225 | 300 | 175 | 1.15 | 0.45 |
| 31 | 225 | 300 | 75 | 1.35 | 0.45 |
| 32 | 225 | 300 | 175 | 1.35 | 0.45 |
| 33 | 125 | 700 | 75 | 1.15 | 0.05 |
| 34 | 125 | 700 | 175 | 1.35 | 0.05 |
|  |  |  |  |  |  |


| 35 | 125 | 700 | 75 | 1.15 | 0.25 |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 36 | 125 | 700 | 175 | 1.15 | 0.25 |
| 37 | 125 | 700 | 75 | 1.35 | 0.25 |
| 38 | 125 | 700 | 175 | 1.35 | 0.25 |
| 39 | 125 | 700 | 75 | 1.15 | 0.35 |
| 40 | 125 | 700 | 175 | 1.15 | 0.35 |
| 41 | 125 | 700 | 75 | 1.35 | 0.35 |
| 42 | 125 | 700 | 175 | 1.35 | 0.35 |
| 43 | 125 | 700 | 75 | 1.15 | 0.45 |
| 44 | 125 | 700 | 175 | 1.15 | 0.45 |
| 45 | 125 | 700 | 75 | 1.35 | 0.45 |
| 46 | 125 | 700 | 175 | 1.35 | 0.45 |
| 47 | 225 | 700 | 175 | 1.35 | 0.05 |
| 48 | 225 | 700 | 75 | 1.15 | 0.25 |
| 49 | 225 | 700 | 175 | 1.15 | 0.25 |
| 50 | 225 | 700 | 75 | 1.35 | 0.25 |
| 51 | 225 | 700 | 175 | 1.35 | 0.25 |
| 52 | 225 | 700 | 75 | 1.15 | 0.35 |
| 53 | 225 | 700 | 175 | 1.15 | 0.35 |
| 54 | 225 | 700 | 75 | 1.35 | 0.35 |
| 55 | 225 | 700 | 175 | 1.35 | 0.35 |
| 56 | 225 | 700 | 75 | 1.15 | 0.45 |
| 57 | 225 | 700 | 175 | 1.15 | 0.45 |
| 58 | 225 | 700 | 75 | 1.35 | 0.45 |
| 59 | 225 | 700 | 175 | 1.35 | 0.45 |
| 60 | 100 | 200 | 100 | 1 | 0.45 |
| 61 | 50 | 200 | 50 | 1 | 0.35 |
| 62 | 125 | 500 | 125 | 1.25 | 0.25 |
| 63 | 100 | 600 | 100 | 1.1 | 0.35 |
| 64 | 100 | 600 | 100 | 1.1 | 0.45 |
| 65 | 150 | 400 | 150 | 1.2 | 0.35 |
| 66 | 150 | 400 | 150 | 1.2 | 0.45 |
| 67 | 200 | 1000 | 200 | 1.3 | 0.35 |
| 68 | 200 | 1000 | 200 | 1.3 | 0.45 |
| 69 | 250 | 800 | 250 | 1.4 | 0.35 |
| 70 | 250 | 800 | 250 | 1.4 | 0.45 |
| 71 | 50 | 200 | 150 | 1.3 | 0.2 |
| 72 | 100 | 200 | 100 | 1 | 0.3 |
| 73 | 150 | 200 | 50 | 1.2 | 0 |
| 74 | 150 | 200 | 200 | 1.1 | 0.1 |
| 75 | 200 | 200 | 200 | 1.4 | 0.5 |
| 76 | 250 | 200 | 250 | 1.5 | 0.4 |


| 77 | 50 | 400 | 50 | 1.2 | 0.4 |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 78 | 100 | 400 | 100 | 1 | 0.5 |
| 79 | 150 | 400 | 150 | 1.1 | 0.1 |
| 80 | 150 | 400 | 200 | 1.3 | 0 |
| 81 | 200 | 400 | 200 | 1.4 | 0.2 |
| 82 | 250 | 400 | 250 | 1.5 | 0.3 |
| 83 | 50 | 600 | 200 | 1.1 | 0.3 |
| 84 | 100 | 600 | 250 | 1.4 | 0 |
| 85 | 150 | 600 | 100 | 1 | 0.5 |
| 86 | 150 | 600 | 50 | 1.3 | 0.1 |
| 87 | 200 | 600 | 150 | 1.2 | 0.2 |
| 88 | 250 | 600 | 150 | 1.5 | 0.4 |
| 89 | 50 | 800 | 200 | 1.3 | 0.1 |
| 90 | 100 | 800 | 50 | 1 | 0.3 |
| 91 | 150 | 800 | 100 | 1.1 | 0 |
| 92 | 150 | 800 | 250 | 1.4 | 0.2 |
| 93 | 200 | 800 | 150 | 1.2 | 0.5 |
| 94 | 250 | 800 | 200 | 1.5 | 0.4 |
| 95 | 50 | 1000 | 200 | 1.3 | 0.1 |
| 96 | 100 | 1000 | 50 | 1 | 0.3 |
| 97 | 150 | 1000 | 100 | 1.1 | 0.2 |
| 98 | 150 | 1000 | 150 | 1.2 | 0 |
| 99 | 200 | 1000 | 250 | 1.5 | 0.4 |
| 100 | 250 | 1000 | 200 | 1.4 | 0.5 |

## Appendix D

The output properties of all 100 test samples through the reverse analysis

| sample \# | $E_{f}(\mathrm{GPa})$ | $\sigma_{f}(\mathrm{MPa})$ | $E_{S}(\mathrm{GPa})$ | $\sigma_{f L} / \sigma_{f T}$ | $n$ |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 1 | 122 | 310 | 75 | 1.06 | 0.05 |
| 2 | 126 | 298 | 175 | 1.03 | 0.06 |
| 3 | 121 | 308 | 75 | 1.14 | 0.06 |
| 4 | 125 | 309 | 175 | 1.21 | 0.05 |
| 5 | 130 | 314 | 74 | 0.94 | 0.25 |
| 6 | 129 | 288 | 174 | 1.10 | 0.26 |
| 7 | 123 | 305 | 75 | 1.30 | 0.25 |
| 8 | 127 | 299 | 174 | 1.42 | 0.25 |
| 9 | 129 | 269 | 74 | 1.37 | 0.35 |
| 10 | 124 | 313 | 175 | 1.05 | 0.35 |
| 11 | 122 | 312 | 75 | 1.46 | 0.34 |
| 12 | 126 | 303 | 175 | 1.32 | 0.35 |
| 13 | 125 | 294 | 75 | 1.28 | 0.44 |
| 14 | 127 | 288 | 175 | 1.07 | 0.46 |
| 15 | 125 | 312 | 75 | 1.38 | 0.44 |
| 16 | 125 | 290 | 175 | 1.26 | 0.46 |
| 17 | 222 | 302 | 75 | 1.09 | 0.05 |
| 18 | 230 | 292 | 175 | 1.26 | 0.05 |
| 19 | 241 | 301 | 75 | 1.45 | 0.04 |
| 20 | 233 | 280 | 175 | 1.50 | 0.06 |
| 21 | 212 | 311 | 76 | 1.25 | 0.25 |
| 22 | 224 | 289 | 175 | 1.30 | 0.25 |
| 23 | 218 | 328 | 75 | 1.28 | 0.25 |
| 24 | 226 | 296 | 175 | 1.45 | 0.25 |
| 25 | 222 | 338 | 75 | 1.33 | 0.32 |
| 26 | 223 | 306 | 175 | 1.16 | 0.35 |
| 27 | 230 | 286 | 75 | 1.79 | 0.33 |
| 28 | 223 | 306 | 176 | 1.26 | 0.35 |
| 29 | 223 | 305 | 75 | 1.05 | 0.46 |
| 30 | 225 | 300 | 175 | 1.26 | 0.44 |
| 31 | 230 | 296 | 75 | 1.20 | 0.46 |
| 32 | 223 | 315 | 175 | 1.41 | 0.44 |
| 33 | 108 | 712 | 76 | 1.68 | 0.01 |
| 34 | 130 | 691 | 174 | 1.14 | 0.07 |
|  |  |  |  |  |  |


| 35 | 121 | 722 | 75 | 1.25 | 0.24 |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 36 | 122 | 734 | 176 | 1.10 | 0.24 |
| 37 | 128 | 663 | 75 | 1.58 | 0.24 |
| 38 | 125 | 702 | 175 | 1.28 | 0.25 |
| 39 | 127 | 700 | 75 | 0.97 | 0.37 |
| 40 | 125 | 692 | 176 | 1.24 | 0.34 |
| 41 | 125 | 755 | 75 | 1.19 | 0.35 |
| 42 | 130 | 653 | 173 | 1.45 | 0.35 |
| 43 | 134 | 609 | 75 | 0.89 | 0.50 |
| 44 | 127 | 671 | 175 | 1.11 | 0.46 |
| 45 | 131 | 518 | 75 | 1.17 | 0.51 |
| 46 | 126 | 691 | 175 | 1.33 | 0.45 |
| 47 | 237 | 693 | 173 | 1.18 | 0.06 |
| 48 | 240 | 607 | 74 | 1.44 | 0.25 |
| 49 | 223 | 721 | 175 | 1.13 | 0.25 |
| 50 | 229 | 681 | 75 | 1.48 | 0.24 |
| 51 | 225 | 718 | 175 | 1.14 | 0.26 |
| 52 | 225 | 753 | 75 | 1.02 | 0.35 |
| 53 | 222 | 708 | 175 | 1.22 | 0.34 |
| 54 | 219 | 820 | 75 | 1.15 | 0.34 |
| 55 | 227 | 683 | 175 | 1.34 | 0.35 |
| 56 | 223 | 588 | 75 | 1.40 | 0.45 |
| 57 | 229 | 686 | 175 | 1.06 | 0.46 |
| 58 | 221 | 627 | 75 | 1.48 | 0.46 |
| 59 | 227 | 712 | 175 | 1.23 | 0.46 |
| 60 | 101 | 193 | 100 | 1.12 | 0.44 |
| 61 | 49 | 241 | 49 | 0.98 | 0.30 |
| 62 | 125 | 507 | 125 | 1.20 | 0.25 |
| 63 | 93 | 705 | 101 | 1.12 | 0.32 |
| 64 | 99 | 649 | 100 | 1.02 | 0.45 |
| 65 | 150 | 411 | 150 | 1.11 | 0.35 |
| 66 | 153 | 377 | 149 | 1.23 | 0.45 |
| 67 | 199 | 984 | 201 | 1.25 | 0.36 |
| 68 | 199 | 956 | 200 | 1.32 | 0.46 |
| 69 | 251 | 800 | 250 | 1.31 | 0.36 |
| 70 | 249 | 834 | 250 | 1.40 | 0.44 |
| 71 | 51 | 196 | 149 | 1.34 | 0.20 |
| 72 | 99 | 212 | 100 | 0.93 | 0.30 |
| 73 | 137 | 201 | 50 | 1.03 | 0.01 |
| 74 | 149 | 206 | 200 | 1.03 | 0.10 |
| 75 | 207 | 179 | 199 | 1.05 | 0.53 |
| 76 | 248 | 204 | 250 | 1.54 | 0.40 |


| 77 | 46 | 517 | 51 | 1.08 | 0.37 |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 78 | 102 | 364 | 100 | 1.03 | 0.51 |
| 79 | 147 | 409 | 151 | 0.97 | 0.10 |
| 80 | 148 | 418 | 200 | 1.10 | 0.00 |
| 81 | 199 | 400 | 200 | 1.46 | 0.20 |
| 82 | 247 | 409 | 251 | 1.44 | 0.30 |
| 83 | 50 | 597 | 199 | 1.06 | 0.30 |
| 84 | 99 | 607 | 251 | 1.33 | 0.00 |
| 85 | 154 | 462 | 100 | 1.07 | 0.53 |
| 86 | 153 | 617 | 50 | 1.11 | 0.10 |
| 87 | 200 | 595 | 150 | 1.16 | 0.20 |
| 88 | 255 | 576 | 149 | 1.53 | 0.40 |
| 89 | 51 | 802 | 198 | 1.14 | 0.11 |
| 90 | 102 | 665 | 50 | 1.31 | 0.30 |
| 91 | 160 | 782 | 100 | 1.11 | 0.01 |
| 92 | 150 | 804 | 250 | 1.48 | 0.19 |
| 93 | 200 | 824 | 150 | 1.18 | 0.50 |
| 94 | 258 | 752 | 199 | 1.33 | 0.42 |
| 95 | 50 | 1055 | 199 | 1.14 | 0.10 |
| 96 | 119 | 723 | 49 | 0.81 | 0.39 |
| 97 | 152 | 1031 | 100 | 0.95 | 0.20 |
| 98 | 156 | 993 | 149 | 1.14 | 0.01 |
| 99 | 202 | 982 | 250 | 1.46 | 0.41 |
| 100 | 257 | 994 | 199 | 1.47 | 0.49 |

